

Chapter Twenty-two

GENERAL INFORMATION

(relating to overhaul, reconditioning, repair, and salvage)

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THIS CHAPTER which is applicable to both the Ghost 48 Mk. 1 and the Ghost 48 Mk. 2, contains general information relating to overhaul, reconditioning, repair, and salvage, and should be read in conjunction with chapters 23 to 52.

DISMANTLING THE ENGINE

The following information should be read in conjunction with chapters 23 and 24 which respectively contain instructions for dismantling the engine and dismantling the sub-assemblies.

It is assumed that the engine will be received without the air compressor, generator, tachometer generator, hydraulic pump, and turbo starter, or that they are removed from the engine before dismantling commences. The exhaust cone and fire-guard will be received separately.

As the engine is dismantled, each sub-assembly should be treated as a complete unit and care must be taken that small components do not become mixed or placed with the wrong assembly.

Locking devices

Where the removal of split pins and tab washers is not specified, it is assumed that these and any other locking devices will be released by the operator before proceeding with dismantling. To avoid damage to the hole in union nuts and locking tabs, etc., it is important that locking wire is cut and then removed from the locking locations. The wire must not be broken by twisting or stretching

either with pliers or by unscrewing the nut or part concerned. All tab washers, split pins, locking washers and joint washers should be discarded immediately after removal.

Shims

Distance washers and shims should be tied to their associated parts and subjected to inspection. Retaining shims with their associated parts will save time during subsequent reassembly.

Component marking

Before a component is removed, ensure that it is clearly marked to indicate its position on the engine, unless the design is such as to ensure refitment in the correct position. This is of particular importance with components which can be turned through 90 or 180 deg. Unless the operator is familiar with the particular engine he is concerned with, it is sound practice to make written notes of the location of, or to label, all small parts so that each part can be refitted with absolute certainty in its original position during reassembly.

Where temporary marking is necessary, as, for instance, when it is required to draw the inspector's attention to a crack, or other defect, either with a view to determining its rate of increase or to indicate the area where a more thorough examination is considered necessary, chalk, soapstone, or some other similar substance which will not leave any carbon deposit should be used. Marking media such as lead pencil, wax crayons and grease pencils

should not be used to mark the position of items, or defects, as instances of failure have occurred on items subject to high temperature, due to carburisation and subsequent hardening or embrittlement of the immediate area surrounding the carbon deposit when subjected to intense heat. Chalk and soapstone will not leave any carbon deposit.

Evidence

Filters, and the oil which drains from the engine should be examined for evidence of scoring or wear in the form of metallic particles; similarly, an indication of the condition of components can be obtained if they are examined whilst burnt oil and signs of running are still visible.

Use of tools

When components are difficult to remove or to separate, due to the adhesion of jointing compound between the mating faces, or the close fitting of dowels or studs, light tapping with a rubber or raw-hide mallet around the periphery of the component should be sufficient to break the joint. In no circumstances should any form of wedge be inserted between the mating faces in an endeavour to part them.

The method of dismantling described, assumes that the extractors provided in the tool kit will be used. In some instances, however, the fit of mating parts may render a tool unnecessary as the operation can be effected by hand.

Where, in the following chapters, reference is made to "pressing off" or "pressing out" a component, i.e., ball bearing, bearing housing, shaft, etc., it is assumed that a suitable press will be utilised when necessary, although certain pressing operations may possibly be effected by hand without any mechanical assistance.

Blanking

All apertures, pipes, etc., should be blanked off on both the engine casing and accessories directly they are exposed, unless the component in question is being completely dismantled immediately. In the absence of proper blanking plates, caps or plugs, cavities should be blanked off with improvised plates of metal or plywood; pipes and similar openings may be blanked off with rubber caps. Cotton waste, rags, etc., must never be used for these purposes, as small particles may enter the oilways or the fuel system with disastrous results.

Certain obvious operations, such as the removal of blanking plugs, pipe union bodies, etc., have been omitted. Such items must be removed for cleaning, and unless definite instructions to the contrary are given, the necessity for their removal must be assumed.

Miscellaneous

As each sub-assembly is dismantled the respective components should be placed in a box allocated to the particular sub-assembly, and the box

marked with the engine number. Ball and roller bearings should be wrapped in greaseproof paper for protection, and when the parts of a bearing are separated the outer and inner races should be tied to the centre cage before wrapping. Re-usable washers, distance pieces, spacers, etc., should be tied to their respective main component, so that upon reassembly time will not be lost in checking components to obtain the correct washer thickness.

Where reference is made to turning or bending down the locking tabs of tab washers, a small chisel-pointed punch in conjunction with a light hammer should be used. Similarly, a cup locking washer is best opened with a small flat-nosed punch and a light hammer.

Seeger circlips should be removed with a pair of Seeger circlip pliers. The use of a screwdriver or pointed tool for this purpose is liable to damage the circlip groove as well as the circlip.

Where jointing compound has been employed, all traces of the previous coating must be removed and the mating surfaces thoroughly cleaned before reassembly. When the joint faces are perfectly clean, ensure that there are no nicks, burrs, scratches or other damage which might prevent the components seating correctly.

REASSEMBLING THE ENGINE

The following information should be read in conjunction with chapters 33 and 34 which respectively contain instructions for reassembling the sub-assemblies and reassembling the complete engine.

Scrupulous cleanliness, the proper use of the correct tools, and attention to detail are vital at every stage of the rebuilding operations. Ball and roller bearings must be treated with particular care.

Application of blanks

Any loose parts, small tools, or foreign bodies left within, or entering the engine after assembly might result in extensive damage when the engine is started and run. Even a microscopic particle can cause failure of a fuel system component. Therefore, blanks and covers should not be removed until the aperture they protect can be closed by the assembly of the related parts, and as assembly proceeds all unconnected unions and apertures should be blanked off.

Splined and serrated drives

Splined and serrated drives should be lubricated with Shell grease Rhodina No. 2 or a grease such as British Specification D.T.D. 825. If, however, these are not available some other lubricant may be used—e.g. engine oil. On no account should these drives be assembled dry.

