

CHAP 2 AIRFRAME SP 108 AL 4 SHEET 1 OF 10	SERVICING PROCEDURE F53 T55	BAC F53 & T55 (SA) 5A3A Section 1 2nd Edition
Undercarriage Emergency Lowering	AFSC 43151 43171 42152 42172	TIME EST
Safety and Servicing Notes are to be complied with throughout the work detailed on this card.		
SPECIAL TOOLS AND EQUIPMENT		
Pitot static test set Mk3 (6C/1042139). Tyre inflation rig (4G/1050542). Spring balance (1A/4232170). Canopy seal guards F53 (26DK/95129 - Left). Canopy seal guards F53 (26DK/95130 - Right). Canopy seal guards T55 (26DK/95252 - Left). Canopy seal guards T55 (26DK/95253 - Right).		
NOTE: During this servicing procedure No.2 engine hatch or jury strut must be fitted at all times.		
<u>43151</u>		
1. PREPARATION		
1.1 Aircraft.	(i) Raise on jacks until all wheels are clear of ground. (ii) Trestle at Frame 59.	
<u>43151/42152</u>		
2. PREPARATION		
2.1 Ground air charging/ release connexion (Access panel 63P(left)).	Remove blank.	
2.2 Services system hydraulic pressure.	Release pressure by operating brake lever.	
2.3 Hydraulic test trolleys.	(i) Prime. (ii) Bleed.	
2.4 No.1 services ground test connexions (Access panel 45P(left)).	Connect hydraulic test trolleys.	
2.5 No.1 controls ground test connexions (Access panel 45P(left)).	Connect hydraulic test trolleys.	
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2. PREPARATION (Contd)

2.6 Hydraulic reservoirs. Replenish (SP 603(AF)).

2.7 Tyre inflation rig. (4G/1050542).
(i) Connect to ground air charging/release connexion (Access panel 63P(left)).
(ii) Set rig to deliver a pressure of between 16 and 18 lbf/in².

2.8 Canopy seal guards. Ensure fitted.

2.9 Slave battery. Fit.

2.10 Battery isolating switch. Set to ON.

2.11 Pitot/static test set. Connect.

2.12 Armament safety break.
(i) Remove from lanyard.
(ii) Stow lanyard and flag in stowage pocket.

2.13 Undercarriage toggle switch. Ensure set to DOWN.

2.14 Undercarriage ground locks. Remove.

3. TESTING

3.1 Services system. Pressurize to 3000 lbf/in².

3.2 Nose undercarriage. Ensure aligned fore and aft.

3.3 Pitot/static system. Pressurize to an equivalent of 165Kt.

3.4 Airbrakes. Open.

3.5 Flaps. Lower.

3.6 Undercarriage toggle switch. Set to UP.

3.7 Undercarriage indicator lamps (F53: quantity 3, T55: quantity 6). Ensure extinguished.

3.8 Services system test trolley. Stop.

3.9 Pitot/static system. Release pressure.

Continued

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3. TESTING (Contd)

3.10 No.1 controls system. Pressurize to 3000 lbf/in².

3.11 Undercarriage emergency lowering handle (F53 - See Fig.1), (T55 - See Fig.2). (i) Set to down using spring balance 1A/5232170. (ii) Check max force required to pull lever is 20 lbs.

NOTE: Maximum force to operate release button is 5 lbs.

3.12 Undercarriage. Check time taken to lower and lock does not exceed 21 secs.

3.13 Undercarriage indicator lamps (F53: quantity 3, F55: quantity 6). (i) Ensure lit, RED whilst undercarriage is lowering. (ii) Ensure lit (GREEN) when undercarriage is locked down.

3.14 Controls system test trolley. Stop.

3.15 Undercarriage toggle switch. Set to DOWN.

4. RESETTING PROCEDURE

NOTE 1 : Sub-items 4.1 and 4.2 must be carried out simultaneously.

NOTE 2 : During Sub-item 4.1 the cable is to be tensioned by pulling aft of access panel 26P (left) F53, access panel 2OP (left) T55, to avoid kinking of cable at cockpit undercarriage emergency lowering handle.

4.1 Undercarriage emergency lowering handle. (i) Reset to unoperated position (pushed in and spring loaded plunger engaged). (ii) Fit indicator wire, 28 SWG enamelled copper wire (See Fig.3).

