

## Chapter 34

### ACTUATORS, PLESSEY, CUB PUMA SERIES

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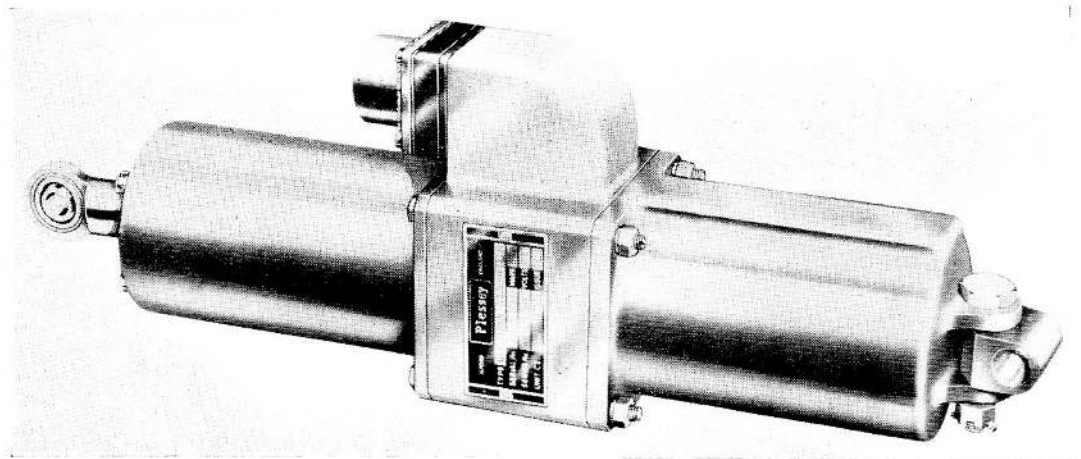


Fig. 1. Typical Cub Puma actuator

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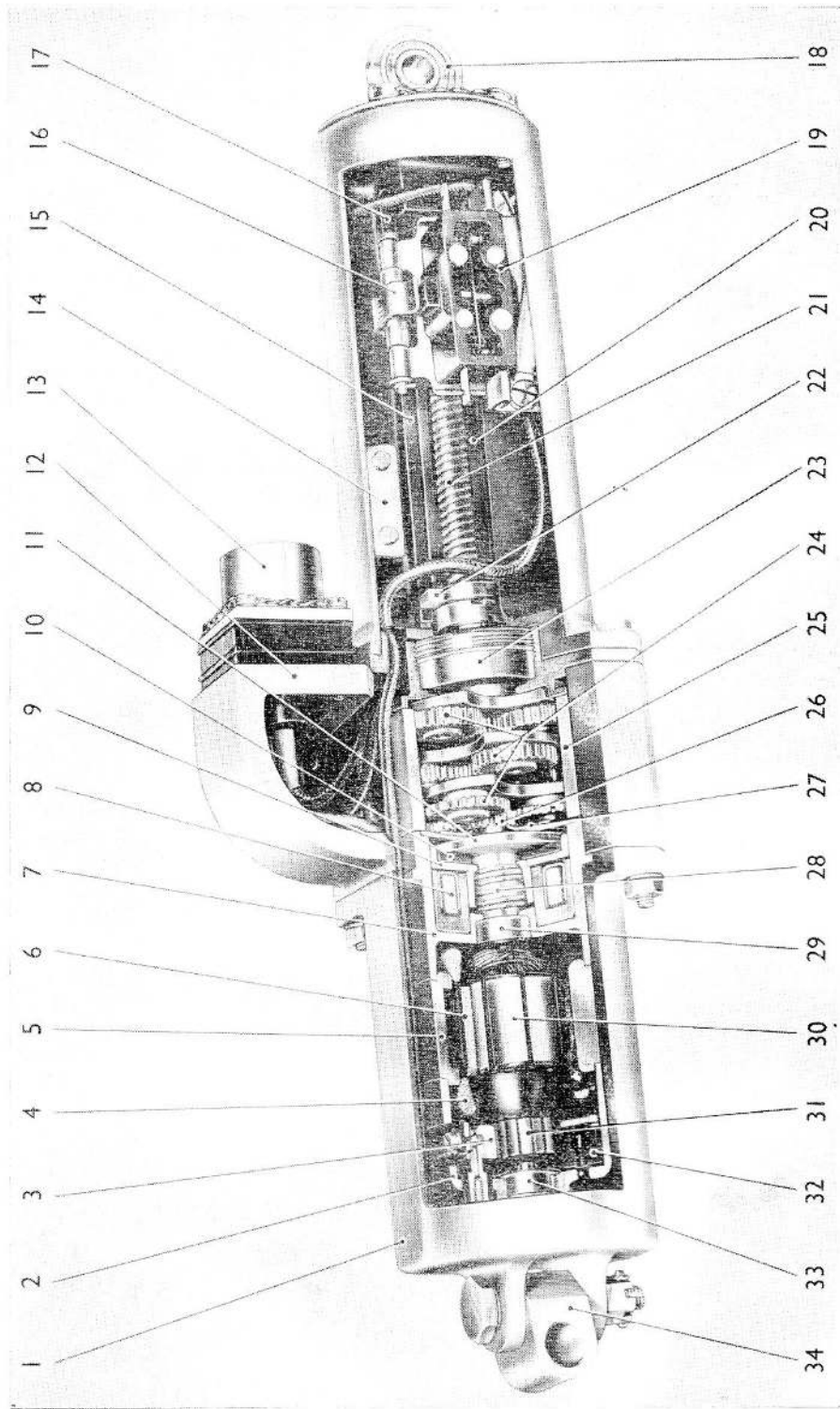


