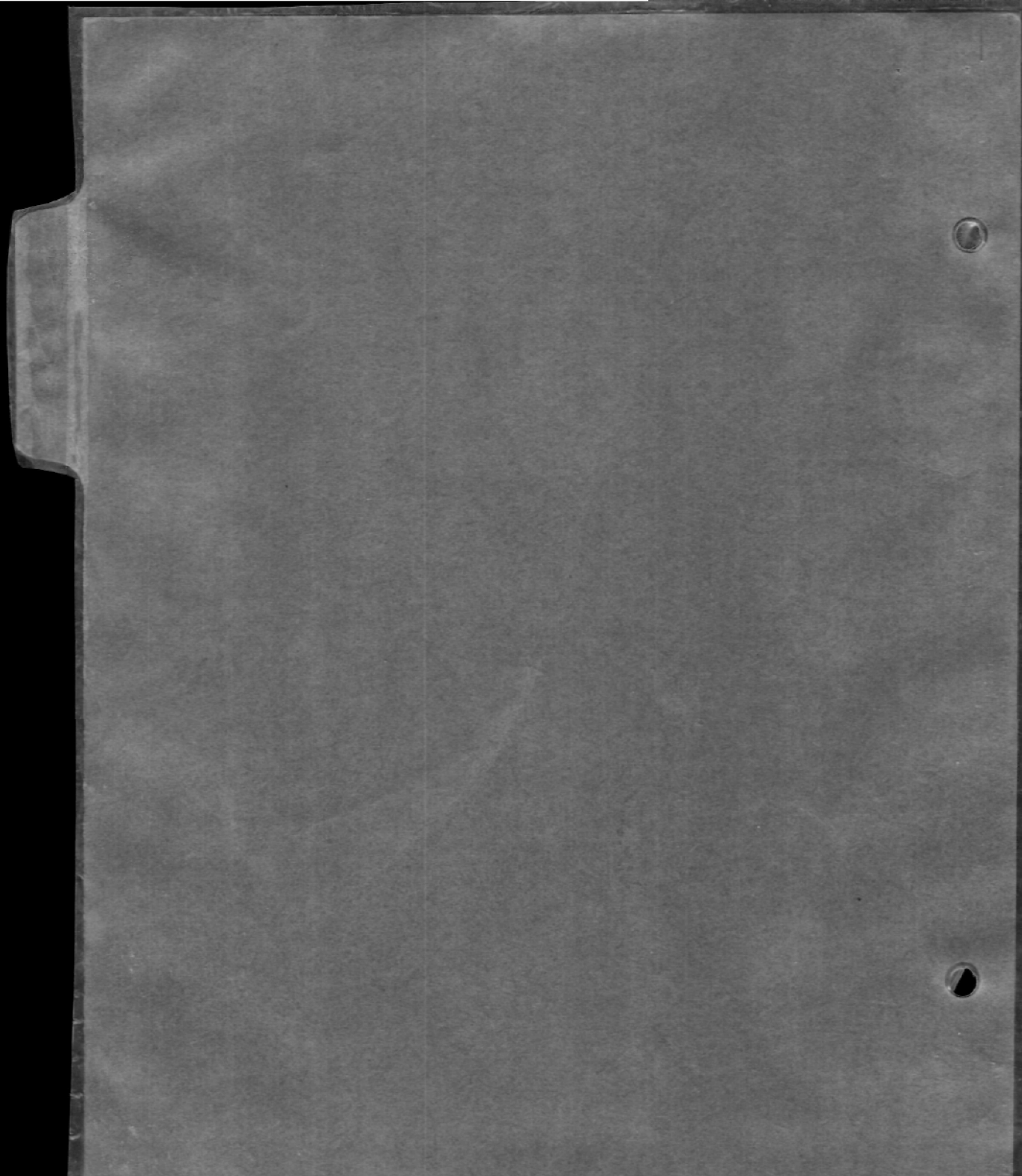


CHAPTER 2

CONTROLS

(July, 58)

CHAPTER 2
CONTROLS



Chapter 2

CONTROLS

List of Illustrations

Controls in cabin	Fig. 1
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General

1. The cabin controls are illustrated in fig.1. Items which will be used for ground handling are annotated, and the method of operation of a particular control is indicated if it is not already obvious.

Access to cabin

2. The cabin hood is electro-hydraulically operated, but provision is also made for manual operation.

Operation from inside

3. The hood is controlled by a three-position spring-loaded centre-off switch marked OPEN and CLOSE, which is mounted, together with the hood lock indicator lamp, at the top of the centre instrument panel. The battery master switch on the port leg panel must be placed in the ground position. When using the hood control switch, it is necessary to retain it in the selected position until the hood completes its full travel. In the event of electrical or hydraulic failure while the aircraft is on the ground and the cabin hood is closed, operation of a handle mounted on the hood jettison release unit, located behind the port seat, will release the hood locks and allow the hood to be raised manually, after first ensuring that the hood jettison release unit is disconnected from the hood jettison gun by withdrawing the quick-release pip-pin in accordance with the instructions given in the warning notice on the handle of the release unit.