4.2 Emergency selector valve lever
Access panel 26P (left-F53),
Access panel 2OP (left-T55).

Reset to unoperated position.

4.3 Spring loaded cam (See Fig.4). Ensure ball race on end of selector lever is correctly positioned on cam.

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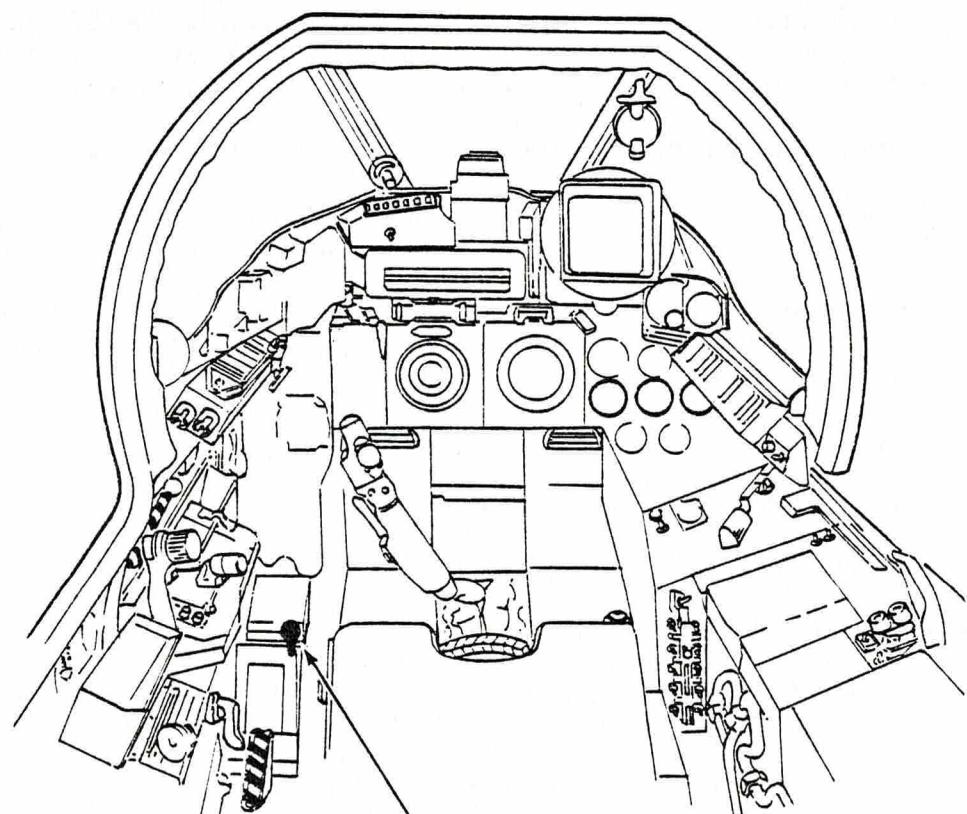
SERVICING PROCEDURE

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EMERGENCY
LOWER
SELECTOR
HANDLE

