

PROPULSION

SERVICING PROCEDURES
FUNCTIONAL CHECKS AND TESTS
HUNTER ALL MARKS
AMENDED TO INCLUDE MK 9 CONTENT

AP101B-1300-5A3B

Sect 2

Chap 4

List of Contents

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PROPULSION SP 211A (1) (1 to 13)				SERVICING PROCEDURES FUNCTIONAL CHECKS AND TESTS HUNTER ALL MARKS		AP101B-1300-5A3B Sect 2 Chap 4		SERVICING RECORD RAF Form 2988B			
<u>ENGINE STARTING, TESTING AND STOPPING</u> Applicable only to Mk6, 6A and 9 A/C.								Aircraft Ser No: Date:			
SAFETY AND SERVICING NOTES ARE TO BE COMPLIED WITH THROUGHOUT THE WORK DETAILED ON THIS CARD						TRADESMAN		Brief Details of Suspected Defect and SNOW When Applicable		SUPERVISOR	
TRADE	M/HRS	TRADE	M/HRS	ASSOCIATED PROCEDURE CARDS		MAN HRS 1	INITS & TDM 2	3	MAN HRS 4	INITS & TDM 5	
Prop.				SP 526, SP CSDE/Hunter/Vib./1							
<u>Special Tools and Equipment:</u> Air Intake Safety Guards 26FX/95413 (Port), 26FX/95414 (Stbd). Accelerometer and Bracket. Test Resistor 5CZ/4337552.											
NB: During this Procedure ground running of engine is to be kept to a minimum and the checks detailed are only to be carried out when required.											
BLOCK 1 1. <u>Preparation</u> 1.1 Engine oil sump. } 1.2 Accessory gearbox oil sump. } Replenish as necessary. (Oil OX-38). 1.3 Turret drive oil sump. } 1.4 Aircraft fuel system. Replenish as necessary. (5A2, Sect 2). 1.5 IPN tank. Replenish as necessary. 1.6 Brake system pressure. Ensure 1500 lbf/in2 minimum. Sub-item 1.7 is applicable only if ECU has been newly installed, inhibited, or if air is in the fuel system. 1.7 Engine fuel system. Bleed.											

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SP 211A (2)

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BLOCK 1

PROPULSION

2. Preparation

- | | | |
|-----|----------------------------|---|
| 2.1 | Aircraft. | Position and secure for ground running. |
| 2.2 | Control locks. | } Remove. |
| 2.3 | Air intake blanks. | |
| 2.4 | Air intakes. | Ensure clear. |
| 2.5 | Air intake safety guards. | Fit. |
| 2.6 | Accelerometer and bracket. | Assemble. (SP CSDE/Hunter/Vib/1.) |

BLOCK 2

PROPULSION NCO

3. Pre-Start Checks

- | | | |
|-----|---------------------------------|----------------------------------|
| 3.1 | Undercarriage selector. | Ensure set to 'DOWN' and locked. |
| 3.2 | Brake lever. | Ensure locked 'ON'. |
| 3.3 | Flap selector. | Ensure set to 'UP'. |
| 3.4 | Top temperature control switch. | Ensure set to 'ON'. |
| 3.5 | Cabin pressurization. | Ensure set to 'OFF'. |
| 3.6 | Cabin temperature. | } Ensure set to 'AUTO'. |
| 3.7 | Flood flow. | |
| 3.8 | Power controls. | Ensure set to 'OFF'. |

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MAN HRS 1	INITS & TDM 2		MAN HRS 4	INITS & TDM 5

BLOCK 1

PROPULSION NCO

3. Pre-Start Checks

- 3.9 Battery master switch. }
3.10 Engine master switch. } Set to 'ON'.
3.11 Fire warning test (i) Press.
button. (ii) Ensure fire warning illuminates.
3.12 Relight system. (i) Operate.
 (ii) Ensure intermittent and audible.
3.13 Anti-icing system switch. Ensure closed and indicator showing 'SHUT'

BLOCK 2

PROPULSION NCO

- NB: If engine fails to start, close throttle and after one minute has elapsed a further attempt may be made.
After three attempts to start, starter is to be allowed to cool before any further attempt is made.
4. Starting
- 4.1 Ignition switch. }
4.2 LP fuel cock. }
4.3 Tank pump switches. } Set to 'ON'.
4.4 Throttle/HP cock. Set to 1 inch open.
- NB: During sub-item 4.5 ensure engine lights up within 4 to 8 seconds and that the JPT does not exceed 560°C.
- 4.5 Starter button. Press for 2 seconds and release.

