# CANBERRA T Mk 17

# PILOT'S NOTES

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BY COMMAND OF THE DEFENCE COUNCIL

Prepared by Royal Air Force Handling Squadron

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Ministry of Defence (Air) July 1986 Reprint of AP101B-0417-15

#### CANBERRA T Mk 17

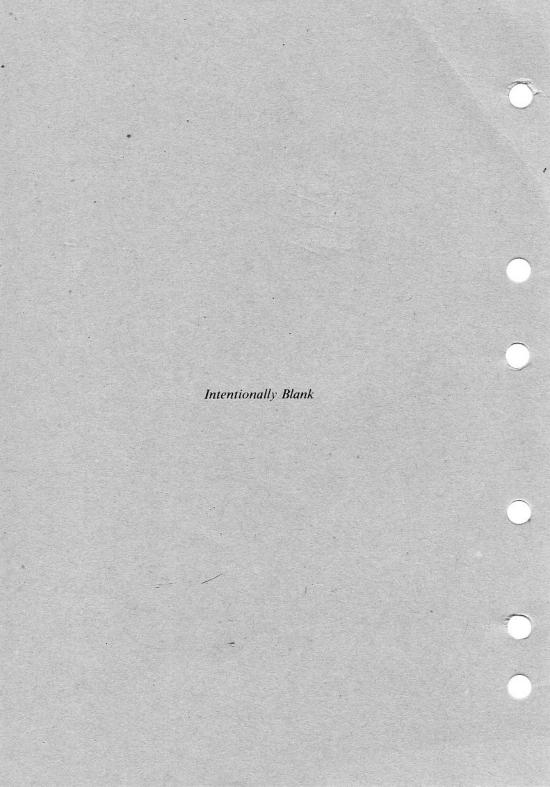
#### REPRINT OF PILOT'S NOTES

- 1. This reprint of the Pilot's Notes for the Canberra T Mk 17 (to AL 13 standard) should be placed in an anonymous 4-ring, A5 size binder. The 3-ring binder used for the previous issue is now obsolete, and the reprint has therefore been carried out in A5 size.
- 2. All copies of the previous issue (November 1967) of the Pilot's Notes are now obsolete and should be destroyed.
- 3. The major change introduced into the reprint under AL 13 is the addition of Supplement No 1. This Supplement describes the changes to the ECM, navigation, and communications equipment brought about by the incorporation of Mod 5466.

# AMENDMENT RECORD SHEET

Record the incorporation of each Amendment List in the publication by signing against the appropriate AL No and inserting the date of incorporation.

AL No	Amended By	Date
1 to 13	Already incorporated	
14	Jan	11 1/89
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# ANA RECORD

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### NOTES TO USERS

1 This book is divided by marker cards, as follows:

Preliminary Matter

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Part 1	Description and	Part 3	Handling
	Management of	Part 4	Emergencies
	Systems	Part 5	Illustrations

Part 2 Limitations

Where applicable, the parts are divided into chapters as listed on the marker cards. A Folio Sheet reference number is at the top left-hand corner of each sheet, each Part starting at FS1.

- 2 The limitations quoted in Part 2 are mandatory and are not to be exceeded except in an emergency. Instructions containing the word 'must' are also mandatory.
- 3 This book and its associated Flight Reference Cards aim to provide the best operating instructions and advice currently available. Although they provide guidance for most eventualities, they are not substitutes for sound judgement and good airmanship; moreover, they assume an adequate knowledge of the pertinent volumes of AP 3456. Series (Flying). Furthermore, circumstances might require aircrew to depart from or modify the prescribed procedure and drills. Consequently the Pilot's Notes and Flight Reference Cards should not be regarded as documents which are to be adhered to inflexibly at all times other than as explained in para 2 above.
- 4 Amendment Lists will be issued as necessary and each amendment list instruction sheet will state the main purpose of the amendment and will include a list of modifications covered in the text. New or amended matter of importance ◆ will be indicated by symbols in the text thus: ◀ .... ▶ or thus
- ♦....♦ to show the extent of amended text and thus ▶ or thus ♦ to show where text has been deleted. The number of the amendment list by which a sheet was initially issued, or re-issued, will appear at the bottom of the front pages and any amendment marks on either page forming a sheet will relate to that amendment. However, when a new chapter is issued with an amendment list or an existing chapter is completely revised, the fact will be indicated

within the heading of the chapter and the amendment marks ill not appear on the pages.

- 5 The following conventions are observed throughout this Book:
  - (a) The actual markings on controls are indicated in the text by capital letters.
  - (b) Unless otherwise stated all airspeeds, mach numbers, accelerations, temperatures and altitudes quoted are indicated values.
  - (c) **WARNINGS** are inserted only when the serious consequences of not following a certain procedure might otherwise be overlooked.
  - (d) Information which requires to be emphasised is printed in italics.
  - (e) Notes are inserted to clarify the reason for a procedure or to give information which, while not essential to the understanding of the subject, is useful to the reader.
  - (f) Cross references given in the text refer to chapters in the same part unless otherwise stated.