Lubrication of moving parts

All moving parts should be oiled with clean engine oil during assembly to ensure adequate lubrication during the few seconds immediately after the first start when the normal pressure lubrication may not have reached the bearing surfaces. Plain

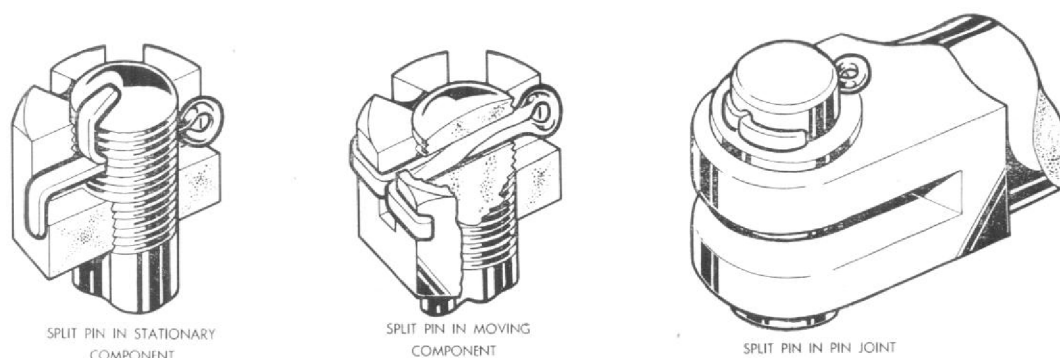


Fig. 1. Correct method of fitting split pins.

bearings, such as the bushes in the bottom wheel-case of the 48 Mk. 1, should be lubricated with anti-scuffing paste (British specification D.T.D. 900/4284) or Ragosine L.M. grease (D.T.D. 900/4424) immediately prior to assembly.

Joint faces and application of compound

Ensure that all joint faces are clean and that there are no burrs, nicks, or other damage which might prevent the components seating properly or an oil-tight joint being obtained. Where jointing compound has been employed, all traces of the previous coating must be removed. Where the use of jointing compound is specified in this section, apply a thin even film of Wellseal jointing compound to each joint face.

Anti-seize compound on screw threads

To reduce the risk of seizure, all screw threads which are subjected to high temperature should be coated with an anti-seize compound. Because of the critical torque loading which must be applied when tightening the nuts and bolts which secure the turbine disc to the hub shaft, the threads of these nuts and bolts must be lubricated with either Ragosine L.M. grease (British specification D.T.D. 900/4424) or anti-scuffing paste to British specification D.T.D.900/4284. For all other screw threads which are subjected to heat, anti-seize grease to British specification D.T.D.392 is recommended, but any other reputable graphite anti-seize grease may be used as an alternative.

Screw threads which are not subjected to heat should be coated with a mixture of engine oil and a reputable graphite anti-seize grease.

Locking devices

New locking devices, tab washers, split pins, etc., and new circlips, joint washers, oil seals, and similar parts must be used; each chapter includes a list of consumable parts essential to the rebuild described in that chapter.

Split pins

The normal recommended practice for stationary components is for the head of the split pin to be fitted into the slot in the nut and for one leg of the split pin to be bent up over the end of the bolt and the other down over the nut as illustrated in Fig. 1.

For moving components the split pin must be carefully twisted through 90 degrees before insertion, so that with the head fitting correctly in the slot

in the nut, the legs of the split pin can be wrapped round the hexagon of the nut as illustrated in Fig. 1. To reduce the possibility of fretting due to split pin slackness, the split pins for all moving components are a very close fit in their holes and must be of the correct length. Split pins fitted in pin joints are to be fitted as indicated in Fig. 1. Except in the case of pin joints, the head of the split pin must not protrude beyond the slot of the nut. Split pins should be inserted and the legs bent over with pliers and not by hitting with a hammer, as use of the latter tool may cause fracture of the legs at the point of bending or at the base of the head on the sharp edges of the nut or of the hole. After locking, the bends of all split pins should be examined for cracks, especially in highly stressed components.

Tab washers

After tightening a bolt or nut which is locked by a tab washer, examine the washer to ensure that the initially fixed limb has not sheared whilst tightening the bolt or nut. The locking limb should be turned up to a flat on the hexagon and not on a corner. Finally, examine the bend for cracking.

Angular contact thrust bearings

If an angular contact thrust bearing is fitted the wrong way round, so that the thrust is taken in the wrong direction, premature failure of the bearing and the gears which it locates can be anticipated. Before fitting an angular contact thrust bearing therefore, it is essential to determine the direction of thrust of the individual shaft and to fit the bearing with the "closed end" to the front in relation to the direction of thrust to the shaft as shown in Fig. 2.

Fig. 2 shows that the inner diameter of the outer race tapers from the balls towards one end to form a narrow end face (A in Fig. 2), and that the balls between the cage and the narrow face of the outer race can be readily seen. This is referred to as the "open end". At the "closed end" (B in Fig. 2) the balls cannot be seen as the end face of the outer race is approximately the same width as the space between the cage and the outside diameter of the complete bearing. These bearings are marked "THRUST" on the face of the outer race at the "closed end".

Rigid pipes

Sharp edges at the end of the conical seat of unions can cause deep grooves in the belled end of

The angular contact thrust bearings fitted to these engines are tabulated below.

Sub-assembly	Engine	Location	Fitment	Chapter Ref.
Centre housing	48 Mk. 1	Upper, for upper vertical drive	"Closed end" nearest the ring-nut on the upper vertical drive	33C
	48 Mk. 2			
	48 Mk. 1	Front ball bearing, for horizontal drive shaft (Pre-mod. 1148)	"Open end" nearest the ring-nut at the front end of the horizontal drive shaft	33C
	48 Mk. 2			
Oil sump	48 Mk. 2	Bearing, for the pump drive assembly	"Closed end" nearest the ring-nut on the fuel pump driving gear	33F
	48 Mk. 2	Bearing, for accessory drive	"Closed end" nearest the ring-nut on the accessory drive bevel gear spindle	33F

rigid pipes and this may result in complete fracture of the pipe at that point. Before fitting any rigid pipe therefore, each union should be examined to ensure that there is neither a sharp edge nor a burr which might damage the pipe when the union nut

pipe has a flanged end fitting secured by studs and nuts, ensure that malalignment is not caused by the holes in the flange binding on the studs.

Bundy pipes

When tightening Bundy type union nuts, the nut must be tightened sufficiently to deform the olive so that slight but positive necking of the pipe is produced; Bundy pipes may be recognised by their characteristic bright silvery (tinned) finish. Excessive tightening must, however, be avoided as each time a particular pipe is fitted the nut will require just a little extra movement to ensure a leak-free connection, and it is the sum of these extra movements, plus the initial distortion, which dictates the life of the pipe assembly. The slight necking of the pipe is necessary to ensure a mechanical lock; friction pressure of the olive on the pipe, although probably leak-free, is not sufficient. For the foregoing reasons, whenever a pipe which is to be refitted is disturbed, it is most important that the olive remains in the same position on the pipe; any lateral movement of the olive on the pipe will destroy the initial seal.

When fitting a new assembly, the union nuts should be tightened as shown in the following table.

