

# GROUP B.1

## GENERATORS AND BATTERIES (CODE GA)

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#### Equipment employed

1. The major components in the generators and batteries circuit are listed below, together with the appropriate Air Publications to which reference should be made for detailed descriptive and servicing information.

#### DESCRIPTION

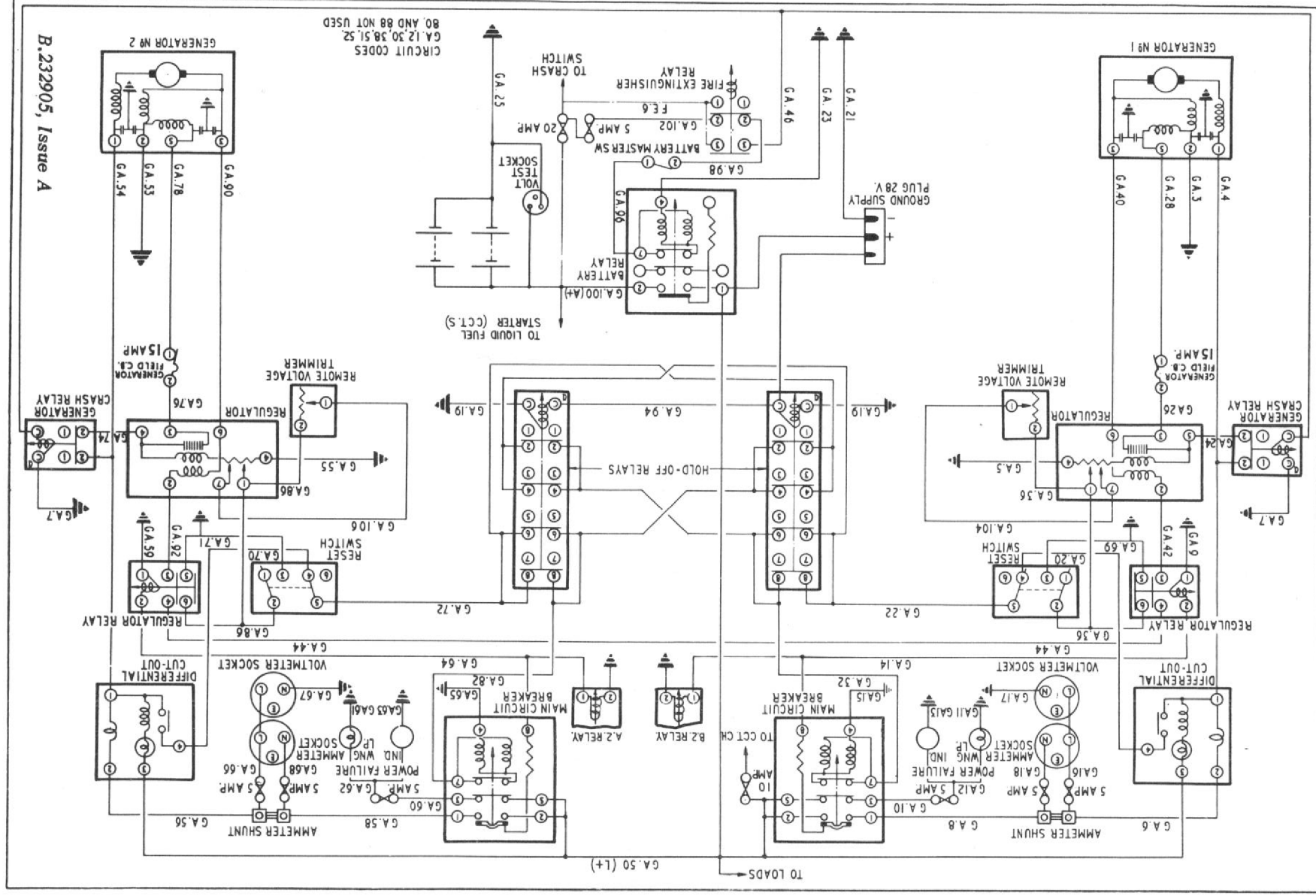
##### Generators and batteries

##### Main supply

2. Two d.c. generators, mounted on and driven by the engine accessories gearbox, supply the power for all the electrical services. The generators are connected in parallel and charge two batteries, also