Fig. 2. Sectional view of actuator

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KEY TO FIG. 2  
SECTIONAL VIEW OF ACTUATOR

- 1 MOTOR HOUSING
- 2 COMMUTATOR END PLATE
- 3 BRUSH ROCKER ASSEMBLY
- 4 FIELD COIL WINDINGS
- 5 YOKE
- 6 POLE-PIECE
- 7 BRAKE HOUSING
- 8 BRAKE COIL
- 9 BRAKE SHOE
- 10 FRICTION DISC
- 11 FRICTION PLATE
- 12 GEARBOX HOUSING
- 13 PLUG ASSEMBLY
- 14 GUIDE PLATE ASSEMBLY
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- 30 WOUND ARMATURE
- 31 COMMUTATOR
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- 33 COMMUTATOR END BEARING
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### Introduction

1. Plessey linear actuators of the Cub Puma series follow the general design described in this Chapter; the machine illustrated, the CZ64250, is typical of the series, and specific details of individual actuators will be found in appendices to this chapter. They vary in such details as the working load, the type of end fittings and the length and time of plunger travel.

2. The actuators of the Cub Puma series are designed to provide remotely controlled linear motion against either compressive or tensile loads. Motive power is provided by a small reversible fractional horse power motor, the drive shaft of which engages a train of gears. Snap action limit switches are fitted; these automatically switch off the motor supply when the actuator reaches the end of its travel.

### DESCRIPTION

#### Motor

3. The motor (fig. 2) is a reversible split-field series type with an integral electromagnetic brake. The motor is located by a spigot in the gear housing; the motor pinion is part of an assembly and is pin-driven by the armature shaft. The armature is dynamically balanced to eliminate vibration.

#### Gearbox

4. The gearbox assembly (fig. 2) comprises the main housing (enclosing the toothed annulus) with the three-stage epicyclic gear train. The lead screw is extended into the gearbox, where it is supported in a bearing. One stage of the gearing is supported on a carrier integral with the lead screw extension; the other two gear stage carriers, for support and bearing purposes, are freely located on the lead screw extension, where they are retained by a circlip.