Operation from outside

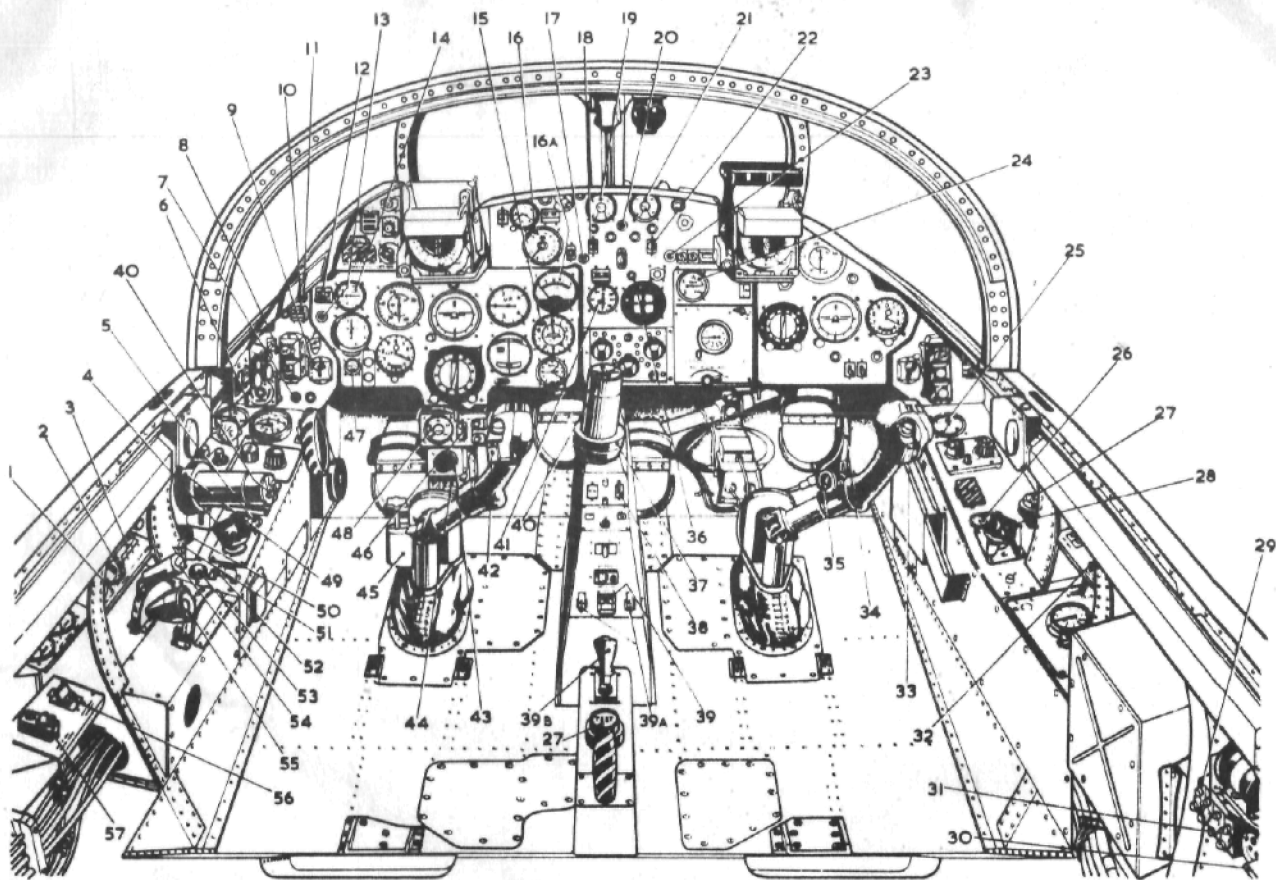
4. A three-position centre-off switch located on the aft face of frame 3 inside the nose-wheel bay is provided for external operation of the hood. It is necessary to ensure that the battery switch on the port leg panel in the cabin is in the ground position and to hold the control switch in the selected position until the hood has completed its full travel.

WARNING

Before using the switch, ensure that the hood is clear of obstruction and that the jury strut, if in use, has been removed.