Generators, Type 517, 24-volt, 6 kilowatt	...	...	...	...	A.P.4343A, Vol.1, Sect. 3
Main batteries, Type J, 24 volt, 25 amp.-hr.	}	...	...	...	A.P.4343A, Vol.1, Sect.11
Standby batteries, 12 volt, 4 amp.-hr.		...	...	...	
Voltage regulators, Type 94	...	...	...	...	A.P.4343B, Vol.1, Book 1, Sect. 1
Differential cut-outs, Type A, Mk.1	}	...	...	...	A.P.4343B, Vol.1, Book 2, Sect.10
Main circuit breakers, Type D, 200-amp.		...	...	...	
Field circuit breakers, Type A.3, 15-amp.		...	...	...	
Remote voltage trimmers, Type 3, 5-ohm	...	...	...	...	A.P.4343B, Vol.1, Book 1, Sect. 1
Battery relay, Type R	}	...	...	...	A.P.4343C, Vol.1, Book 2, Sect. 3
Hold-off and crash relays, Type S, No.3		...	...	...	
Regulator relays, Type 9B, No.1		...	...	...	
Reset switches, Rotax Type D.5506	...	...	...	...	A.P.4343C, Vol.1, Book 1, Sect. 1
Power failure warning lamps, Type B	...	...	...	...	A.P.4343E, Vol.1, Sect.18
Power failure warning indicators,					
Type C.5165Y, Mk.1 or Type A.2	...	...	...	...	A.P.4343, Vol.1, Sect.23

**Fig. 1 Generators and batteries (theoretical)**



connected in parallel, mounted on a platform in the radio bay. The generator negative supplies are earthed to the aircraft structure at earth points 10 and 16, adjacent to the generators.

#### *Generator controls*

3. The generator controls are mounted on a panel, in the radio bay, which is hinged to the top starboard longeron and anchored to the supply panel by four Dzus fasteners. Each generator has its own set of control equipment. This comprises a voltage regulator, with external trimmer, a differential cut-out, and circuit breakers for the generator main output line and the generator field. There is, in addition, a re-set switch, control relays, a power failure magnetic indicator and test equipment. Two eight-way fuse blocks, containing the control fuses, are also mounted on this control panel together with a ten-way terminal block which is used to link all the earth leads of the control equipment.

4. A battery master switch is fitted to the leg panel in the cabin. When placed in the OFF position, the switch isolates the aircraft batteries from all the electrical services, with the exception of the essential load line and fire extinguisher circuit. The generator power failure warning lamps, which light whenever a failure of the supply circuit occurs, are also on the leg panel. An external supply socket is provided on the battery support structure. It is most important that an external supply is used whenever an electrical supply is required for servicing, thus preventing discharge of the aircraft batteries.

#### *Standby supply*

5. The standby batteries are carried in a battery box mounted on the starboard side of frame 16 in the radio bay, and are a pair of 12 volt, 4 amp. hour batteries connected in series. They provide a 24 volt standby supply for certain aircraft services should the main power supply fail, namely the turn and slip indicator Chap.2, Group 3 and the cabin emergency lighting Group F.1 of this chapter.

#### *Operation*

##### *General*

6. When two generators are operating in parallel it is essential that they are regulated and aligned so that their outputs are as near equal as possible, under all operating conditions, to ensure equal load sharing and stable operation. Each generator is, therefore, provided with its own control equipment and although it operates as an independent unit, having its own characteristics when correctly regulated and aligned, it will tend to operate in conjunction with the other generator to provide a common supply.

7. To bring a generator into operation on a line to which another generator is already connected, it is necessary to momentarily boost the regulator controlled voltage of the additional generator so as to overcome the higher electro-motive force of the first generator, which tends to oppose the current flow from the additional generator. This boost is obtained by increasing

the resistance in series with the operating coil of the voltage regulator and consequently reducing the resistance in series with the generator field windings. This operation is effected by a 3-ohm trimmer resistance in the voltage regulator, which, prior to operation of the generator, is in circuit. During normal operation of the generator, contacts 5 and 6 of the regulator relay, which are closed when the relay is energized, automatically short-circuit the trimmer, while during alignment of the generators, the trimmer may be short-circuited by use of the re-set switch (*para.20*). Additionally, contacts 3 and 4 of the relay complete the load balancing line when the relay is closed (*para.8*). As each generator control circuit is duplicated, it is only necessary to follow the operation of one generator and its control equipment to fully understand the circuit.