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						MAN HRS 1	INITS & TDM 2	3	MAN HRS 4	INITS & TDM 5	
BLOCK 1 PROPULSION NCO											
4. <u>Starting cont.</u>											
4.6 Idling rev/min. Ensure 2500 PLUS 200 MINUS NIL.											
4.7 Oil pressure. (At 3600 rpm). Ensure indicated. (minimum 20 lbf/in2).											
4.8 Hydraulic pressure }											
warning light. }											
4.9 Generator warning light. }											
BLOCK 2 PROPULSION NCO											
5. <u>Hydraulics</u>											
5.1 Throttle. Set to obtain 4500 rev/min.											
5.2 Engine. Ensure smooth running.											
5.3 Hydraulic pressure Ensure indicates 2850 PLUS 150 MINUS											
gauge. 50 lbf/in2.											
5.4 Hood control clutch. Set to 'IN'.											
5.5 Hood selector switch. Set to close.											
5.6 Hood seal. Ensure inflated.											
5.7 Hood selector switch. Set to open.											
5.8 Throttle. Set to 'IDLING'.											
5.9 Hood control clutch. Set to 'NEUTRAL'.											

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BLOCK 1 PROPULSION NCO

6. Air Bleed Valve Check

NB: Variation in ambient air temperature will affect air bleed valve and
IGV ram settings.

- 6.1 Outside air temperature. Obtain outside air temperature and
record in degrees centigrade.
- 6.2 Anti-icing system switch. Ensure set to shut.
- 6.3 Engine.
- (i) Accelerate smoothly from idling
RPM to ensure PV ram and bleed
valve are functioning.
 - (ii) Set engine at 7600 RPM and allow
to soak for 3 minutes.
 - (iii) Exercise engine throughout PV ram
range twice then set engine to
7000 RPM.
 - (iv) Using a throttle rate of approx.
25 RPM per second. smoothly
decelerate until bleed valve opens.
Record RPM for reference.
 - (v) Reduce engine speed to 6000 RPM
then smoothly accelerate at 25 RPM
per second until bleed valve closes.
Record RPM for reference.
- NB: The bleed valve should operate
on acceleration at 6700 + 50 RPM
at 15°C. (Refer to Figure 1 for
temperature correction).

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BLOCK 1 PROPULSION NCO 6. <u>Air Bleed Valve Check cont.</u> 6.3 Engine. <div style="margin-left: 200px;"> (vi) Ensure bleed valve operating rev/min are within the limits of Fig 1, and that the engine is free from surge, bleed valve 'THUMP' or burbling during a rapid deceleration from governed speed. Record RPM at which bleed valve opens on rapid deceleration. (vii) Record BV setting RPM and OAT on Form 707B or as appropriate and on STC Form 1222 (STC UNITS ONLY). </div>											
BLOCK 2 PROPULSION NCO 7. <u>Intake Guide Vane Ram Check</u> 7.1 Outside air temperature. Obtain outside air temperature and record in degrees centigrade. 7.2 Anti-icing system switch. Ensure set to 'SHUT'. 7.3 Engine. <div style="margin-left: 200px;"> (i) Accelerate smoothly from idling RPM to ensure PV ram and bleed valve are functioning. (ii) Set engine at 7600 RPM and allow to soak for 3 minutes. </div>											

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BLOCK 1 PROPULSION NCO

7. Intake Guide Vane Ram Check cont.

- 7.3 Engine. cont.
- (iii) Exercise engine throughout the PV ram range twice.
 - (iv) Slowly reduce RPM and ensure IGV ram leaves the -0° position within limits (Fig 1).
 - (v) Reduce engine speed to 6000 RPM then slowly increase RPM and ensure the IGV ram leaves $+25^{\circ}$ position within limits (Fig 1).

NB: Engine operation will normally be more satisfactory if PV ram is set near top of 25° limit; this will also reduce the requirement to re-adjust when checking bleed valve operation (Item 6).

- (vi) After change or adjustment of IGV ram it is essential to repeat the air bleed valve check detailed in Item 6.