COCKPIT LAYOUT MK F53
FIGURE 1

Continued

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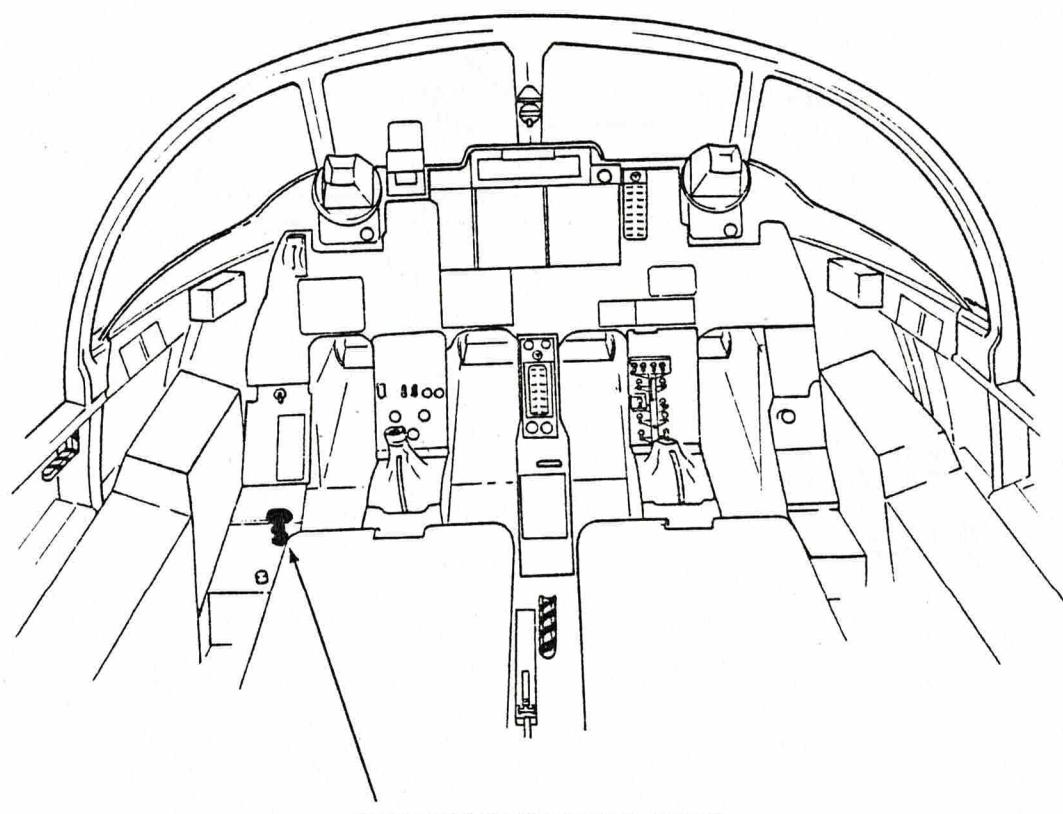
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COCKPIT LAYOUT MK T55
FIGURE 2

Continued Overleaf

SERVICING PROCEDURE INSPECTION STAGES DO NOT EXCLUDE ADDITIONAL INSPECTION STAGES
INCORPORATED AS NECESSARY IN MAINTENANCE CERTIFICATION DOCUMENTS

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SERVICING PROCEDURE

F53

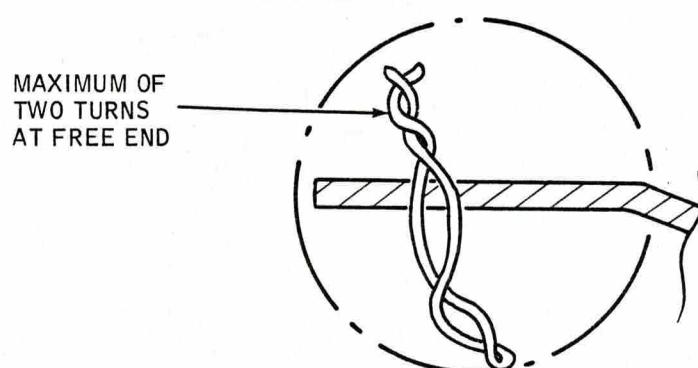
T55

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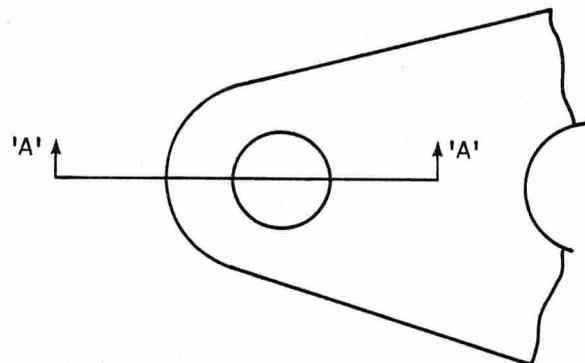
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TYPICAL SECTION
THROUGH ANCHOR
LUG SHOWING
METHOD OF
APPLICATION

SECTION A-A



EMERGENCY UNDERCARRIAGE HANDLE STANDARD APPLICATION
OF INDICATOR WIRES

FIGURE 3

Continued

SERVICING PROCEDURE INSPECTION STAGES DO NOT EXCLUDE ADDITIONAL INSPECTION STAGES
INCORPORATED AS NECESSARY IN MAINTENANCE CERTIFICATION DOCUMENTS

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4. RESETTING PROCEDURE (Contd)

4.4 Emergency selector valve lever. Using probe light and mirror ensure hard against rear end of slot.

4.5 Operating cable (See Fig. 4). (i) Ensure lying in guard channel.
(ii) Ensure red marker band on the cable is in alignment with red marker band on the inside of the cable guard.