##### *Generation*

8. When a generator commences to rotate, an increasing voltage is developed across the output terminals, due to the residual magnetism in the field. This output is fed to the differential coil of the cut-out, the operating coil of the voltage regulator, via the normally closed contacts (2 and 2a) of the generator crash relay (*para.13*) and to the generator field windings, via the carbon-pile resistance in the voltage regulator. The generator field windings receive extra energization by this current which permits the output voltage to rise rapidly and this voltage builds up in opposition to the battery voltage applied to the differential coil windings and the ballast lamp in the cut-

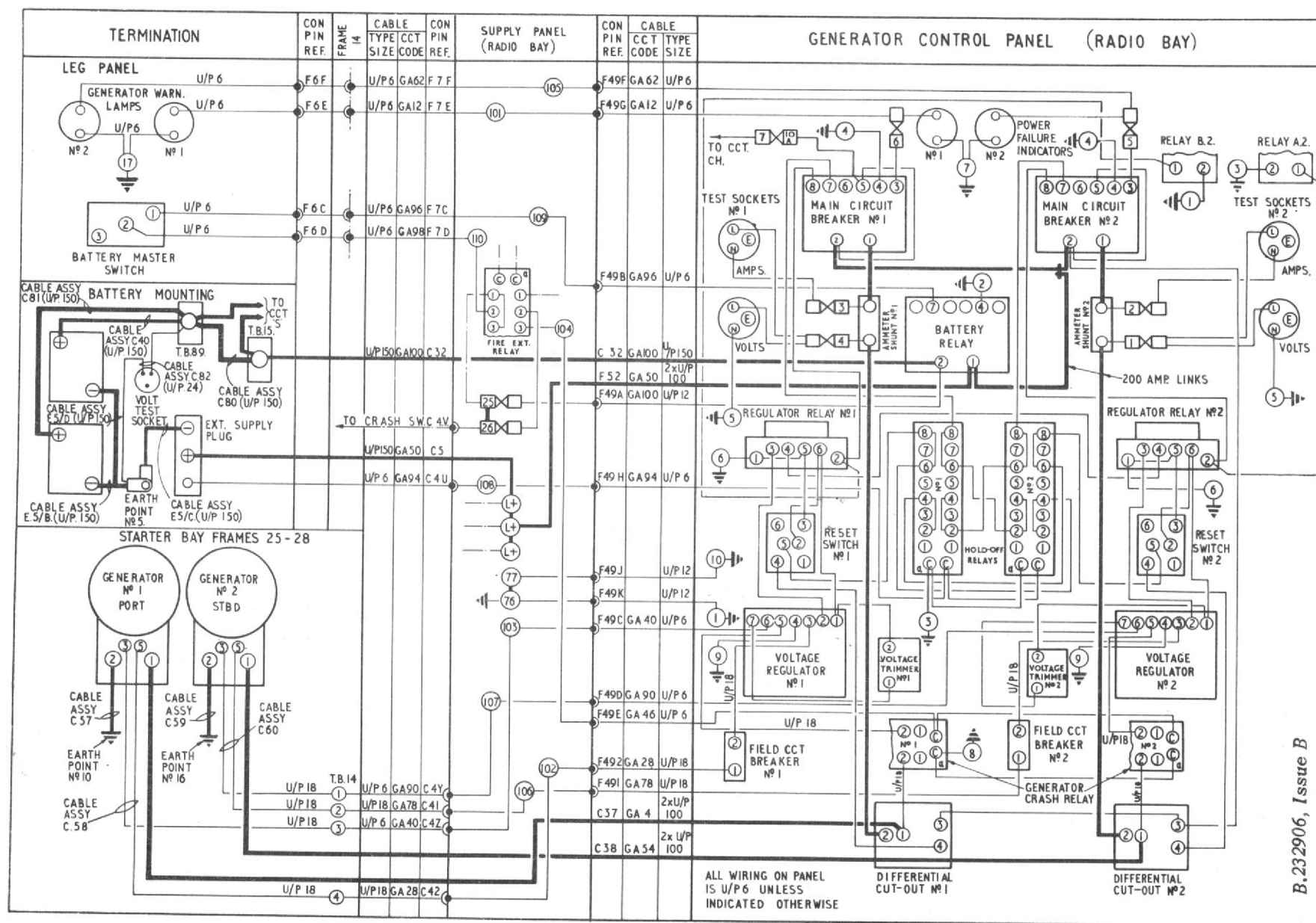


Fig.2 Generators and batteries (routing)



out. The differential coil windings are such that, when the generator output rises to a figure of 0.35 to 0.75 volts above that of the batteries the current in the coil polarizes the armature sufficiently to cause it to move over and so close the contacts. This action energizes the closing coil of the main circuit breaker, via the re-set switch and hold-off relays, thus closing the main contacts so that the differential coil and ballast lamp are shorted out. The current now flows from the generator, through the series coil which holds the armature in the contacts-closed position and in this position the polarizing magnets also bias the armature in this direction. At the same time, the auxiliary contacts of the circuit breaker are opened, thus inserting the hold-in coil into the circuit and breaking the circuit to the power failure warning lamp and magnetic indicator.