5. Each of the planet gears has two small holes for lubrication of the gear support pins, and from the motor end each gear stage has increased tooth face width. The annulus for the epicyclic gears is keyed and positioned in the gearbox housing.

#### Plunger

6. The plunger assembly comprises a chromium finished tube and eye end, screwed and pinned together. At the gearbox end is fitted the detachable torque reaction and switch operating key (plunger ears) with locking washer and ring nut.

#### Plunger housing

7. The plunger housing, when assembled to the unit, encloses the lead screw and plunger. It is bolted to the gearbox by its end flange plate which registers against the lead screw bearing housing. To minimize wear, the chromium plated plungers run in phosphor bronze bushes.

#### Limit switches

8. These are micro snap-action lever-operated type, on which the moulding of

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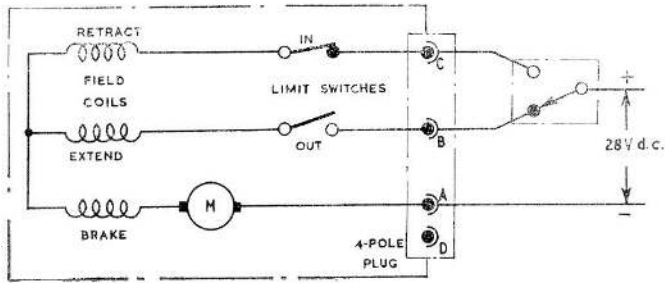


Fig. 3. Circuit diagram

the switch housing completely shrouds the contacts and micro-break mechanism. Two electrical connections are made to each switch; these are screwed to the threaded lugs located on each side of the moulding.

9 One arm of the switch-operating lever depresses the switch button when the other arm is actuated by the plunger ear as the plunger moves through its stroke.

### INSTALLATION

10. Before coupling the actuator to the installation, it must be ascertained that all the links and joints, bearing pivots, etc., of the component being actuated are free to move.

11. The installation centres must be checked against the relevant installation

drawing for the actuator concerned, and the fixing bolts should be passed through the location holes to make certain that a free fit results.

12. To install the actuator, first couple the rear end fitting to the fixed fork end of the installation. The unit should be rocked slightly to check the fitting of the fixing bolt before and after the attachment of washer and nut, etc. The actuator should then be coupled to the moving link, checking that the front (plunger) end fitting enters the fork end of the linkage freely, i.e., without touching the sides of the form arm.

13. While the fixing bolts should be well lubricated with the approved grease, no lubricant should be applied to the actuator plunger as this is adequately lubricated on assembly.

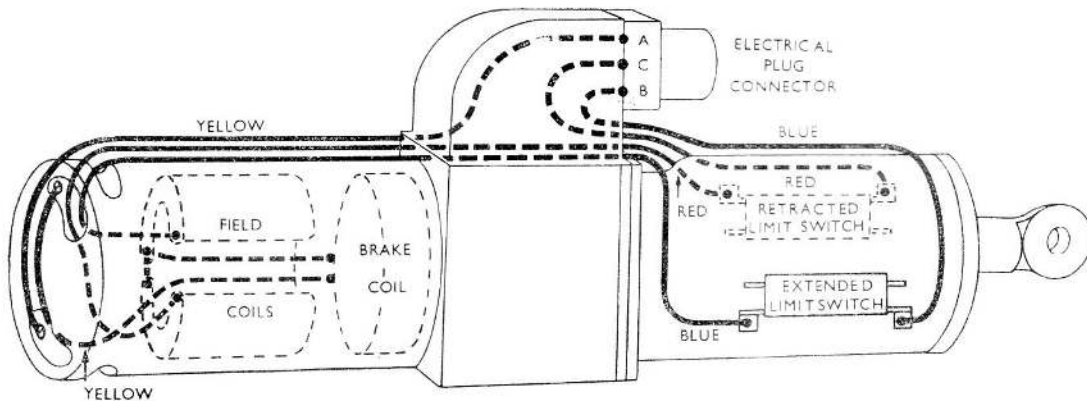


Fig. 4. Schematic wiring diagram

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14. After installation, the actuator should be given a test to ensure that it functions correctly.