Modification numbers are only referred to in the text nen it is necessary to differentiate between pre- and postmod states. For ease of reference, a list of the modifications mentioned in the text is included in these preliminary pages, with a cross reference to the location in the text of the modification details.

7 This book is complementary to the Flight Reference Cards (AP 101B-0417-14) and the Operating Data Manual (AP 101B-0417-16) for the aircraft.

# **IMPORTANT**

Comments and suggestions should be forwarded to the Officer Commanding, Royal Air Force, Handling Squadron, Boscombe Down, Salisbury, SP4 0JF.

FS1A CANBERRA T MK.17 OXYGEN CHARGING VALVE IN LOWER EQUIPMENT BAY (PORT SIDE) FUEL FILLER HYDRAULIC ACCUMULATOR (BRAKES) AIR CHARGING VALVE, PRESSURE GAUGE AND NITROGEN CHARGING PANEL (PORT SIDE) ALTERNATOR SUMP HYDRAULIC TANK FILLER CAP (PORT AND STARBOARD) ALTERNATOR HYDRAULIC ACCUMULATOR AIR CHARGING VALVE (PORT AND STARBOARD) GROUND PRESSURE CABIN PRESSURE TEST CONNECTION CONNECTION VENTILATED SUIT GROUND SUPPLY CONNECTION CABIN PRESSURE TEST CONNECTION AND GROUND PRESSURE CONNECTION ARMAMENT BREAK ON THE PRESSURE BULKHEAD, ACCESSIBLE THROUGH THE NOSE WHEEL WELL ENGINE SUMP OIL FILLER ON PORT SIDE OF EACH ENGINE HYDRAULIC ACCUMULATOR BRAKE HYDRAULIC TEST AIR CHARGING VALVE AND PRESSURE GAUGE IN MAIN U/C WHEEL WELL (STARBOARD EXTERNAL A.C. AND D.C. SUPPLY SOCKETS IN STARBOARD CONNECTION ON MAIN U/C FAIRING (PORT AND STARBOARD) ONLY) EQUIPMENT BAY

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# LIST OF FOLIO SHEETS

This list shows all the Folio Sheets which should be present in these Pilot's Notes at AL16. Sheets which should have manuscript amendments are marked with an asterisk.

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FS4	AL14	Marker card	Initial Issue
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FS4	AL6	FS4	AL14
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	AL6	FS6	AL14
FS6	AL9	FS7	AL14
FS7	AL6	FS8	AL16
FS8		FS9	AL13
FS8A	AL6	FS10	AL13
FS9	AL4	FS10 FS11	AL13
FS9A	AL14	ron	ALIJ

#### INTRODUCTION

#### 1 General

The Canberra T Mk 17 is employed in the electronic countermeasures (ECM) training role. It is powered by two Avon Mk 102 engines. Three crew (pilot, navigator and air electronics officer (AEO)) are normally carried.

#### 2 Electrical System

- (a) DC is provided by two 28-volt, engine-driven generators and four 12-volt batteries connected in series-parallel. If the main DC supplies fail an emergency battery supplies various emergency circuits.
- (b) AC is supplied by two turbo-alternators driven by air from the compressor on each engine. Either alternator can supply the entire AC load required in flight, but provision is made for alternative supplies to instruments by two inverters. A dedicated inverter supplies Omega.

#### 3 Fuel System

Fuel is carried in three fuselage tanks and may be augmented by wing tip tanks. The forward and centre tanks are self-sealing and the rear tank is a crash-proof collapsible bag. Fuel can be fed, via LP pumps, from the fuselage tanks to either engine.

#### 4 Hydraulic System

Hydraulic power, produced by two hydraulic pumps, one on each engine accessory gearbox, is used to operate the undercarriage, flaps, airbrakes, wheelbrakes and air inlet scoops. A handpump circuit is provided for emergency lowering of the undercarriage and to charge the wheelbrakes accumulator if the main system fails.

### 5 Engines

The Avon Mk 102 engine is a turbo-jet having a twelve-stage axial-flow compressor directly coupled to a two-stage turbine. It gives approximately 6500 lb static thrust at sea level. The throttle controls incorporate a fast idling stop, set at approximately 4800 RPM, below which the turbo-alternators come off line.

#### 6 Aircraft Controls

The rudder, aileron and elevator controls are conventional and manually operated. The variable incidence tailplane, rudder trim tab and the aileron bias system, are electrically operated. To provide adequate clearance for the pilot during ejection, a snatch unit pulls the control column forward automatically when the pilot ejects. Selectors for undercarriage, flaps, airbrakes and air inlet scoops are electrically actuated.