9. As the circuit breaker main contacts close, a supply is fed to energize the regulator relay, thus removing the regulator voltage boost (*para.7*) and completing the circuit of the load balancing coil in the voltage regulator, the coil being supplied from the generator interpole windings.

10. The action of the operating coil in the voltage regulator is to adjust the resistance of the carbon-pile, in series with the generator field windings, relative to the load on the generator and thus maintain the voltage constant throughout the range of generator speed and output. The function of the load balancing coil is to act on the

carbon-pile to reduce the voltage of the generator when it is overloaded in relation to the other generator and to increase the voltage when it is underloaded, thus the output is varied according to the load imposed and the load is shared more or less equally between the two generators.

#### *Power failure*

11. Power failure is indicated by one warning lamp and one magnetic indicator for each generator. When the generator voltage falls below that of the batteries, a reverse current flows in the series coil of the cut-out and this reverses the polarity of the armature and opens the contacts. This action de-energizes the holding coil of the circuit breaker and breaks the main contacts, thus disconnecting the generator from the batteries. The contacts feeding the power failure warning lamp and magnetic indicator are made when the main contacts break and the lamp and indicator operate to indicate failure. At the same time, the regulator relay is de-energized to break the load balancing line and connect into circuit the 3-ohm trimmer in the voltage regulator, thus providing voltage boost, which, if the fault was of a temporary nature, will enable the generator to re-commence operation in the normal manner. When the generator is shut down, i.e., engine stopped and batteries isolated, the armature of the cut-out is biased in the open position by the polarizing magnets.

#### *Batteries*

12. The main batteries are connected to the positive bus-bars via the battery relay which is energized from the battery essential load line and controlled by the fire extinguisher relay and battery master switch. Placing the master switch in the ON position, with the fire extinguisher relay de-energized, i.e., in its normal condition, energizes the closing coil of the battery relay and causes the main contacts to close and the auxiliary contacts to open. The batteries are thus connected to the bus-bars through the main contacts while the hold-on coil of the battery relay is energized by the breaking of the auxiliary contacts.

#### *Off-loading*

13. In the event of a crash landing, the inertia switches in the fire extinguisher circuit (Group C.2 of this chapter) will operate and energize the fire extinguisher relay which, apart from operating the fire extinguisher, will also break the supply to the battery master switch and battery relay, thus opening the main contacts of the battery relay and isolating the batteries from all but the essential load line and fire extinguisher circuit. At the same time, the fire extinguisher relay makes the supply to the generator crash relays, which become energized and break the supply to the voltage regulator operating coils, thus off-loading the generators.



RESTRICTED

#### *External supply*

14. When an external supply is connected to the external plug for testing the aircraft electrical equipment, the coils of the hold-off relays are fed from the external supply, via a "loose" positive link, before the main positive connection is made. The hold-off relays are thus energized and break the feeds to the generator circuit breakers, the main contacts of which open to isolate the generators so preventing the external supply attempting to "motor" the generators. It is also advisable to place the battery master switch to OFF when the external supply is connected, to prevent the batteries being discharged should the external supply voltage fall below that of the batteries.

#### *Test sockets*

15. Voltmeter and ammeter test-sockets, together with an ammeter shunt, are provided for each generator for use when adjusting the circuit as described in para.19.

#### *Standby supply*

16. The operation of the standby supply circuit (B+) will be obvious when reference is made to fig.3 of this group and to the routing and theoretical diagrams contained in Group F.1. For information regarding the turn and slip indicator, reference should be made to Chapter 2, Group 3 of this section.

### **SERVICING**

#### **General**

17. For general servicing of the circuit as a whole, reference should be made to Group

A.1 of this chapter. Absolute cleanliness of all parts, particularly the generator brush gear and commutator, together with the immediate remedy of any defects, however small, is essential for the reliable operation of the circuit. The contacts of the cut-outs, circuit breakers and relays must be kept clean and the terminals of all components must be kept tight and free from corrosion. For functional tests and detailed servicing of the standard components used, reference should be made to the appropriate Air Publications quoted in para.1.