Bundy pipe	Thread	Tighten
$\frac{5}{16}$ in.	$\frac{1}{4}$ in. B.S.P.	Hand-tight plus one complete turn
$\frac{1}{2}$ in.	$\frac{1}{2}$ in. B.S.P. or $\frac{7}{16}$ in. by 26 T.P.I.	Hand-tight plus one complete turn plus one-sixth (60 deg.) of a turn

When fitting a used assembly, the union nuts should be tightened hand-tight plus one-sixth (60 deg.) of a turn regardless of the pipe size. A radial pencil line should be marked on the nut end to provide a datum of the nut movement.

It must be fully understood that no attempt must be made to continue tightening these nuts until appreciable resistance is felt. The check for sufficient tightening, having complied with the foregoing instructions, is freedom from leaks during engine test. If it is necessary to tighten a union nut

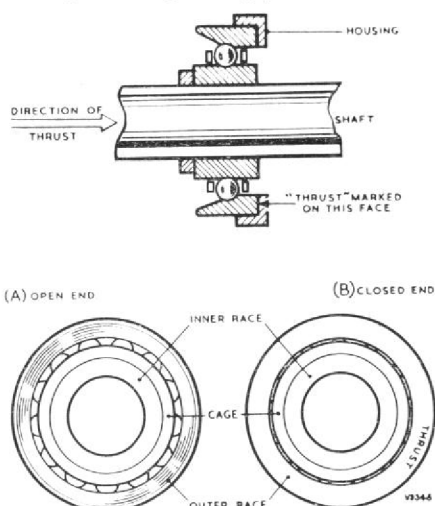


Fig. 2. Correct method of fitting angular contact thrust bearing.

is tightened. Burrs found should be carefully removed and any sharp edge radiused very slightly.

Similarly, examine the belled end of each pipe for evidence of circumferential ridging in the belled mouth. If a ridge has been formed it should be removed with a small hand scraper and the belling lightly polished with metal polish; finally, wash away all traces of abrasive. After polishing, visually examine the pipe for cracks. Cracks in any form are not acceptable, and defective pipes must be replaced by new pipes.

Rigid pipes must never be forced into position against their natural spring. If this appears to be necessary, verify that the pipe is the correct one for the position, and if this is confirmed, check that the run is correct. If the foregoing points are assured, carefully set the pipe to a correct fit. When the

further to correct slight leakage, this must be done to the least possible extent that will cure the leak, tighten the nut a few degrees at a time only. If excessive necking of the pipe, or excessive distortion of the olive has occurred, a new pipe assembly should be fitted.

Flexible pipes

Reference should be made to chapter 51 for the special requirements applicable in the treatment and handling of flexible pipes.

Accessories

New or reconditioned fuel system components and engine-driven accessories will have received protective treatment, and before fitment to the

engine the necessary blanks must be removed and all inhibiting fluid drained off. If engine-driven accessories are not to be connected to the test bench immediately, the connections must be blanked with air-tight caps immediately after draining.

Torque loading

The bolts and nuts listed in the following table must be tightened to the torque loading specified.

BALL AND ROLLER BEARINGS

It should be noted that manufacturers of ball and roller bearings build their bearings to two standards "Commercial" and "Aircraft". The Aircraft standard bearings are manufactured from

<i>Sub-assembly</i>	<i>Engine</i>	<i>Torque load lb. in.</i>	<i>Chapter Ref.</i>
Front bearing nut on impeller front pivot	48 Mk. 1 48 Mk. 2	1920-2450	33A
Nuts, for front pivot studs	48 Mk. 1 48 Mk. 2	500-600	33A
Nuts, for impeller to centre shaft studs	48 Mk. 1 48 Mk. 2	500-600	33A
Nuts, for centre shaft to extension shaft bolts	48 Mk. 1 48 Mk. 2	190-230	33A and 34
Nuts, for turbine disc to hub shaft bolts	48 Mk. 1 48 Mk. 2	850-950	33A and 34
Bolts, for turbine disc to hub shaft	48 Mk. 1 48 Mk. 2	850-950	33A and 34
Nut, for hub shaft	48 Mk. 1 48 Mk. 2	8000-8500	33A
Nut, for starter dog	48 Mk. 1 48 Mk. 2	2040-2400	33D
Ring nut on accessory drive gear (vertical)	48 Mk. 2	450-500	33F
Ring nut on front end of centre housing horizontal shaft	48 Mk. 1 48 Mk. 2	500-550	33C
Ring nut on lower drive gear	48 Mk. 1	500	33F
Diffuser bolts	48 Mk. 1 48 Mk. 2	800-850	33G and 34
Lucas burner	48 Mk. 1	700-750	33J
Igniter plug screen tube (pre-mod. 966) or igniter plug (mod. 966) to adapter in com- bustion chamber	48 Mk. 1	240-300	34
Screen tube, or igniter plug, to torch igniter body	48 Mk. 2	Pre-mod. 1199 240-300 Mod. 1199 336-396	—
Fuel jet to torch igniter body	48 Mk. 2	336-396	—
Banjo pillar nut to torch igniter fuel jet	48 Mk. 2	216-240	—
Ignition cable union nuts to igniter plug and to high energy ignition unit	48 Mk. 1 48 Mk. 2	150-160	34
Welded nuts (sheet metal)	48 Mk. 1 48 Mk. 2	360	33K and 34

specially selected materials and have more rigid inspection standards than the Commercial standard bearings and, although the bearings may have the same type number and may look absolutely identical, the aircraft standard bearings are obviously of a much higher quality than the commercial standard.

In view of the serious consequences which may result from fitting commercial standard bearings in place of those to aircraft standard, the practice of obtaining these bearings from the local agent of the manufacturers is not advised, as no responsibility will be accepted for damage to engines which result from the failure of bearings obtained other than through The de Havilland Engine Company Ltd.

FUEL SYSTEM COMPONENTS

The following information should be read in conjunction with these chapters dealing with fuel system components.

The importance attached to cleanliness cannot be over-emphasized and extreme precautions must be taken in every instance to ensure that all details are thoroughly clean.

Box or ring spanners must be used for nuts and bolts whenever possible to ensure that damage is not caused to nuts, bolts, etc., by rounding off the flats and corners. Slave nuts, washers, etc., should be used in those instances where components are assembled temporarily, to enable sound parts to be fitted on final assembly.

Dismantling

Before attempting to dismantle any unit, study any complaints reported and examine the exterior for any visible signs of damage or interference (e.g. broken locking wire), and check that all exposed connections have been securely covered to prevent the entry of any foreign matter into the interior of the unit while off the engine. Examine all exposed threads to ensure that no damage has been caused in transit.

The special tools required for dismantling and assembling each unit are listed at the beginning of each "Dismantling" section of each chapter and illustrated in the appropriate portions of the text. It is assumed that hide hammers and a wide range of punches and drifts are available.

It is advisable to have close at hand small receptacles or boxes in which such small items as nuts, washers, bolts, springs, screws, etc., may be stored for safety until required for inspection and subsequently for rebuild.