BLOCK 2 PROPULSION NCO

8. Anti-icing Check

- 8.1 Throttle. Set to obtain 7600 rev/min.
- 8.2 Anti-icing system switch. Set to 'OPEN'.
- 8.3 Rev/min. Note a change.

9.6 Top temperature control switch. Set to 'ON'.

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						MAN HRS 1	INITS & TDM 2	3	MAN HRS 4	INITS & TDM 5	
BLOCK 1 PROPULSION NCO											
10. <u>Acceleration Check</u> NB: Correct acceleration time by adding 1 second for every 10°C temperature increase above +15°C and subtract 1 second for every 10°C temperature below +15°C.											
10.1 Outside air temperature. Note.											
10.2 Top temperature control switch. Ensure set to 'ON'.											
10.3 Anti-icing system switch. Ensure set to 'OFF'.											
10.4 Throttle. Open rapidly to fully open position.											
10.5 Engine rev/min. Check time taken to accelerate from idling to within 50 rev/min of governed rev/min. (Ensure within 9 to 14 seconds).											
10.6 Throttle. (i) Set to obtain 4500 rev/min. (ii) Open rapidly to fully open position.											
10.7 Engine. Check time to accelerate to within 50 rev/min of governed rev/min. (Ensure within 3 to 5 seconds).											
10.8 Throttle. Close to 'IDLING' position.											

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BLOCK 1

PROPULSION NCO

11. Top Temperature Control Check

11.1 Temperature resistance Remove.
(Normal).

11.2 Temperature resistance Fit.
(600 degrees C).

11.3 Top temperature control Operate and hold.
micro switch (Nosewheel
compartment)

11.4 Top temperature control Ensure set to 'ON'.
switch.

11.5 Throttle. Move to open position until 600 degrees
C JPT is reached. No increase in JPT
should be recorded on further opening
of throttle.

11.6 Throttle. Close to 'IDLING' position.

11.7 Top temperature control Set to 'OFF'.
switch.

11.8 Throttle. Move to fully open position.

11.9 Rev/min. }
11.10 JPT. } Ensure normal readings.

11.11 Top temperature control Set to 'ON'.
switch.

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BLOCK 1 PROPULSION NCO

11. Top Temperature Control Check cont.
- 11.12 Throttle. Close to 'IDLING' position.
- 11.13 Temperature resistance Remove.
(600 degrees C).
- 11.14 Temperature resistance Refit.
(Normal).
- 11.15 Top temperature control Release.
micro switch.

BLOCK 2 PROPULSION NCO

12. Checks Before Engine Shut Down
- 12.1 Vibration survey. Carry out using technique CSDE/Hunter/VIB/1
on following occasions:
a. On engine installation after repair,
recondition, or in-service repair.
b. On defect (installed).
c. On test bed.
- NB: A physical record is to be kept with engine documents.
- 12.2 Flaps. }
12.3 Air brakes. } Ensure correct operation.
12.4 Fuel transfer. }
- 12.5 Power controls. Ensure engagement.

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BLOCK 1

PROPULSION NCO

13. Stopping

13.1 Idling rev/min.

Note.

13.2 Throttle/HP cock control.

Set to 'HP COCK OFF'.

13.3 Engine master switch.

Set to 'OFF'.

13.4 Engine.

Ensure runs down freely (Minimum time
75 seconds).

13.5 Tank pump switches.

Set to 'OFF'.

13.6 Generator warning light.

Ensure illuminates at 1500 to 1700 RPM.

13.7 Manifold drain.

Ensure fuel discharges on engine run
down.

13.8 LP fuel cock.

13.9 Battery master switch.

} Set to 'OFF'.

BLOCK 2

PROPULSION

14. Completion

14.1 Air intake safety guards.

Remove.