### 7 Flight Instruments, Radio and Radar

Flight instruments are conventional. A combined omnibearing selector and deviation indicator combines the functions of a VOR bearing selector and ILS deviation indicator. Radio and radar equipment includes UHF, VHF, HF, VOR/ILS, radio compass (ADF), radio altimeter, Tacan, IFF/SSR and Green Satin (pre-Mod 5466) and, post-Mod 5466, an Omega system. A radio mixing system, incorporating intercom, is provided. Communications recording equipment is installed.

#### 8 Equipment Compartments

- (a) Three bays for various items of aircraft equipment are aft of the cabin pressure bulkhead. The upper equipment bay is above the nose undercarriage well and access to it is via a removable hatch on top of the fuselage. The left and right equipment bays are on either side of the nose undercarriage well, with access doors in the fuselage wall.
- (b) Additional equipment may be carried in two packs in the pack bay in the lower centre fuselage and in wing tip pods which may be fitted in lieu of wing tip tanks.

#### 9 Fire Protection

Fire detection, warning and extinguishing systems are fitted for each engine and for the pack bay. Two non-toxic hand-operated fire extinguishers are in the cabin. Provision is made for automatic operation of the engine and pack bay extinguishers in crash conditions.

#### 10 Doors and Emergency Exits

The entrance door, pilot's canopy and navigator's hatch are all jettisonable and serve as emergency exits. The navigator's hatch is automatically jettisoned when either the navigator or AEO eject.

#### 11 Ejection Seats

All three crew are provided with ejection seats incorporating single lever ejection. The pilot ejects through the canopy.

### 12 Oxygen System

Oxygen is carried in bottles in the upper equipment bay and supplied to crew members via Mk 17 pressure demand regulators. An emergency oxygen bottle is fixed to each ejection seat.

### 13 Air Conditioning and Pressurization

The air conditioning and pressurization systems are supplied by air bleed from the engine compressors. The temperature is controlled by a mixing valve by means of which a proportion of hot air from the compressors is passed through coolers in the inner plane leading edge and a cold air unit in the left inner plane.

#### 14 Location of Controls and Indicators

(a) The location of controls and indicators is referred to in these notes by reference to the following panels and compartments.

Left wall Take-off panel Left console Left front panel Flight instrument panel Engine instrument panel Engine starter panel Right instrument panel Coaming panel Electrical control panel (front face)	) ) ) ) ) ) )	Cockpi†
Left wall Navigator's instrument panel Coaming panel AEO's control unit panel Right wall Electric control panel (rear face)	) ) ) )	Cabin

(b) The location of controls is illustrated in Part 5.

# MODIFICATIONS MENTIONED IN THE TEXT

	Mod No	Title		ocatio Chapte		2
•	5466	Introduces new ECM equipment plus Omega	Suj	pleme	nt 1	330
	5541	Introduces pack bay overheat warning light	1	8	13	
		SEM				
	175	Introduces sealed lead-acid batteries	1	1	4	
	185	Introduces high intensity strobe lights	8	7	•	

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# ASSOCIATED AIR PUBLICATIONS

Canberra T Mk 17 Flight Reference Ca	ards	AP 101B-0417-14
Canberra T Mk 17 Operating Data Ma	nual	AP 101B-0417-16
Canberra T Mk 17 Aircraft, General ar	nd	
Technical Information		AP 101B-0417-1A & 1B
Avon Mk 1 Aero-Engine and Avon Mk	k 10201	
Engine Change Units		AP 102C-1522-1

# **LEADING PARTICULARS**

## Principal Dimensions

							Feet	Inches
Span without wing-	tip tanks/poc	ls		•••	•••	•••	64	0
Span with wing-tip	tanks/pods			•••			65	6
Length overall					•••		67	4
Height to top of fin	•••		•••		•••		15	7
Height to top of canopy (unladen aircraft)				•••		•••	8	8

### Undercarriage

### Mainwheel Units (two)

Type	***			Single wheel, inwards retracting
Shock abs	orber	•••		Oleo pneumatic
Nitrogen pressure				Refer to AP 101B-0400 -5A2
Fluid		•••	•••	OM-15 (NATO H-515)
Capacity	<i>/</i>			12 pints
Tyre press	sure	•••		Refer to AP 101B- 0400A -5A2
Brakes				sales services services
Pressure at reducing valve			alve	2500 +0 PSI
Pressure at brakes				1500 +150 DCI

### Nosewheel Unit

Туре		•••		Twin wheel, non-steerable castering, rearward retracting
Shock al	bsorber	•••		Levered suspension, liquid spring
Pressur	re (whee	ls off g	round)	1500 PSI
Fluid	•••	•••	•••	OM-15 (NATO H-515)
Capaci	ty	•••		1.5 pints
Tyre pre	ssure	•••		Refer to AP 101B - 0400A -5A2