It is recommended that only one set of components at a time be in use on the bench, in order to prevent the accidental inclusion in the assembly of small items, such as washers, etc., from other sets. If one set only is in use at a time, the possibility of shortages or interchanging of parts will be considerably reduced.

Cleaning

It is important that the bench upon which

work is to be effected be scrupulously cleaned and kept free from all dirt, swarf, etc. It is recommended that the bench be covered with zinc sheeting or other material to facilitate cleaning, and that several clean containers be available in which to keep the various parts removed.

Prior to complete washing and cleaning, remove all gaskets except where stated to the contrary. All parts must be thoroughly washed and cleaned before reassembling, for which purpose a paraffin bath and spray gun with adequate filtering facilities are recommended. Hard brushes or abrasives of any kind should on no account be employed. Use of rags is not recommended since this may result in clogging.

Inspection

The inspection and viewing checks are all contained in the "Inspection" portions of each chapter, but they do not contain dimensional and clearance checks. The latter can be carried out by reference to the Schedules of Fits, Clearances and Repair Tolerances at the end of each chapter.

Each detail should receive close scrutiny for any signs of damage or wear, and, if necessary, discarded and a replacement fitted.

The detail components are grouped as nearly as possible into sub-assemblies. Separate details which do not form part of a sub-assembly are dealt with nearest to their associated group, and the viewing and inspection checks are listed to correspond with the normal dismantling sequence of operations.

All paper joints or gaskets, banjo washers, etc., must be replaced by new ones whenever the seal is disturbed, and new split pins and locking wire must always be used. Details of those parts which are not interchangeable are given under "Replacements" in each instance.

It is usual during inspection to reject all parts rendered redundant by the embodiment of essential modifications and the inspector must thus have complete information concerning modifications so that redundant parts can be rejected. Tab washers, split pins, gaskets and other consumable parts will be rejected during dismantling.

Scores and surface damage. The standard of acceptance when dealing with scores and other forms of surface damage may vary according to the number of hours the unit is to run, the purpose and region of operation, accessibility of spares and the issue of up-to-date instructions.

In general, scores on moving parts or joint faces will impair the operational efficiency of the unit, and may cause leakage, hence the parts concerned must be replaced.

Where the region of operation is an outlying one and spares are not readily accessible, very slight surface scoring insufficient to seriously affect the efficiency and safety of the unit, may be permissible at the discretion of the appropriate inspecting authority.

ALPHABETICAL INDEX OF COMPONENTS SHOWING APPLICABLE REPAIR SCHEMES

For the titles of the repair schemes which are listed beside each component, reference should be made to the relevant T.R. No. in the Numerical Index which follows this index. For further details reference should be made to the relevant chapter.

Component	T.R. No.	Chapter
Air-intake	123, 124, 213, 260, 284, 320	28D
Air compressor housing	See wheelcases	28D
Bottom wheelcase	See wheelcases	28D
Centre casing	312, 389	28F
Rear bearing housing	327	28B
Centre housing	123, 270, 399	28D
Centre shaft	See main shaft	28B
Combustion chambers		30
Expansion chamber	272, 273, 287, 289, 300, 305, 314, 339, 373, 418	30
Flame tube	344, 345, 346, 347, 352, 353, 354, 388, 397, 412	30
Diffuser casing and rear cover	227, 234, 250, 268, 275, 277, 279, 281, 308, 313, 315, 338, 366, 384, 387, 401, 402, 408, 415	28E
Deflector assembly	322, 351, 371, 410	28E
Sealing plate	299, 374, 380, 406	28E
Discharge nozzles	See nozzle ring	28F
Exhaust system		31
Fairings	241, 311, 381	31
Fireguard	375, 379, 409	31
Inner cone	226, 242, 297, 382	31
Outer cone	333, 378, 385, 405, 413, 414	31
Propelling nozzle	304	31
Support tubes	217	31
Expansion chamber	See combustion chambers	30
Extension shaft	See main shaft	28B
Fairings	See exhaust system	31
Fireguard	See exhaust system	31
Flame tube	See combustion chambers	30
Generator drive	See wheelcases	28D
Hub shaft	See main shaft	28B
Impeller	See main shaft	28A
Inner exhaust cone	See exhaust system	31
J		
K		
L		
Main shaft		28A, B, C
Centre shaft	111, 204, 221, 403	28B
Extension shaft	221, 403	28B
Hub shaft	215, 225, 266	28B
Impeller	111, 133, 134, 201, 222, 253, 254, 357, 396	28A
Rear bearing housing	See centre casing	28B
Turbine blades	151	28C
Turbine disc	215, 225, 252, 266, 340	28C

<i>Component</i>	<i>T.R. No.</i>	<i>Chapter</i>
Nozzle ring assembly		28F
Discharge nozzles	247, 341	28F
Inner nozzle ring	343	28F
Nozzle blades	152	28F
Nozzle shroud	416	28F
Turbine shroud	376, 417	28F
Oil pump	342, 367, 421	29
Oil sump	307, 334, 394, 411	29
Outer exhaust cone	See exhaust system	31
Planet gear carrier cover	See wheelcases	28D
Propelling nozzle	See exhaust system	31
Q		
Rear bearing housing	See centre casing	28B
Sealing plate	See diffuser casing	28E
Starter gear casing	See wheelcases	28D
Support tubes	See exhaust system	31
Top wheelcase	See wheelcases	28D
Turbine blades	See main shaft	28C
Turbine disc	See main shaft	28C
Turbine entry duct	See nozzle ring assembly	28F
Turbine shroud	See nozzle ring assembly	28F
U		
V		
Wheelcases		28D
Air compressor housing	364	28D
Bottom wheelcase	393	28D
Generator drive	350, 383	28D
Planet gear carrier cover	390	28D
Starter gear casing	398	28D
Top wheelcase	124, 404	28D

NUMERICAL INDEX OF TURBINE REPAIR SCHEMES

This index contains a complete sequence of T.R. numbers but titles are given only where the repair is applicable to the engines dealt with in this handbook. The chapter and page listed against a repair indicates where working instructions, or other information, will be found. Where modifications are referred to further information will be found in the Modification News Sheets issued by The Service Department of The de Havilland Engine Company. "Not issued by D.O. (Drawing Office)" is self-explanatory, "Not issued" in the "Refer to" column indicates that drawings have been issued by the D.O. but instructions have not yet been compiled for publication in this handbook.