14.2 Accelerometer and bracket.

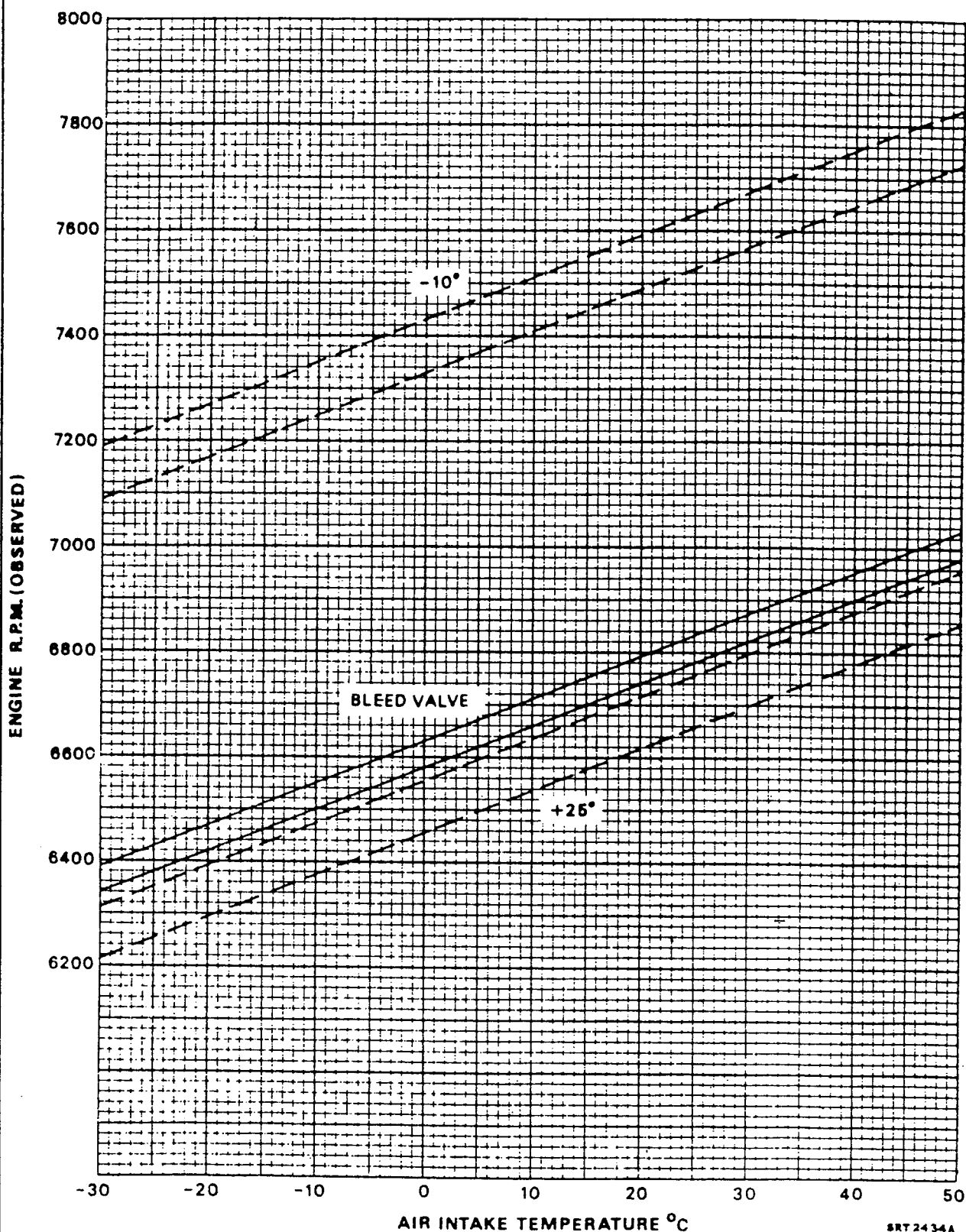
Remove.

14.3 Air intake blanks.

Refit.

14.4 Top temperature control
switch.

(i) Ensure set to 'ON'.
(ii) Fit tell-tale wire.