#### **Servicing main batteries**

18. The state of charge of the main batteries installed in this aircraft is determined by voltage, as no check of specific gravity is possible. A small three-pole test socket is located adjacent to the batteries for this purpose. When using this socket, it is essential to check one battery at a time and this may be accomplished by removing the Cannon plug from one of the batteries while checking the other and vice versa. After test, ensure that both Cannon plugs are reconnected to the batteries.

19. The open circuit voltage of a fully-charged battery is 25.8 to 25.1 volts, a quarter to half discharged battery is 25.1 to 24.5 volts and a half discharged battery is 24.5 to 24.2 volts. To obtain more definite information on the state of charge, an off-load reading should be taken immediately after taking a reading with the battery connected to a load of approximately 20

amp. for 15 seconds, i.e., radio, navigation lamps and cabin lamps switched on. The increase in reading from on load to off load should be approximately 1 volt. If the aircraft is not flown or the engine is not run-up for more than two weeks, the batteries must be removed and given a freshening charge. Under no circumstances should the batteries be left without a freshening charge for more than four weeks or serious deterioration will result. For detailed servicing of the batteries, reference should be made to the Air Publication quoted in para.1 of this group.

#### **Paralleling of generators**

20. The output of the generators should always be equalized after the fitment of a new generator, a new voltage regulator, a new cut-out or after any servicing which may have disturbed the alignment of the circuit and also when a check indicates that adjustment is necessary. The procedure, after the units have been adjusted on the bench as described in the relevant Air Publications, is as follows:-

#### **Note . . .**

*Before commencing operations, ensure that the aircraft batteries are at least 80 per cent charged.*

- (1) Start the engine in accordance with the instructions contained in A.P.4347J—G.H.N. (Ground Handling Notes) and in A.P.4282, Vol.1.
- (2) Connect the test leads of a 0–30 volt voltmeter to No.1 generator voltmeter

test socket.

- (3) With the engine running at the normal cruising speed (7,800 r.p.m.) and No.1 generator re-set switch set to RE-SET (generator off-load, power failure magnetic indicators indicating white and warning lamps illuminated) adjust the external voltage regulator trimmer resistance until  $28 \pm 0.25$  volts is indicated on the voltmeter.
  - (4) Increase and decrease the engine speed from ground idling (2,500 r.p.m.) to the full engine speed (7,900 r.p.m.) at least three times and then check the voltage again at the original speed to ensure that it is still at  $28 \pm 0.25$  volts.
  - (5) With the engine running at approximately 6,000 r.p.m., check the load balancing coil by connecting a voltage of 0.25 volts across terminals 2 and 6 of No.1 generator voltage regulator, terminal 2 being connected to the positive side of the supply. The regulated voltage should be reduced by 2 to 2.5 volts.
  - (6) Ensure that both re-set switches are set to RE-SET and connect a ground supply socket to the external supply plug in order to energize the hold-off relays.
  - (7) With the engine still running at 6,000 r.p.m., check that the controlled voltage is increased by  $1 \pm 0.25$  volts.
- whenever the re-set switch is set to NORMAL.
- (8) Disconnect the ground supply socket from the external supply plug, remove the voltmeter from No.1 generator voltmeter test socket and connect it to No.2 generator voltmeter test socket.
  - (9) Adjust the voltage regulator for No.2 generator by repeating operations (3), (4), (5), (6) and (7) with No.2 generator re-set switch set to RE-SET.
  - (10) Ensure that both re-set switches are set to RE-SET and disconnect the ground supply socket from the external supply plug.
  - (11) Decrease the engine speed to ground idling (2,500 r.p.m.) switch both re-set switches to NORMAL and check:-
    - (a) That both the power failure magnetic indicators indicate black.
    - (b) That both warning lamps are extinguished without flickering.
    - (c) That there is no tendency for the cut-outs and main circuit breakers to chatter.
  - (12) Should the warning lamps flicker or the cut-outs and main circuit breakers chatter, it is essential that the cut-outs and/or circuit breakers are disconnected from the equipment on the generator control panel and re-adjusted as described in the relevant Air Publications. After adjustment and recon-

nection, the operations in sub-para. (11) must be repeated to ensure that the equipment is now operating correctly.

- (13) With voltmeters in both voltmeter test sockets, increase engine speed gradually to the full engine speed (7,900 r.p.m.) and check that the voltage indicated on both instruments is  $28 \pm 0.25$  volts at all speeds.
- (14) Remove the voltmeters from the test sockets and connect ammeters to both ammeter test sockets.