T.R. No.	Title	Refer to Chapter	Page	T.R. No.	Title	Refer to Chapter	Page
1 to 110	_____	Goblin		205	Cancelled by T.R.254	—	
111	Impeller, oversize centre shaft dowels	28A	18	206	Cancelled by T.R.253	—	
112 to 118	_____	Goblin		207	Cancelled by T.R.277	—	
119 & 120	Not issued by D.O.	—		208	Cancelled by T.R.278	—	
121	_____	Goblin		209	Cancelled by T.R.279	—	
122	Cancelled by T.R.274	—		210	Cancelled by T.R.280	—	
123	Air-intake, centre-housing dowel	28D	5	211	Cancelled by T.R.281	—	
124	Air-intake, top wheel-case dowel	28D	8	212	Cancelled by T.R.282	—	
125	Not issued by D.O.	—		213	Air-intake, bearing housing bore	28D	12
126 to 132	_____	Goblin		214	_____	Ghost Fifty	
133	Impeller, oversize centre-shaft studs	28A	21	215	Turbine disc, hub shaft spigot	28C	30
134	Impeller, oversize pivot studs	28A	21	216	_____	Ghost Fifty	
135 to 141	_____	Goblin		217	Support tubes, wear	31	2
142	_____	Ghost Fifty		218 & 219	Cancelled by mod. 1018†	—	
143	Cancelled by T.R.275	—		220	Cancelled by T.R.221	—	
144	_____	Ghost Fifty		221	Centre and extension shafts, oversize bolts	28B	10
145 to 150	_____	Goblin		222	Impeller bore, wear or damage	28A	20
151	Turbine blades, repair	28C	21	223	Cancelled by T.R.270	—	
152	Nozzle blades, repair	28F	11	224	_____	Ghost Fifty	
153 to 180	_____	Goblin		225	Turbine disc, oversize bolts	28C	25
181 & 182	Cancelled by mod. 1018†	—		226	Inner exhaust cone, cracks	31	3
183 to 200	_____	Goblin		227	Diffuser casing, oversize diffuser-bolt dowels	28E	6
201	Impeller, undersize centre-shaft spigot	28A	26	228 & 229	Not issued by D.O.	—	
202	Cancelled by T.R.222	—		230	Cancelled by T.R. 227	—	
203	Cancelled by T.R.269	—		231 & 232	Cancelled by T.R.338	—	
204	Centre shaft, damaged impeller-spigot recess	28B	8	233	Cancelled by T.R.376	—	

† Mod. 1018 cancelled repair schemes for type 4 flame tubes

Revised by Amendment No. 135
November, 1957

<i>T.R. No.</i>	<i>Title</i>	<i>Refer to Chapter Page</i>	<i>T.R. No.</i>	<i>Title</i>	<i>Refer to Chapter Page</i>
234	Diffuser casing, oversize cooling-ring insert	28E 13	265	Cancelled. Applicable to pre-mod. 794 rear covers only	—
235	Cancelled by T.R.234	—	266	Turbine disc, oversize bolts	28C 26
236	Cancelled by T.R.307	—	267	Cancelled by T.R.268	—
237	Cancelled by mod. 1018†	—	268	Diffuser casing, rear cover, damaged stud holes	28E 2
238	Not issued by D.O.	—	269	—————	Ghost Fifty
239	Cancelled by T.R.283	—	270	Centre housing, damaged stud holes	28D 4
240	Cancelled by T.R.284	—	271	—————	Ghost Fifty
241	Exhaust cone fairings No. 1 and 2, cracks	31 3	272	Expansion chamber, drain-boss stud holes	30 12
242	Inner exhaust cone, front diaphragm, renewal	31 5	273	Expansion chamber locating plug stud holes	30 12
243	Cancelled by T.R.267	—	274	Cancelled by T.R.275	—
244	Cancelled by T.R.268	—	275	Diffuser casing, deflector ports, damaged stud holes	28E 9
245 & 246	Cancelled by mod. 1018†	—	276	Cancelled by mod. 1018†	—
247	Nozzle ring, discharge nozzle, cracks	28F 4	277	Diffuser casing, rear cover, damaged stud holes	28E 10
248	Not issued by D.O.	—	278	Cancelled by T.R.277	—
249	—————	Ghost Fifty	279	Diffuser casing, rear cover, damaged stud holes	28E 10
250	Diffuser casing, rear cover, securing and mounting bolts	28E 4	280	Cancelled by T.R.279	—
251	Not issued by D.O.	—	281	Diffuser casing, damaged stud holes	28E 16
252	Turbine disc, eccentric nut holes	28C 27	282	Cancelled by T.R.281	—
253	Impeller, oversize centre-shaft studs	28A 23	283	—————	Ghost Fifty
254	Impeller, oversize pivot studs	28A 24	284	Air-intake, damaged $\frac{1}{4}$ in. B.S.F. stud holes	28D 3
255	Cancelled by T.R.274	—	285	Not issued by D.O.	—
256	Cancelled by T.R.275	—	286	Cancelled by T.R.287	—
257	Cancelled by T.R.271	—	287	Expansion chamber, loose interconnector bush	30 13
258	Cancelled by T.R.272	—	288	Cancelled by T.R.289	—
259	Cancelled by T.R.273	—	289	Expansion chamber outlet-flange stud holes	30 13
260	Air-intake, centre housing bore	28D 13	290	—————	Ghost Fifty
261 & 262	—————	Ghost Fifty	291	Cancelled by mod. 1018†	—
263	Cancelled by mod. 1018†	—			
264	Cancelled by T.R.357	—			

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<i>T.R. No.</i>	<i>Title</i>	<i>Refer to Chapter Page</i>		<i>T.R. No.</i>	<i>Title</i>	<i>Refer to Chapter Page</i>	
292 & 293	—————	Ghost Fifty		321	Not issued by D.O.	—	
294	Cancelled by T.R.263	—		322	Diffuser casing, deflector assemblies, damaged threads	28E	7
295	Cancelled by T.R.330	—		323 to 326	—————	Ghost Fifty	
296	—————	Ghost Fifty		327	Rear bearing housing, damaged threads	28B	9
297	Inner exhaust cone, front flange	31	8	328 to 332	—————	Ghost Fifty	
298	Cancelled by T.R.391	—		333	Outer exhaust cone, bosses	31	7
299	Diffuser casing, sealing plate, inspection cover stud holes	28E	14	334	Oil sump, dowels	29	2
300	Expansion chamber, pressure connection	30	14	335 to 337	—————	Ghost Fifty	
301	Cancelled by T.R.300	—		338	Diffuser vane tips, repair	28E	12
302 & 303	—————	Ghost Fifty		339	Type 4 and 5 expansion chamber, locating plates	30	15
304	Propelling nozzle spacer-strips	31	3	340	Turbine disc, corrosion	28C	28
305	Expansion chamber, igniter plug stud holes	30	14	341	Discharge nozzle assembly, joint ring replacement	28F	10
306	Cancelled by T.R.305	—		342	Oil pump cover, stud holes	29	3
307	Oil sump, rear bearing connection	29	2	343	Inner nozzle ring, diaphragm recess	28F	13
308	Diffuser casing and rear cover mounting and securing bolts	28E	5	344	Type 5 flame tube and head, cracks	30	6
309	Cancelled by T.R.233	—		345	Type 5 flame tube and head, worn inter-connector sleeve	30	16
310	Cancelled by T.R.312	—		346	Type 5 flame tube and head, loose or damaged rivets	30	18
311	Exhaust cone fairings, bosses	31	4	347	Type 5 flame tube head, replacement	30	11
312	Centre casing, front and rear cones, rear flange	28F	12	348	—————	Ghost Fifty	
313	Diffuser casing, rear cover, air annulus cover studs	28E	11	349	Not issued by D.O.	—	
314	Expansion chamber, inlet flange face	30	23	350	Alternator and generator gear drive shaft (Pre-mod. 776), wear	28D	3
315	Diffuser casing, deflector ports	28E	8	351	Deflector assemblies, corrosion	28E	7
316 & 317	Cancelled by mod. 1018†	—		352	Type 5 flame tube and head, rear section, replacement	30	10
318 & 319	—————	Ghost Fifty		353	Type 5 flame tube and head, locating ring, wear	30	18
320	Fitment of protection plates to anti-resonance holes introduced by mod. 336 Part 2.	28D	13	354	Type 5 flame tube and head, mounting pads, wear	Not issued	

