

AIRFRAME MP 19(1) (1 and 2)		MAINTENANCE PROCEDURES COMPONENT REPLACEMENTS CANBERRA PR9		AP101B-0409A-5A3A 2nd Edition Sect 1 Chap 1		Maintenance Record Aircraft or Equipment Serial No: _____ Date: _____				MOD Form 707MS (Revised Jan 01)	
STARBOARD ELEVATOR - FITTING						Tradesman		Brief Details of suspected fault(s) and SNOW(s)		Supervisor	
INFORMATION IN THE TOPIC 5A2 IS TO BE CONSULTED THROUGHOUT THE WORK DETAILED ON THIS CARD											
Associated Procedures						Code		Man Hrs		Certification and TDM	
MP 28 and 225											
Special Tools and Equipment: Nil.											
BLOCK 1 AIRFRAME											
1. <u>Fitting</u>  Before carrying out Sub-item 1.1 the balance weight and arm is to be positioned in the aircraft tail section.											
1.1 Elevator. (i) Position. (ii) Move inboard to engage inboard and outboard hinge pins.											
1.2 Balance weight and arm. ) 1.3 Centre hinge bracket. ) Refit. 1.4 Centre hinge bracket bolt ) (3 off). )											
1.5 Centre hinge bracket nut (i) Refit. (3 off). (ii) Lock with split pin.											
1.6 Balance weight washer. Refit.											
1.7 Balance weight nut. (i) Refit. (ii) Lock with split pin.											
E0393(49)						Continued					

AIRFRAME MP 19(2)		MAINTENANCE PROCEDURES COMPONENT REPLACEMENTS CANBERRA PR9		AP101B-0409A-5A3A 2nd Edition Sect 1 Chap 1		Maintenance Record Aircraft or Equipment Serial No:		MOD Form 707MS (Revised Jan 01) Date:			
STARBOARD ELEVATOR - FITTING											
INFORMATION IN THE TOPIC 5A2 IS TO BE CONSULTED THROUGHOUT THE WORK DETAILED ON THIS CARD						Tradesman		Brief Details of		Supervisor	
BLOCK 1 AIRFRAME						Code	Man Hrs	Certification and TDM	suspected fault(s) and SNOW(s)	Man Hrs	Certification and TDM
1. <u>Fitting</u> (Contd)  When carrying out Sub-item 1.8 the aft bolt is to be located with bolt head to port.											
1.8 Coupling link. Reconnect.  When carrying out Sub-item 1.9 the connecting screw must be fitted into the lever and connecting rod simultaneously.  When carrying out Sub-item 1.9 the tab must be positioned as noted in MP 18 Sub-item 3.1.											
1.9 Tab operating rod. Reconnect.											
BLOCK 2 AIRFRAME											
2. <u>Completion</u>											
2.1 Tail cone. Refit (MP 28).											
2.2 Tab operating lever access panel. Refit.											
2.2 Tailplane rear wall and elevator leading edge. Check clearance through full range of elevator movement 1.27 PLUS OR MINUS 0.50 mm (0.05 PLUS OR MINUS 0.02 in.).											
2.3 Tailplane and elevator horn. Check clearance 6.35 PLUS OR MINUS 2.5 mm (0.25 PLUS OR MINUS 0.10 in.).											
2.4 Elevator. Functional test (MP 225).											
E0393(49A)											