#### WARNING

*Before inserting the ammeters, ensure that the voltmeter is not in circuit, as damage may be caused to an ammeter should its shunt be isolated due to the absence or failure of its fuse while both instruments are in circuit.*

- (15) Switch off the engine and during the run-down time, check by reference to the power failure warning lamps and magnetic indicators, that the cut-out contacts open at a speed of approximately 1,800 r.p.m. Also during this time check that the reverse current recorded on the ammeters is not below 15 amp. minimum or above 25 amp. maximum.
- (16) If the cut-out contacts do not open at a speed of approximately 1,800 r.p.m. or the reverse current is above the top

limit of 25 amp., the cut-outs must be disconnected from the equipment of the panel and the reverse current setting re-adjusted as described in the relevant Air Publication. After adjustment and reconnection, operation (15) must be repeated to ensure that the cut-outs are now operating correctly.

- (17) If adjustments are made, it is important that the adjusting screws are sealed with shellac varnish and the top covers re-fitted.

**Note . . .**

*The covers should be refitted as soon as possible.*

- (18) Both generators are now correctly regulated and are ready for parallelling.
- (19) Place both re-set switches to the NORMAL position and re-start the engine as in operation (1).
- (20) During the starting period, check that at approximately 2,000 r.p.m. the power failure magnetic indicators on the generator control panel change from white to black and that the warning lamps in the cabin are extinguished.
- (21) With the engine running at ground idling speed, insert a voltmeter into one of the voltmeter test sockets. Increase engine speed gradually to the full engine speed (7,900 r.p.m.) and check that the voltage indicated is  $28 \pm 0.25$  volts at all speeds.

- (22) Slow down the engine to idling speed and remove the voltmeter. Insert ammeters into both ammeter test sockets and increase engine speed gradually to max. r.p.m. Check that the current indicated on both instruments does not differ by more than 10 per cent at all speeds (A.P.4343, Vol.1, Sect.2, Chap.5, Para.56 refers).
- (23) The above conditions should be obtained if the foregoing instructions have been carried out correctly.
- (24) Check the voltage regulators for stability by switching on a typical flight load, i.e. flight instruments, radio and radar ranging. At an engine speed of 6,000 r.p.m. switch one of these loads, e.g., flight instruments, on and off at least three times. Under these conditions the regulators should respond without any tendency to hunt.
- (25) The generating circuit is now aligned and the engine may be switched off and the voltmeters and ammeters removed.

## REMOVAL AND ASSEMBLY

### General

21. The removal of the generator control panel, which carries the majority of the equipment forming the generating circuit, is described in Group A.2 of this chapter. The removal of the main and standby batteries is described in the following paragraphs. Once clear access has been obtained, the removal of the generators and other com-

ponents of the circuit, should present no special difficulties.

### Changing the main batteries

22. The two main batteries are carried on platforms located on frame 19 in the radio bay and access may be gained via the radio bay access doors. The procedure for changing the batteries is as follows:-

- (1) Ensure that the battery master switch is in the OFF position.
- (2) Disconnect the vent pipes from each battery.
- (3) Remove the Cannon plugs from each battery and place in such a position so that they will not short on the aircraft structure.
- (4) Disengage the special quick release bolts securing each battery and carefully remove the batteries from their mounting platforms.
- (5) Examine the replacement batteries to ensure that their voltage is within  $\pm 0.5$  volts of each other. Place the replacement batteries on the platforms and reverse the procedure in sub-para. (2) to (4).

**Note . . .**

*Before attempting to connect the batteries, ensure that the Cannon plugs are not shorting against the aircraft structure.*

#### **Removal of standby batteries**

**23.** These batteries are carried in a box located on the starboard side of frame 16 in the radio bay, access being gained by opening the radio bay access doors. To remove the batteries, proceed as follows:-

- (1) Render the aircraft electrically safe, as described in Group A.1 of this chapter.
- (2) Disconnect cable assembly P.C.13 from cable assembly P.C.17 at the plug and socket connection on the side of the battery box.
- (3) Withdraw the two draw bolts at the top of the box from the mounting structure and lift the box upwards to disengage its lower mountings.
- (4) Remove the box and batteries from the aircraft.
- (5) The batteries may be removed from the box by opening the lid and disconnecting the leads of the cable assembly P.C.13 from the battery terminals.





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