† See page 9

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TURBINE REPAIR SCHEMES

de Havilland
Ghost Forty-eight

<i>T.R. No.</i>	<i>Title</i>	<i>Refer to Chapter Page</i>	<i>T.R. No.</i>	<i>Title</i>	<i>Refer to Chapter Page</i>
355 & 356	_____	Ghost Fifty	383	Generator drive casing, oil feed union ferrule (Pre-mod. 860)	<i>Not issued</i>
357	Impeller, damaged vanes	28A 2	384	Diffuser casing vane, damaged stud holes	28E 15
358 to 363	_____	Ghost Fifty	385	Outer exhaust cone, cracked holes in front flange	<i>Not issued</i>
364	Air compressor housing, oil feed hole	28D 12	386	Cancelled by T.R.393	—
365	_____	Ghost Fifty	387	Diffuser casing, damaged vane tips (Mod. 1105)	<i>Not issued</i>
366	Diffuser casing or rear cover, renewal of either component	28E 17	388	Flame tube and head	30 23
367	Oil pump, spindle bores	<i>Not issued</i>	389	Centre casing	28F 14
368	Not issued by D.O.	—	390	Planet gear carrier cover, oversize dowel	<i>Not issued</i>
369*	Expansion chamber, loose interconnector bush (Mod. 1019)	30 20	391	Cancelled by T.R.413	—
370*	Type 5 flame tube and head, worn inter-connector sleeve (Mod. 1019)	30 20	392	Not issued by D.O.	—
371	Deflector assemblies, radii of side plates (Pre-mod. 843)	28E 17	393	Bottom wheelcase, over- size bearing bushes	<i>Not issued</i>
372	Cancelled by mod. 1147	30 22	394	Oil sump, union for prim- ing connection, damaged threads	29 9
373	Expansion chamber, pres- sure connection, threads or ferrules stripped or damaged	30 22	395	Cancelled by T.R.414	—
374	Diffuser casing, sealing plate, damaged face	28E 16	396	Impeller, local application of cold setting lacquer	28A 25
375	Fireguard, damaged drain trough pipe	31 13	397	Type 5 flame tube and head, damaged suspension sleeve	<i>Not issued</i>
376	Turbine shroud, distorted bore	28F 8	398	Starter gear casing, dam- aged threads	28D 2
377	_____	Ghost Fifty	399	Centre housing, worn or damaged sleeve	<i>Not issued</i>
378	Outer exhaust cone, stripped or damaged bosses	31 9	400*	Bottom wheelcase, over- size bearing bushes	—
379	Fireguard, repair to thermocouple flange and cracks in front flange	31 12	401	Diffuser casing and rear cover, damaged holes for mounting and securing bolts	<i>Not issued</i>
380	Sealing plate, cracked bore	28E 15	402	Diffuser casing, damaged vane tips	<i>Not issued</i>
381	Exhaust cone assembly, fairings, damaged thread in bosses	31 11	403	Centre and extension shafts, oversize bolts	28B 10
382	Front baffle assembly, cracked stiffeners	31 10			

* Limited to certain licenses only

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<i>T.R. No.</i>	<i>Title</i>	<i>Refer to Chapter Page</i>	<i>T.R. No.</i>	<i>Title</i>	<i>Refer to Chapter Page</i>
404	Top wheelcase, bottom bearing friction ring cover assembly, mod. 1132, worn or damaged friction ring	<i>Not issued</i>	414	Outer cone, cracked or damaged front flange (mod. 1140)	<i>Not issued</i>
405	Outer exhaust cone, damaged fireguard supports	<i>Not issued</i>	415	Diffuser casing	28E 17
406	Sealing plate, pre-mod. 897, oversize dowels	<i>Not issued</i>	416	Nozzle shroud, distorted bore	28F 9
407	Diffuser casing, damaged identification-plate securing holes	Not applicable to civil engines	417	Turbine shroud, front flange, enlarged or elongated bolt holes	28F 13
408	Diffuser casing, damaged threads	<i>Not issued</i>	418	Expansion chamber	30 24
409	Fireguard, worn locating holes	<i>Not issued</i>	419	Not issued by D.O.	—
410	Deflector, damaged vanes	<i>Not issued</i>	420	Combustion chamber, foul with centre casing	<i>Not issued</i>
411	Sightglass housing, taper pin	<i>Not issued</i>	421	Oil pump, spindle bores	29 4
412	Flame tubes, worn mounting pads	<i>Not issued</i>	501 to 602	—————	Goblin
413	Outer cone, cracked or damaged front flange (Pre-mod. 1140)	<i>Not issued</i>	71500 to 98935	—————	For incorporation of modifications only

MISCELLANEOUS REPAIRS AND PROCESSES

	<i>Refer to</i> <i>Chapter Page</i>			<i>Refer to</i> <i>Chapter Page</i>	
Argon-arc welding	30	3	Oil pump test, T.T.S.26	29	6
Chromate of magnesium	32	18	Oxy-acetylene welding	30	5
Control of resistance welding and hot riveting	32	4	Preparation for welding	30	3
			Re-anodising impeller	28A	12
"Cross" wire inserts	32	2	Repair identification	32	3
Electric hot riveting, acid etch test	30	9	Rockhard, impeller	28A	17
Flash chromium	32	16	Soft nickel	32	17
Hard chromium	32	3	Stress relieving after welding	30	5
Interpretation of machining symbols	22	15	Stripping nickel plate	32	16
			Vapour blasting impeller	28A	9
Material Identification and Fusion Welding Code	30	5	Vibro etch marking	30	2

INTERPRETATION OF WORKING DRAWINGS AND SYMBOLS

The repair schemes contained in chapters 28 to 31 include reproductions of the engine manufacturers' gas turbine repair (T.R.) drawings. These drawings conform to the manufacturers' standard drawing office practice, and, consequently, include symbols and other data with which the users of this handbook may not be familiar. Whilst these repair drawings, and similar drawings elsewhere in this handbook, conform closely to British Standard (B.S.) 308, correct interpretation of the drawings will be obtained only by reference to the following paragraphs which describe the manufacturers' drawing office practice, so far as it affects these drawings.