4.6 Services system. Pressurize to 3000 lbf/in² to reset shuttle and relay valve.

4.7 Protection unit indicator rod. Ensure protruding at bottom of protection unit.

4.8 Airbrakes. Close.

4.9 Flaps. Raise.

NOTE: During sub-items 4.8 and 4.9 measure quantity of fluid emitted from No.1 services reservoir overboard drain using suitable container. The measured quantity should be approximately 5 pints.

4.10 Services system test trolley. Stop.

43171/42172 (INSPECTOR)

5. INSPECTION STAGE

5.1 Inspect. After component replacement or if the system has been disassembled.

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6. GENERAL

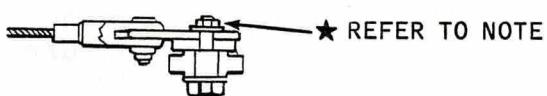
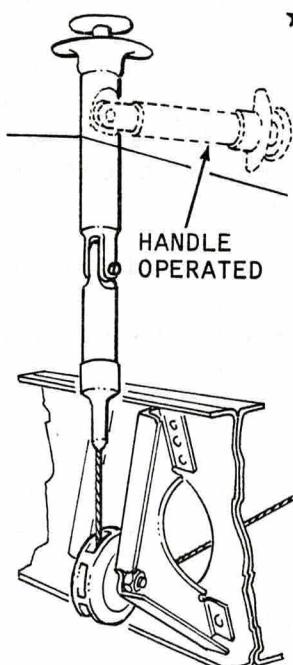
6.1 Services system. Pressurize to 3000 lbf/in².

6.2 Pitot/static system. Pressurize to an equivalent of more than 165 Kt.

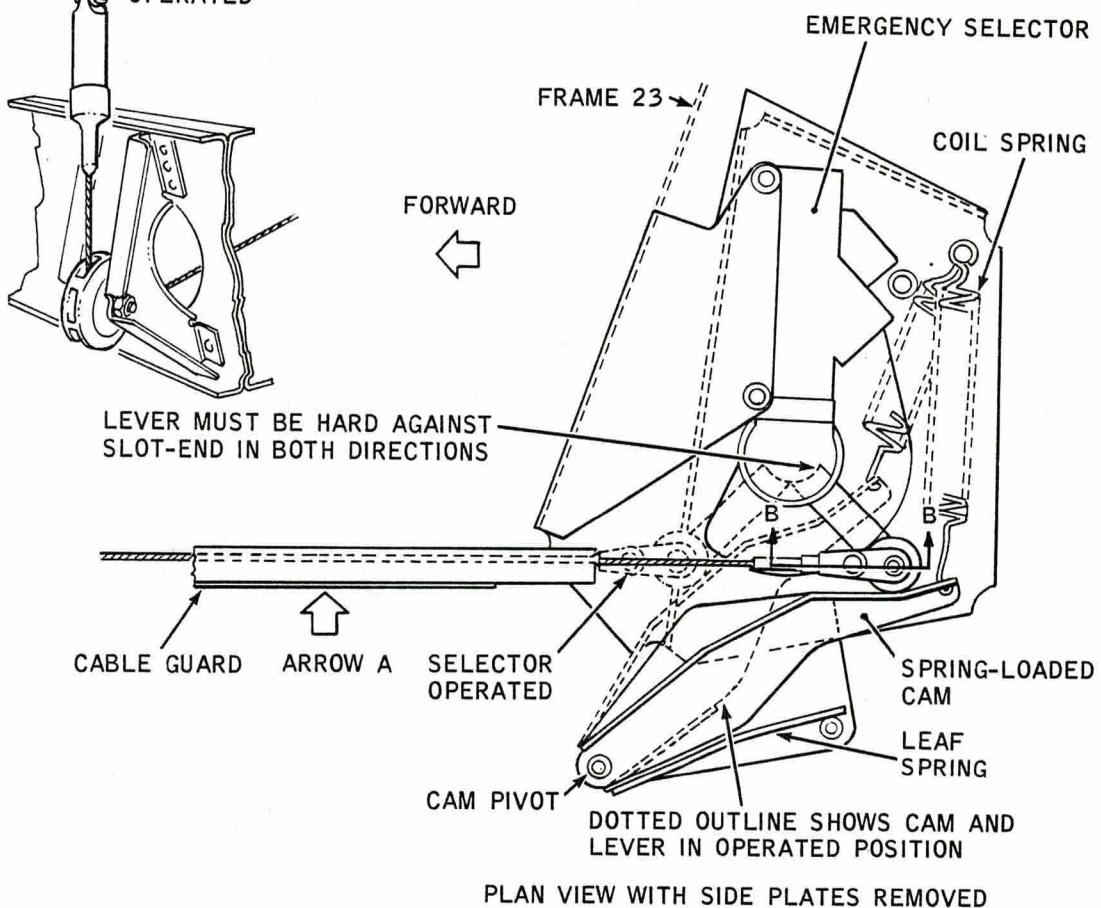
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★ NOTE: THE CABLE EYE-END MUST BE FREE TO PIVOT AFTER TIGHTENING THIS BOLT