8RT 24 34A

Fig.1 I.G.V. ram and air bleed valve setting curve

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<u>ENGINE STARTER SYSTEM - COMBINED FUEL PRIMING AND DRAINAGE CHECK</u> Applicable only to Mk6, 6A and 9 aircraft.								Aircraft Ser No: Date:					
SAFETY AND SERVICING NOTES ARE TO BE COMPLIED WITH THROUGHOUT THE WORK DETAILED ON THIS CARD								TRADESMAN		Brief Details of Suspected Defect and SNOW When Applicable		SUPERVISOR	
TRADE Prop. Elect		M/HRS		TRADE M/HRS		ASSOCIATED PROCEDURE CARDS		MAN HRS 1	INITS & TDM 2	3	MAN HRS 4	INITS & TDM 5	
<u>Special Tools and Equipment:</u> Calibrated Container.													
NB 1 At least one hour is to elapse after a 'live' start before this check is carried out. NB 2 This procedure is applicable whenever the starter fuel system is disturbed or on initial installation of starting system and is to be carried out BEFORE ANY ATTEMPT IS MADE TO START THE ENGINE. NB 3 The quantity of fuel from all drainage checks is to be recorded in the F700 with other IPN starter installation details. NB 4 Where fuel drainage does not follow the correct pattern, in that drainage stops early, or continues to drip steadily or is less than specified minimum, a blocked nozzle plate drain hole is indicated. This is a SERIOUS HAZARD and a start is not to be attempted until drain hole has been cleared. NB 5 If the fuel pump/air blower motor unit continues to run on after the air cycle is complete, an excessive amount of fuel will be discharged into, or through, the combustion chamber and may cause combustion products to be dislodged and block the nozzle plate holes. Should this situation arise, the nozzle plate must be examined and cleared as detailed in Item 6.													

2.3 External d.c. power supply.

- (i) Connect.
- (ii) Switch on.
- (iii) Adjust input to between 23 and 26 volt d.c. on load.

NB: Ensure voltage is maintained within limits during the fuel priming. High or low voltages will give incorrect fuel drainage.

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						MAN HRS 1	INITS & TDM 2		MAN HRS 4	INITS & TDM 5
BLOCK 1 PROPULSION										
3. <u>Fuel Priming</u>										
3.1 Engine master switch. Set to 'ON'.										
3.2 Starter button. Press and release.										
3.3 Starter exhaust. (i) Ensure air flows for 2 to 3 secs. (ii) Look for fuel drainage. (iii) Ensure that any fuel drainage is accompanied by a brief flash of the test lamp.										
3.4 Starter control box motor. Ensure stops approximately 18 seconds after operation of starter button.										
3.5 System. Repeat operations detailed in Sub-items 3.2 and 3.4 at one minute intervals until fuel flows from starter exhaust. NB: If more than four runs are required, the nozzle plate drain hole is to be examined for obstruction.										
BLOCK 2 PROPULSION										
Item 4 is to be carried out during Fuel Drainage Check.										
4. <u>Leak Check</u>										
4.1 System. Look for leaks.										

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BLOCK 1

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5. Fuel Drainage Check

5.1 Starter button.

Press and release.

5.2 Starter exhaust.

Measure and record quantity of fuel
drainage (80 to 110 cc).
NB: Approximately three quarters of
fuel drainage should occur in first
minute, remainder during next minute.

5.3 Engine master switch.

Set to 'OFF'.

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Item 6 is applicable only if more than four runs are required in Item 3
or if fuel drainage pattern is incorrect.

6. Combustion Chamber Check

6.1 Safety disc holder.

Remove.

6.2 Combustion chamber.

Ensure free from accumulated fuel.

6.3 Nozzle plate drain hole.

Ensure unobstructed.

6.4 Safety disc holder.

- (i) Ensure disc intact.
- (ii) Refit. (Torque 384 lbf in.).
- (iii) Lock.

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TRADE Prop. Elect	M/HRS	TRADE	M/HRS	ASSOCIATED PROCEDURE CARDS									
<u>Special Tools and Equipment:</u> Calibrated Container.													
NB 1 At least one hour is to elapse after a 'live' start before this check is carried out. NB 2 This Procedure is applicable whenever any part of the starter electrical system has been disturbed and is to be carried out BEFORE ANY ATTEMPT IS MADE TO START THE ENGINE. NB 3 The quantity of fuel from all the drainage checks is to be recorded in F700 with other IPN starter installation details. NB 4 Where fuel drainage does not follow the correct pattern, in that drainage stops early, or continues to drip steadily, or it is less than specified minimum, a blocked nozzle plate drain hole is indicated. THIS IS A SERIOUS HAZARD and a start is not to be attempted until drain hole has been cleared. NB 5 If the fuel pump/air blower motor unit continues to run on after the air cycle is complete; an excessive amount of fuel will be discharged into the combustion chamber and may cause combustion products to be dislodged and block the nozzle plate holes. Should this situation arise, the nozzle plate must be examined and cleared as detailed in Item 5. NB 6 Following a successful combustion cycle where more than two fuel drainage checks are required during servicing, the combustion chamber must be examined for fuel content, as detailed in Item 5, as the last operation just prior to an attempt to start.													