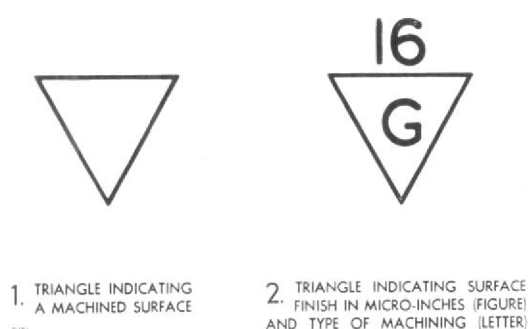


Fig. 3. Machining symbols.

Machining symbols

When the symbols illustrated in Fig. 3 are found on working drawings, they should be interpreted as follows.

1 on Fig. 3. A triangle of this type denotes that the surface to which it is adjacent, or which is indicated by a leader line from the triangle, is a machined surface.

2 on Fig. 3. A figure outside the triangle (in this instance 16) expresses the surface finish in micro-inches. A letter, or group of letters, within the triangle (in this instance G) indicates the type of machining.

Below is a list of the letters used to denote types of machining, and their interpretation.

BF = Buff	BU = Burnish
D = Drill	DB = Diamond Bore
DT = Diamond Turn	G = Grind
H = Hone	LP = Lap
LI = Linnish	PO = Polish
R = Ream	SP = Spot Face

Positional symbols

The symbols illustrated in Fig. 4 refer to the limits for position, concentricity, or symmetry.

1 on Fig. 4. This symbol indicates a datum surface, and the letter (in this case A) indicates the group to which all other positional symbols containing the same letter belong. It will be appreciated that more than one datum surface may be found on a single working drawing, and each additional datum surface will be indicated by the basic symbol, but each individual datum surface will have a different identification letter. Below the identification letter in a datum symbol is the figure "O", which is constant in all datum symbols and serves to emphasise that the surface indicated is the datum to which all symbols with the same identification letter must be related, see following sub-paragraph.

2 on Fig. 4. This symbol indicates a machined surface whose position, concentricity, or symmetry, is governed by a datum surface. The letter within the symbol (in this case A) indicates the letter group to which the symbol belongs, and thereby the datum surface to which the machined surface is related. The figure within the symbol (in this case .3) which is below the group letter, expresses in thousandths (0.001) of an inch the degree of accuracy required, between the actual position of a surface, and its theoretical position, relative to the datum surface in that letter group. An example of the application of this tolerance is illustrated in Fig. 5. It is essential that these tolerances on position, concentricity or symmetry are not confused with dimensional tolerances. The first are concerned with relative position of planes, or curved surfaces one to the other, whilst the

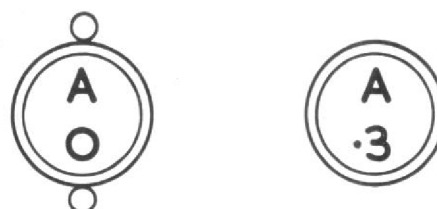


Fig. 4. Positional symbols.

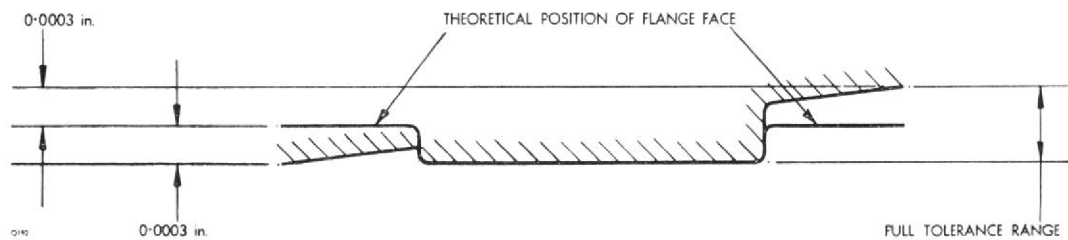


Fig. 5. Diagram showing tolerance between actual and theoretical position of a surface.

B.A. FORM SCREW THREAD, STANDARDS FOR INTERNAL THREADS

B.A. Designation Number	Pitch	Major (Full) Dia.			Effective Dia.			Minor (Core) Dia.		
		Min.	Tol.	Max.	Min.	Tol.	Max.	Min.	Tol.	Max.
	mm.	mm.	mm.	mm.	mm.	mm.	mm.	mm.	mm.	mm.
2	0.810	4.700	0.170	4.870	4.215	0.085	4.300	3.730	0.285	4.015
4	0.660	3.600	0.145	3.745	3.205	0.075	3.280	2.810	0.230	3.040
6	0.530	2.800	0.125	2.925	2.480	0.060	2.540	2.160	0.185	2.345

Approximate Inch Equivalents

B.A. Designation Number	Threads per inch	Major (Full) Dia.			Effective Dia.			Minor (Core) Dia.		
		Min.	Tol.	Max.	Min.	Tol.	Max.	Min.	Tol.	Max.
2	31.4	0.1850	0.0067	0.1917	0.1659	0.0033	0.1692	0.1468	0.0112	0.1580
4	38.5	0.1417	0.0057	0.1474	0.1262	0.0030	0.1292	0.1106	0.0091	0.1197
6	47.9	0.1102	0.0049	0.1151	0.0976	0.0024	0.1000	0.0850	0.0073	0.0923

B.A. FORM SCREW THREAD, STANDARDS FOR EXTERNAL THREADS

B.A. Designation Number	Pitch	Major (Full) Dia.			Effective Dia.			Minor Dia.		
		Max.	Tol.	Min.	Max.	Tol.	Min.	Max.	Tol.	Min.
	mm.	mm.	mm.	mm.	mm.	mm.	mm.	mm.	mm.	mm.
2	0.810	4.700	0.120	4.580	4.215	0.085	4.130	3.730	0.170	3.560
4	0.660	3.600	0.100	3.500	3.205	0.075	3.130	2.810	0.145	2.665
6	0.530	2.800	0.080	2.720	2.480	0.060	2.420	2.160	0.125	2.035

Approximate Inch Equivalents

B.A. Designation Number	Threads per inch	Major (Full) Dia.			Effective Dia.			Minor Dia.		
		Max.	Tol.	Min.	Max.	Tol.	Min.	Max.	Tol.	Min.
2	31.4	0.1850	0.0047	0.1803	0.1659	0.0033	0.1626	0.1468	0.0067	0.1401
4	38.5	0.1417	0.0039	0.1378	0.1262	0.0030	0.1232	0.1106	0.0057	0.1049
6	47.9	0.1102	0.0031	0.1071	0.0976	0.0024	0.0952	0.0850	0.0049	0.0801

latter (the dimensional tolerances) are concerned with the distance between surfaces or points.