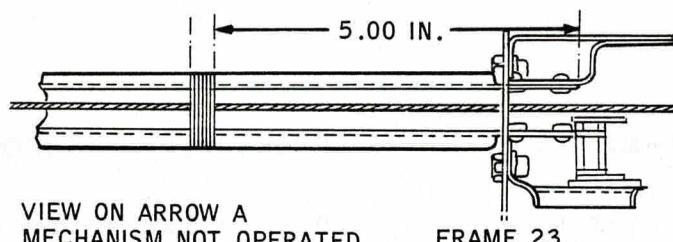


SECTION B-B



MARKER BAND
0.25 IN. WIDE ON
CABLE AND GUARD

TO ASSIST IN CHECKING THE POSITION OF THE LEVER
IT IS PERMISSIBLE TO PAINT MARKERS ON THE CABLE
AND GUARD AS SHOWN BELOW



ADJUSTMENT OF ALIGHTING GEAR EMERGENCY LOWERING CONTROL

FIGURE 4

Continued

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6. GENERAL (Contd)

6.3 Undercarriage. Carry out minimum of two full retractions and lowerings.

6.4 Services test trolley. Stop.

43171/42172 (INSPECTORS)

7. INDEPENDENT CHECKS.

NOTE: The independent checks detailed in this item are to ensure that the Hydel selector slide valve is correctly positioned and must be carried out personally by a specialist inspector who has not been directly concerned with the operations detailed in this servicing procedure.

7.1 Undercarriage toggle switch. Ensure set to DOWN.

7.2 Services system. Pressurize to 3000 lbf/in².

7.3 Undercarriage indicator lamps (green), (F53: quantity 3, T55: quantity 6). Ensure lit.

7.4 Undercarriage ground locks. Fit.

7.5 Aircraft job card. Endorse as follows:
Certified I have personally carried out the independent checks detailed in servicing procedure BAC F53 and T55 (SA), 5A3A Section 1 SP108 (AF).

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8. COMPLETION

8.1 Pitot/static test set. (i) Release pressure.
(ii) Disconnect.

8.2 Services system. Release pressure by operating brake lever.

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8. COMPLETION (Contd)

8.3 Battery isolating switch. Set to OFF.

8.4 Slave battery. Remove.

8.5 Ground equipment.

- (i) Operate to lower aircraft to ground.
- (ii) Remove from vicinity of aircraft.

8.6 Tyre inflation rig (4G/1050542).

- (i) Disconnect at ground air charging/release connexion.
- (ii) Fit blank and wirelock.

8.7 Hydraulic test trolleys.

- (i) Disconnect.
- (ii) Remove.

8.8 Hydraulic pump quick release connexions (Services and controls Access panel 45P (left)).

- (i) Reconnect to aircraft.
- (ii) Wirelock.

8.9 Hydraulic reservoirs. Replenish (SP 603(AF)).

8.10 Access panels. Refit.

8.11 Canopy seal guards. Remove.

8.12 Armament safety break.

- (i) Remove flag and lanyard from stowage.
- (ii) Reconnect lanyard.

NOTE: All wirelocking to be of 22 SWG stainless steel locking wire unless otherwise stated.

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