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NB 6 Cont.

This limit permits three subsequent attempts to start by the pilot.

NB 7 Not more than five starter operations, comprising fuel drainage checks and unsuccessful attempts to start (where combustion does not occur) may be made between successful combustion cycles (or starts). Where more than 5 Starter operations are required during servicing, the combustion chamber must be examined for any blockage of drain hole and nozzles and cleared as detailed in Item 5.

NB 8 An unsuccessful attempt to start, where combustion does not occur, has the same effect on the IPN system, and passes the same amount of fuel, as a simulated start, ie a priming run or a fuel drainage check.

BLOCK 1

PROPULSION

1. Preparation

1.1 Engine master switch.)

1.2 Ignition switch.)

Ensure set to 'OFF'.

1.3 IPN fuel tank.

Replenish.

1.4 Calibrated container.

Position under starter exhaust outlet.

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3

SUPERVISOR

MAN
HRS
1

INITS
& TDM
2

MAN
HRS
4

INITS
& TDM
5

BLOCK 1

ELECTRICAL

2. Preparation

- 2.1 HF ignition unit LT lead. (i) Disconnect.
(ii) Connect a suitable test lamp to the plug to provide an indication of fuel pressure in the atomiser fuel line.
- 2.2 Aircraft batteries. Ensure fully charged.
- 2.3 External d.c. power supply. (i) Connect.
(ii) Switch on.
(iii) Adjust input to between 23 and 26 volts d.c. on load.
NB: Ensure voltage is maintained within limits during the fuel drainage. High or low voltage will give incorrect fuel drainage.

BLOCK 2

PROPULSION

3. Fuel Drainage Check

- 3.1 Engine master switch. Set to 'ON'.
- 3.2 Starter button. Press and release.

Look for leaks.

PROPULSION
SP 216 (5)

SERVICING PROCEDURES
FUNCTIONAL CHECKS AND TESTS
HUNTER ALL MARKS

AP101B-1300-5A3B
Sect 2
Chap 4

SERVICING RECORD

RAF Form 2988B

Aircraft Ser No:
Date:

ENGINE STARTER SYSTEM - FUEL DRAINAGE CHECK

SAFETY AND SERVICING NOTES ARE TO BE COMPLIED WITH THROUGHOUT THE WORK
DETAILED ON THIS CARD

BLOCK 1 PROPULSION

Item 5 is applicable only if drainage pattern is incorrect.

5. Combustion Chamber Check

- | | | |
|-----|---------------------|---|
| 5.1 | Safety disc holder. | Remove. |
| 5.2 | Combustion chamber. | Ensure free from accumulated fuel. |
| 5.3 | Nozzle plate drain. | Ensure unobstructed. |
| 5.4 | Safety disc holder. | (i) Ensure disc intact.
(ii) Refit. (Torque 384 lbf in.).
(iii) Lock. |

BLOCK 2

ELECTRICAL

6. Completion

- 6.1 External d.c. power supply. (i) Switch off.
(ii) Disconnect.
- 6.2 HF ignition unit LT lead. (i) Remove test lamp.
(ii) Reconnect.

BLOCK 3 PROPULSION

7. Completion

- | | | |
|-----|-----------------------|----------------------------|
| 7.1 | Starter exhaust. | Remove all traces of fuel. |
| 7.2 | Calibrated container. | Recover. |
| 7.3 | <u>IPN fuel tank.</u> | Replenish. |