Fig. 6 illustrates various applications of the symbol system for indicating limits on position, concentricity, or symmetry. In the cross section (the left-hand view), the symbol "AO" (which has been marked with an asterisk*) indicates that the 1.0000 + .0006 inch diameter bore is the primary datum, to which all symbols containing the same letter are related. The symbol "AO" (which has been marked with a dagger†) indicates that the 1.500 + .0012 inch diameter counterbore must be perfectly concentric with the 1.0000 inch diameter bore, the figure "O" in this instance indicating that there is no tolerance. The symbol "A1" indicates that the 1.4369 inch thread must be concentric with the 1.0000 inch bore within 0.001 inch, and the symbol "A5" indicates that the 1.4369 - .0006 inch diameter spigot must be concentric with the datum bore within 0.005 inch. Similarly, the symbol "A.3" indicates that the under surface of the flange must be square to the datum bore within 0.0003 inch. In addition to its relationship to the datum bore, the 1.4369 inch spigot is itself a datum surface, as indicated by the symbol "BO". The symbol "B6" in the right-hand view, indicates that the 2.3125 inch diameter circle on which the centres of the $\frac{1}{16}$ inch clearance holes lie, must be concentric to within .006 inch of the centre of the 1.4369 inch spigot. Also, the centres of these $\frac{1}{16}$ inch holes are limited to within a circle of 0.006 inch radius, the centre of which is the theoretical position of the $\frac{1}{16}$ inch holes on the 2.3125 inch diameter.

Screw threads

Where screw threads are indicated on working drawings, the thread notes will have a standard form, giving the following particulars.

1. In all cases of medium fit standard threads, only the basic size and thread form will be given. In these instances, the tolerances permitted are those laid down for medium fit threads in British Standard 84.
2. In cases where a close fit tolerance is required, the thread notes will state the basic size and thread form, followed by a note in brackets (B.S. 84 close fit). This does not apply to B.A. form threads for which see the following sub-paragraph.
3. For B.A. form threads the manufacturer has laid down standards (refer to facing page) which must be adhered to strictly.
4. Where special threads are used, the following particulars will be given in the thread notes.

Basic Thread Size
Threads per inch (T.P.I.)
Thread Form
Effective Diameter
Full Diameter (Truncated) for external threads
Minor Diameter (Truncated) for internal threads

Dimensional tolerances

Where specific tolerances are not given on a working drawing, the following general tolerances apply.

- (1) Linear .. $\pm .010$ inch
- (2) Angular .. ± 2 degrees
- (3) Drilled Holes + .010 inch
- .002 inch
- (4) Screw threads see previous paragraph

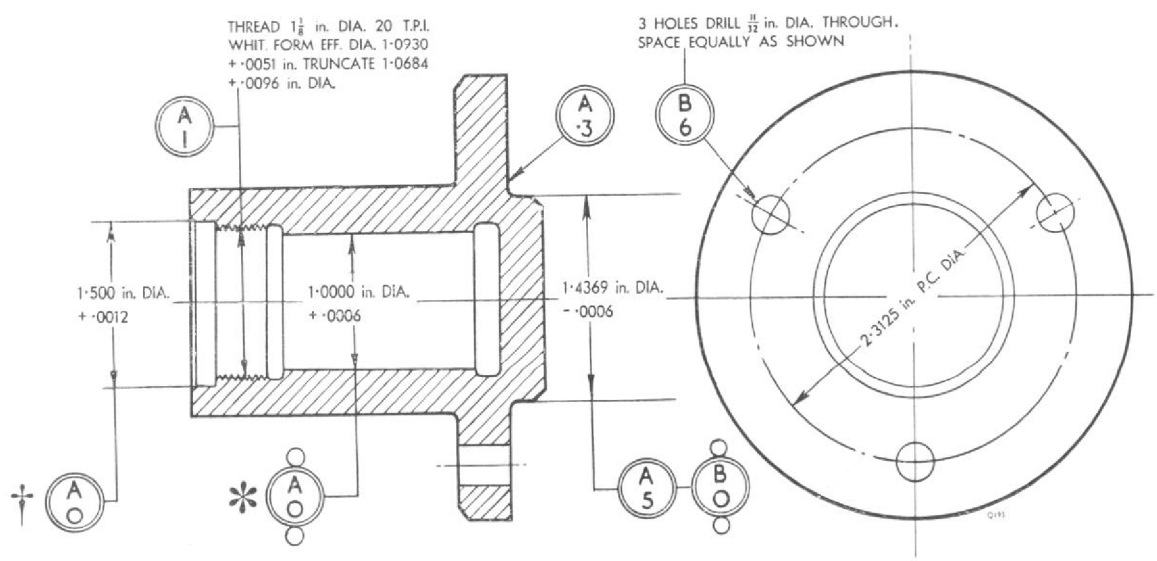


Fig. 6. Diagram showing various applications of the symbol system.

REPAIRS FOR WHICH WORKING INSTRUCTIONS ARE NOT PROVIDED

There are a number of repairs for which detailed working instructions have not been provided. Broadly speaking these repairs can be classified as coming within one of the following categories.

1. The tooling is of such a nature that, normally, it is available in the engine manufacturer's works only.
2. The tools are not available for sale.
3. The repair is intended for the salvage of new parts damaged during initial manufacture.

Although these repairs cannot, normally, be carried out by operators, knowledge of their existence and general effect on the components concerned can be invaluable for the following reasons.

1. To enable an operator to know what has been done to a component on which a repair (T.R.) number is marked, or in respect of which such information is contained in the relevant documents.

2. To inform an operator that such repair is possible, and so avoid the needless destruction of repairable components (even though they may be repairable only by the manufacturers), or the pointless retention of components which are beyond repair.

3. To enable the correct mating replacement components to be obtained where one result of the repair is loss of interchangeability; e.g., where repair to a damaged spigot results in an undersize spigot which necessitates a corresponding undersize counterbore in the mating component.

4. To enable an operator, in an emergency, on his own responsibility, to carry out the repair.

Chapters 28 to 32 contain brief descriptive information covering repairs in the foregoing categories. These brief descriptions do not pretend to provide complete working instructions and are identified by the words 'For Information Only' immediately below the heading.

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