

CHAP. 4 AIRFRAME S.P. 414 A.L. 4 SHEET 1 OF 2	SERVICING PROCEDURE F53 T55	BAC F53 & T55 (SA) 5A3A Section 1 2nd Edition		
Left and Right Mainplane Top Skin Panel - Cracking		AFSC 43151 53450 53470		
Safety and Servicing Notes are to be complied with throughout the work detailed on this card.		TIME EST		
SPECIAL TOOLS AND EQUIPMENT Nil.		ASSOCIATED PROCEDURES		
NOTE: Application for Repair Schemes are to be submitted to British Aerospace, Saudi Technical Support Dept., Warton Lancashire, England quoting Repair Schemes Nos. SR0/20/40 and SRO/20/46.				
<u>43151</u>				
1. EXAMINATION				
1.1 Left and right mainplane panel PFCU's above aileron.	Examine, particularly for cracks.			
<u>53450</u>				
2. RECTIFICATION				
NOTE: Sub-item 2.1 is applicable only if not more than 3 cracks, not more than 1.0 in. long, are found in area above the two hydraulic pipe rubbing plates in Sub-item 1.1.				
2.1 Panel.	Where cracks appear (longer than 0.50 in.), drill 0.125 in. stop holes, unless the centre distance of holes from existing holes would be less than 0.50 in.			
NOTE: Sub-item 2.2 is applicable only if:				
(a) More than three cracks are found in Sub-item 1.1. (b) One or more cracks found in Sub-item 1.1 are longer than 1 in. (c) Cracks longer than 0.50 in. cannot be stop drilled in Sub-item 2.1.				
2.2 Panel.	Repair in accordance with Repair Manual, Part 2, Leaflet C, 1/14 and C 1/15.			
Continued Overleaf				

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S.P. 414 AL 1	F53 T55	5A3A Section 1
SHEET 2 OF 2		2nd Edition

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53470 (INSPECTOR)

3. INSPECTION STAGE

3.1 Inspect. Sub-item 2.1, if applicable.

53470 (QUALITY CONTROL)

4. INSPECTION STAGE

4.1 Inspect. Sub-item 2.2 if applicable.

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