PROPULSION SP217 (1) (1 to 2)				SERVICING PROCEDURES FUNCTIONAL CHECKS AND TESTS HUNTER ALL MARKS		AP101B-1300-5A3B Sect 2 Chap 4		SERVICING RECORD RAF Form 29888 Aircraft Ser No: Date:					
STARTER FUEL TANK FILLER DRAIN PIPE LEAK CHECK. Mk 6, 6A and 9 A/C only.								TRADESMAN		Brief Details of Suspected Defect and SNOW When Applicable		SUPERVISOR	
SAFETY AND SERVICING NOTES ARE TO BE COMPLIED WITH THROUGHOUT THE WORK DETAILED ON THIS CARD								MAN HRS 1	INITS & TDM 2	3		MAN HRS 4	INITS & TDM 5
TRADE Prop.	M/HRS	TRADE	M/HRS	ASSOCIATED PROCEDURE CARDS									
<u>Special Tools and Equipment:</u> Calibrated container - 14B/3498 (2 off).													
BLOCK 1 ENGINES 1. <u>Preparation</u> 1.1 Starter fuel tank filler access panel. Remove. 1.2 Filler drain pipe. Ensure unobstructed.													
BLOCK 2 ENGINES 2. <u>Check</u> 2.1 Filler drain pipe. <ul style="list-style-type: none"> (i) Using calibrated container, pour 300 Cm³ of warm water into drain inlet. (ii) Using second calibrated container, collect and note quantity of water issuing from drain outlet. NB During third check in operation (iii), if input exceeds output by more than 10 Cm³, leak is to be located and rectified. 													
Continued													

PROPULSION
SP217 (2)

SERVICING PROCEDURES
FUNCTIONAL CHECKS AND TESTS
HUNTER ALL MARKS

AP101B-1300-5A3B
Sect 2
Chap 4

SERVICING RECORD

RAF Form 2988B

Aircraft Ser No:
Date:

STARTER FUEL TANK FILLER DRAIN PIPE LEAK CHECK.

SAFETY AND SERVICING NOTES ARE TO BE COMPLIED WITH THROUGHOUT THE WORK
DETAILED ON THIS CARD

TRADESMAN

Brief Details of
Suspected Defect and
SNOW
When Applicable
3

SUPERVISOR

MAN
HRS
1

INITS
& TDM
2

MAN
HRS
4

INITS
& TDM
5

BLOCK 1

ENGINES

2. Check cont.

2.1 (iii) Repeat operations (i) and (ii)
twice.

BLOCK 2

ENGINES

3. Completion

3.1 Starter fuel tank filler Dry.
recess.

3.2 Starter fuel tank filler Refit.
access panel.

PROPULSION SP 218 (1) (1 to 5)				SERVICING PROCEDURES FUNCTIONAL CHECKS AND TESTS HUNTER ALL MARKS Amended to include Mk 9 content.				AP101B-1300-5A3B Sect 2 Chap 4				SERVICING RECORD Aircraft Ser No: Date:				RAF Form 2988B					
<u>FUEL FLOW TEST</u>												TRADESMAN		Brief Details of Suspected Defect and SNOW When Applicable		SUPERVISOR					
SAFETY AND SERVICING NOTES ARE TO BE COMPLIED WITH THROUGHOUT THE WORK DETAILED ON THIS CARD												MAN HRS 1	INITS & TDM 2	3	MAN HRS 4	INITS & TDM 5					
TRADE	M/HRS	TRADE	M/HRS	ASSOCIATED PROCEDURE CARDS																	
Prop.																					
<u>Special Tools and Equipment:</u> Locally Manufactured Hose Assembly. Locally Manufactured Calibrated Fuel Container (Minimum Capacity 10 Gallons).																					
BLOCK 1 PROPULSION 1. <u>Preparation</u> 1.1 Front fuselage fuel tanks. Ensure a minimum of 650 lbs of fuel in each. 1.2 Battery master switch. Set to 'OFF'. 1.3 LP fuel cock. Set to 'OFF'. 1.4 Engine master switch. Set to 'OFF'.																					
BLOCK 2 PROPULSION This Block is applicable only to Mk6, 6A and 9 aircraft. 2. <u>Preparation</u> 2.1 Fuel delivery pipe. (i) Disconnect at flexible couplings. (Pt No C201525/29). (ii) Remove. (iii) Fit locally manufactured pipe assembly to fuel pipe assembly (Pt No C201525/28) at frame 32. (iv) Position calibrated container.																					

PROPULSION
SP 218 (2)

SERVICING PROCEDURES
FUNCTIONAL CHECKS AND TESTS
HUNTER ALL MARKS

AP101B-1300-5A3B
Sect 2
Chap 4

SERVICING RECORD

RAF Form 29888

Aircraft Ser No:
Date:

FUEL FLOW TEST

SAFETY AND SERVICING NOTES ARE TO BE COMPLIED WITH THROUGHOUT THE WORK
DETAILED ON THIS CARD

TRADESMAN		Brief Details of Suspected Defect and SNOW When Applicable 3	SUPERVISOR	
MAN HRS 1	INITS & TDM 2		MAN HRS 4	INITS & TDM 5

BLOCK 1

PROPULSION

This Block is applicable only to Mk7, 7A and 8B aircraft.