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Fig. 1. Controls in cabin



KEY TO FIG.1 (CONTROLS IN CABIN)

- | | |
|--|---|
| 1 CABIN TEMPERATURE SELECTOR | 17 TANK PUMP FAILURE WARNING LIGHT - PORT |
| 2 CABIN TEMPERATURE CONTROL | 18 FUEL PUMP SWITCH - PORT |
| 3 CABIN PRESSURE SWITCH | 19 FUEL GAUGE PORT TANKS - |
| 4 AILERON AND RUDDER TRIM TAB CONTROL SWITCH | 20 FUEL LOW PRESSURE WARNING LIGHT |
| Operation in natural sense, i.e. | 21 FUEL GAUGE STARBOARD TANKS |
| aircraft responds to the attitude of the | 22 FUEL PUMP SWITCH - STARBOARD |
| silhouette on the switch knob. | 23 TANK PUMP FAILURE WARNING LIGHT - |
| 5 AILERON AND RUDDER TAB POSITION INDICATORS | STARBOARD |
| 6 UNDERCARRIAGE INDICATOR | 24 OXYGEN CONTENTS GAUGE |
| Light sequence:- | 25 ANTI-G PRESSURE GAUGE |
| GREEN Undercarriage locked down | 26 AILERON AND RUDDER TAB POSITION |
| RED Undercarriage in process of | INDICATORS |
| retraction or extension | 27 ANTI-G TEST BUTTON |
| ALL OFF Undercarriage locked up | Depress to test installation. |
| 7 UNDERCARRIAGE WARNING LIGHT | 28 AILERON AND RUDDER TRIM TAB CONTROL |
| Lights when throttle is approx.1/3 open | SWITCH |
| and the undercarriage is not locked down. | Operation in natural sense, i.e. air- |
| 8 UNDERCARRIAGE CONTROL | craft responds to the attitude of the |
| Interlocking push-switches. Top for UP: | silhouette on the switch knob. |
| bottom for DOWN. Note.-Clockwise | 29 FLIGHT INSTRUMENT INVERTER CIRCUIT |
| rotation of outer wing of UP button | BREAKERS |
| overrides safety lock. | 30 FUEL PUMPS TEST SWITCH |
| 9 FLAP CONTROL | Used for test prior to flight. |
| 10 AILERON POWER CONTROL SWITCH | 31 AMMETER TEST SOCKET |
| Used to disengage power. | Used for test prior to flight. |
| 11 ELEVATOR POWER CONTROL INDICATOR | 32 FIRE WARNING TEST SWITCH |
| Indicates power off. | 33 TAILPLANE INCIDENCE CONTROL (NORMAL) |
| 12 ELEVATOR POWER CONTROL SWITCH | Up to increase tailplane incidence |
| 13 AILERON POWER CONTROL INDICATOR | and down to decrease. |
| Indicates power off. | 34 HYDRAULIC BRAKE CONTROL |
| 14 TAILPLANE POSITION INDICATOR | 35 BRAKE PARKING LOCK |
| 15 TACHOMETER | For temporary parking. |
| 16 EXHAUST GAS THERMOMETER | 36 CABIN PRESSURE WARNING LIGHT |

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KEY TO FIG.1 (CONTROLS IN CABIN) (Contd.)

- 37 THROTTLE
Aft for idling, forward for full throttle.
- 38 ENGINE ANTI-ICING SWITCH
- 39 AIR BRAKE TEST SWITCH
- 39A INVERTOR AND STANDBY TEST SWITCHES
- 39B ANTI-G SUIT CONTROL
- 40 AIR BRAKE CONTROL
- 41 CABIN ALTIMETER
- 42 TAILPLANE INCIDENCE CONTROL (NORMAL)
Up to increase tailplane incidence and down to decrease.
- 43 STARTER PUSH SWITCH
Initiates engine starting cycle.
- 44 ENGINE MASTER SWITCH
Controls flight instruments, tank pumps, fuel pressure and emergency fuel pump circuits.
- 45 AILERON SPRING FEEL UNIT
- 46 IGNITION SWITCH
When off, isolate igniter units.
- 47 FLAP POSITION INDICATOR
- 48 OIL PRESSURE GAUGE
- 49 THROTTLE
Aft for idling, forward for full throttle.
- 50 THROTTLE DAMPER
- 51 JET PIPE TEMPERATURE CONTROLLER
- 52 FUEL PUMP ISOLATING SWITCH
Normally wire-locked - broken wire indicates use.
- 53 FUEL PUMP MAGNETIC INDICATOR
- 54 LOW PRESSURE FUEL COCK CONTROL
Moved forward from OFF to ON.
- 55 HIGH PRESSURE FUEL COCK CONTROL
Moved forward from OFF to ON.
- 56 CABIN PRESSURE WARNING TEST
Ground use only.
- 57 TAILPLANE MOTOR CIRCUIT BREAKER

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