## **Hydraulic System**

	Fluid		OM-15 (NATO H-515)		
	Capacity of system		31 pints (approximately)		
1	Capacity of tank	•••	2 gallons		
	Pumps (two)		Lockheed Mk 9		
	Accumulator charging gas		Nitrogen		
	Thermal relief valve setting	g	3450 ±100 PSI (See Part 1, Chap 3)		

# LIST OF ABBREVIATIONS USED IN THE TEXT

		Y Y Y	
		let pipe temperature	Tqt
Variation setting control	AZC	Instrument landing system	ILS
range		Identification friend or foe	IEE
VHF omni-directional radio	AOK	Indicated airspeed	SYI
Very low frequency	ALF		
Very high frequency	THV	Hertz (cycles per second)	zH
Visual committal height	ACH	High tension	TH
20 F	11011	High pressure	dH
tinu gni		High frequency	НЕ
Undervolt and phase sequenc-	DALO	2 7	٠
Ultra high frequency	UHF	Ground position indicator	CbI
J 1-11	amı	Ground power circuit breaker	GPB
Transmitter/receiver	A/T		Qub
True airspeed	2AT ⊕T	Flight Reference Cards	FRC
bearrain surT	SAT	Fault free fire detection	EEED
MUNICIPALITY THE AMOUNTS	MCC	moitontab and and ilund	uaaa
Secondary surveillance radar	SSR	DIE DO COL THOU THE	VOT
aminin tod groups cast	TAT IN	Engine out allowance	EOA
Revolutions per minute	RPM	speeds	OGW
Radio magnetic indicator	RMI	Emergency maximum braking	EWB2
Rotary hydraulic arresting gear	RHAG	Electrical control panel	ECP
8		Electronics countermeasures	ECM
Quick release fitting	ОКБ	50/4 (1739) - \$25 (1703) (1759)	
		Direct vision	DΛ
Purpose use arresting gear	<b>DAU9</b>	Direct current	DC
Pounds per square inch	ISd	Drilt angle groundspeed	DAG
Operating Data Manual	ODW	Chain arresting gear	CHAG
Onni-bearing selector	OBS	Centre of gravity	CG
	buo		
Normal maximum braking speed	NWB2	Barostatic time release unit	BTRU
		Breech time delay unit	DOLLA
Magnetic indicator	IM	Busbar tie circuit breaker	BTB
Megahentz	ZHW	Barometric pressure control	BbC
♦ lectrical panel	♦ WEb	Bromochlorodifluoromethane	BCL
Масћ питрег	M		
en von Torrestation (Table ) To Torrestation (Table )		All up weight	WUA
row bicasmic	Γb	Above ground level	YCF
bound	qı	Air electronics officer	VEO
3 3	350	Automatic direction finding	ADF
Kilowatt	ΚM	Acceleration control unit	NOA
Kilovolt amps	ΚΛΨ	Alternating current	YC

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# LIST OF ABBREVIATIONS USED IN THE TEXT

AC ACU ADF	Alternating current Acceleration control unit Automatic direction finding	kVA kW	Kilovolt amps Kilowatt
AEO AGL AUW	Air electronics officer Above ground level All up weight	lb LP	Pound Low pressure
BCF BPC BTB BTDU	Bromochlorodifluoromethane Barometric pressure control Busbar tie circuit breaker Breech time delay unit	M MEP MHz MI	Mach number Main electrical panel Megahertz Magnetic indicator
BTRU	Barostatic time release unit	NMBS	Normal maximum braking speed
CG CH <b>AG</b>	Centre of gravity Chain arresting gear	OBS ODM	Omni-bearing selector Operating Data Manual
DAGS DC DV	Drift angle groundspeed Direct current Direct vision	PSI PUAG	Pounds per square inch Purpose use arresting gear
ECN4		QRF	Quick release fitting
ECM ECP EMBS	Electronics countermeasures Electrical control panel Emergency maximum braking speed Engine out allowance	RHAG RMI RPM	Rotary hydraulic arresting gear Radio magnetic indicator Revolutions per minute
LOA	Englic out anowance	SSR	Secondary surveillance radar
FFFD FRC	Fault free fire detection Flight Reference Cards	TAS T/R	True airspeed Transmitter/receiver
GPB	Ground power circuit breaker		
GPI	Ground position indicator	UHF UVP <b>SU</b>	Ultra high frequency Undervolt and phase sequenc-
HF	High frequency		ing unit
HP	High pressure		
HT	High tension	VCH	Visual committal height
Hz	Hertz (cycles per second)	VHF VLF	Very high frequency Very low frequency
IAS	Indicated airspeed	VOR	VHF omni-directional radio
IFF	Identification friend or foe	011121174	range
ILS	Instrument landing system	VSC	Variation setting control
JPT	Jet pipe temperature		

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