3. Preparation

- 3.1 Fuel delivery pipe. (Pt No C18805/10).
- (i) Disconnect at flexible coupling.
 - (ii) Remove.
 - (iii) Fit locally manufactured pipe assembly to fuel pipe assembly (Pt No A212114).
 - (iv) Position locally manufactured calibrated container.

BLOCK 2

PROPULSION

This Block is applicable only to Mk6, 6A and 9 aircraft.

4. Test

- 4.1 Battery master switch. Set to 'ON'.
- 4.2 Engine master switch. Set to 'ON'.
- 4.3 Port fuel pump. Set to 'ON'.

Note: Ensure locally manufactured hose assembly and calibrated container are in position.

- 4.4 LP cock.
- (i) Open for 5 seconds.
 - (ii) Close.
 - (iii) Check that the minimum quantity of fuel discharged into the calibrated container is 4.0625 gal.

PROPULSION SP 218 (3)		SERVICING PROCEDURES FUNCTIONAL CHECKS AND TESTS HUNTER ALL MARKS		AP101B-1300-5A3B Sect 2 Chap 4		SERVICING RECORD RAF Form 2988B					
FUEL FLOW TEST						Aircraft Ser No: Date:					
SAFETY AND SERVICING NOTES ARE TO BE COMPLIED WITH THROUGHOUT THE WORK DETAILED ON THIS CARD						TRADESMAN		Brief Details of Suspected Defect and SNOW When Applicable		SUPERVISOR	
BLOCK 1 PROPULSION						MAN HRS 1	INITS & TDM 2	3		MAN HRS 4	INITS & TDM 5
4. <u>Test</u> cont.											
4.5 Port fuel pump. Set to 'OFF'.											
4.6 Starboard fuel pump. Set to 'ON'.											
4.7 LP cock.											
(i) Open for 5 seconds.											
(ii) Close.											
(iii) Check that the minimum quantity of fuel discharged into calibrated container is 4.0625 gallons.											
4.8 Starboard fuel pump. Set to 'OFF'.											
4.9 Engine master switch. Set to 'OFF'.											
4.10 Battery master switch. Set to 'OFF'.											
BLOCK 2 PROPULSION											
This Block is applicable only to Mk7, 7A and 8B aircraft.											
5. <u>Test</u>											
5.1 Battery master switch. Set to 'ON'.											
5.2 Engine master switch. Set to 'ON'.											
5.3 Port fuel pump. Set to 'ON'.											
NOTE: Ensure that the locally manufactured hose assembly and calibrated container is in position.											

Set to 'OFF'.

PROPULSION SP 218 (5)		SERVICING PROCEDURES FUNCTIONAL CHECKS AND TESTS HUNTER ALL MARKS		AP101B-1300-5A3B Sect 2 Chap 4		SERVICING RECORD RAF Form 2988B	
<u>FUEL FLOW TEST</u>				Aircraft Ser No: Date:			
SAFETY AND SERVICING NOTES ARE TO BE COMPLIED WITH THROUGHOUT THE WORK DETAILED ON THIS CARD		TRADESMAN		Brief Details of Suspected Defect and SNOW When Applicable 3		SUPERVISOR	
		MAN HRS 1	INITS & TDM 2			MAN HRS 4	INITS & TDM 5
BLOCK 1 PROPULSION This Block is applicable only to Mk6, 6A and 9 aircraft. 6. <u>Completion</u> 6.1 Locally manufactured hose Remove. assembly. 6.2 Fuel delivery pipe. Refit. (Pt No 201525/29).							
BLOCK 2 PROPULSION This Block is applicable only to Mk7, 7A and 8B aircraft. 7. <u>Completion</u> 7.1 Locally manufactured hose Remove. assembly. 7.2 Fuel delivery pipe. Refit. (Pt No 18805/10).							
BLOCK 3 PROPULSION 8. <u>Completion</u> 8.1 ECU fuel system. Bleed.							

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