

Chapter 15 NDI
Sheet 1
AL 5

MASTER SERVICING LIST
CANBERRA
2, 3, 4, 6, 7, 15, 17, 18, 19 & 22

AP101B-0400-5A1
Section 1
(2nd Ed)

Sched Ident No	Item	Operation	Frequency (Weeks)	Marks Applicable												
					B/F	T/R	A/F	P	M	MJ		Crit	StrInt			
<u>NOSE FUSELAGE - 02</u>																
010212	<u>Pressure Cabin</u> Navigators hatch supporting structure.	Servicing deleted by AL 4.														
150201	Navigators hatch supporting structure.	Examine and particularly for cracks. Technique No. Can/Eddy/17 (Introduced by SI/CAN/188C).	250	All						X	X					
<u>CENTRE FUSELAGE - 03</u>																
150301	<u>Main Structure</u> Centresection forgings that have not been changed since manufacture.	Examine. Technique No. Can/Ult/4D.	16 Weeks	2,3,4,6,7,15,17,18,19									(Out of Phase (Special Check).	C	Y	
			Heavy Landing or Excessive 'G'											Out of Phase (SP 804).		
		Examine. Technique No. Can/Ult/18. (Introduced by SI/Can/169).	16 Weeks	2,3,4,6,7,15,17,18,19										Out of Phase (Special Check)	C	Y
		Examine. Technique No. Can/Ult/5. (Introduce by SI/Can/108).	32 Weeks	2,3,4,6,7,15,17,18,19.										Out of Phase (Special Check)		Y
		Examine. Technique No. Can/Ult/22. (Introduced by SI/Can/178).	1000	2,3,4,6,7,15,17,18,19.											X	
			250 Hours when detailed in F700.										Out of Phase (SP 817)		Y	
SMS/77/201/23																

Sched Ident No	Item	Operation	Frequency	Marks Applicable	Servicing							Crit	Str Int	
					B/F	T/R	A/F	P	M	MJ				
150302	Centresection forgings that have been changed since manufacture.	Examine. Technique No. Can/Ult/5. (After Inspection Free Period 10 Years Expired).	32 Weeks	All										Y
		Examine. Technique No. Can/Ult/13. (After Inspection Free Period 10 Years Expired).	16 Weeks										C	Y
		Heavy Landing or Excessive 'G'		All										
		Examine. Technique No. Can/Ult/18. (Introduced by SI/Can/169). (After Inspection Free Period 10 Years Expired).	16 Weeks	All									C	Y
		Examine. Technique No. Can/Ult/22. (Introduced by SI/Can/178).	1000	All							X			
			250 Hours when detailed in F700	All										Y
150303	Longerons between Frames 21 and 31. (Pre Mod 5046 or 5063).	Examine. Technique No. Can/Bdd/14C and Technique No. Can/Ult/26A (Introduced by SI/Can/187C).	32 Weeks	All										Y
<u>TAIL UNIT - 05</u>														
	<u>Variable Incidence Tailplane</u>													
150501	Top spar boom.	Examine. Technique No. Can/Ult/27A. (Introduced by SI/Can/184A).	16 Weeks	All										Y
SM/77/20th/5A											Continued			

Sched Ident No	Item	Operation	Frequency	Marks Applicable	Servicing									
					B/F	T/R	A/F	P	M	M)		Crit	Str Int	
150602	Mainplane upper and lower spar booms.	Examine. Technique No. Can/Ult/10.	250	All						X	X			Y
150603	Inner mainplane diaphragm. (Pre Mod 3156 unrepaired). (Pre Mod 4706).	Examine. Technique No. Can/X.Rad/3A. (Introduced by SI/Can/136A).		All	Initially at 20 Fatigue Index Units and subsequently at increments of 1 Fatigue Index Unit and whenever flying limitations exceeded. Out of Phase (SP803).									
150604	Inner mainplane diaphragm. (Pre Mod 3156 Post repair scheme AP101B-0400-6 Leaflet C3/17). (Pre Mod 4706).	Examine. Technique No. Can/X.Rad/3A. (Introduced by SI/Can/136A).		All	Initially at 20 Fatigue Index Units since repair and subsequently at increments of 1 Fatigue Index Unit and whenever flying limitations exceeded. Out of Phase (SP803).									
150605	Inner mainplane diaphragm. (Pre Mod 3156 Post repair scheme AP101B-0400-6 Leaflet C3/19). (Pre Mod 4706).	Examine. Technique No. Can/X.Rad/3A. (Introduced by SI/Can/136A).		All	Initially at 40 Fatigue Index Units since repair and subsequently at increments of 1 Fatigue Index Unit and whenever flying limitations exceeded. Out of Phase (SP803).									
150606	Inner mainplane diaphragm. (Post Mod Repair Scheme ASD/653). (Pre Mod 4706).	Examine. Technique No. Can/X.Rad/5. (Introduced by SI/Can/136A).		All	Initially at 100 Flying hours since repair and subsequently at 150 and 200 flying hours since repair and whenever flying limitations are exceeded. Out of Phase (SP803). Note: Repair Scheme ASD/653 has life of 250 flying hours only.									
150607	Inner mainplane diaphragm. (Post Mod 3156 Unrepaired). (Pre Mod 4706).	Examine. Technique No. Can/X.Rad/3A. (Introduced by SI/Can/136A).		All	Initially at 15 Fatigue Index Units since embodiment of modification and subsequently at increments of 1 Fatigue Index Unit and whenever flying limitations exceeded. Out of Phase (SP803).									
SMS/77/88/23A/STC					Continued									

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