### SERVICING

15. These actuators should be serviced in accordance with the general chapter in A.P.4343, Vol. 1, Sect. 17, Chap. 1, and the instructions contained in the relevant Servicing Schedule.

#### Lubrication

16. As the actuators are lubricated during

manufacture, they require no attention during normal service other than the application of a smear of approved grease to the pivot pins at each end of the actuator.

#### Testing

◀17. If the serviceability of an actuator is suspect apply the standard serviceability test detailed in the appropriate appendix to this chapter. If the actuator fails to meet the requirements of its standard serviceability test it should be disposed of in accordance with the current service instruction. ▶

## Appendix D

### STANDARD SERVICEABILITY TEST FOR ACTUATOR, PLESSEY, TYPE ICZ80358

#### Introduction

1. When considered necessary the tests detailed in this appendix may be applied to the above-mentioned actuator immediately prior to installation in an aircraft, or when its serviceability is suspect.

#### Test equipment

2. The following test equipment is required:—

Linear actuator test rig (Ref. No. 4G/5420)

250V insulation resistance tester, Type C (Ref. No. 5G/152)

A 29V d.c. supply

Tension gauge (Ref. No. 1H/59)

#### TEST PROCEDURE

#### Insulation resistance test

3. Using the insulation resistance tester,

check the resistance between each lead and the actuator body in turn. The reading in each instance must not be less than 50,000 ohms.

#### Brushgear

4. The brushgear is accessible when the cast end cover of the motor housing is removed. The length of the brushes should not be less than 0.25 in. and the brush spring pressure should be between 3.5 and 4.5 oz. (100 and 127 gm). Refit the motor housing cover.

#### Function test

5. The linear actuator test rig is described and illustrated in A.P.4343S, Vol. 1, Book 2, Sect. 8. Set the actuator to be tested on the rig and ensure that it operates within the limits given in Table 1.

6. Perform several inching strokes in each direction to check for satisfactory brake operation.

TABLE 1

Applied voltage	Load	Time for 1.4 in. stroke (sec.)		Max. current (amp.)	Length of stroke (in.)
		Min.	Max.		
29	50 lb. L.A.	4.7	6.7	1.6	1.4 ± 0.01
29	50 lb. L.O.	5.2	7.4	1.75	1.4 ± 0.01

L.A.—Load assisting motion  
L.O.—Load opposing motion

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## Appendix F

### STANDARD SERVICEABILITY TEST FOR ACTUATOR, PLESSEY, TYPE ICZ82400

#### Introduction

1. When considered necessary the tests detailed in this appendix may be applied to the above-mentioned actuator immediately prior to installation in an aircraft, or when its serviceability is suspect.

#### Test equipment

2. The following test equipment is required:—

Linear actuator test rig (Ref. No. 4G/5420)

250V insulation resistance tester, Type C (Ref. No. 5G/152)

A 29V d.c. supply.

Tension gauge (Ref. No. 1H/59).

#### TEST PROCEDURE

##### Insulation resistance test

3. Using the insulation resistance tester.

check the resistance between each lead and the actuator body in turn. The reading in each instance must not be less than 50,000 ohms.

##### Brushgear

4. The brushgear is accessible when the cast end cover of the motor housing is removed. The length of the brushes should not be less than 0.25 in. and the brush spring pressure should be between 3.5 and 4.5 oz. (100 and 127 gm.). Refit the motor housing cover.

##### Function test

5. The linear actuator test rig is described and illustrated in A.P.4343S, Vol. 1, Book 2, Sect. 8. Set the actuator to be tested on the rig and ensure that it operates within the limits given in Table 1.

6. Perform several inching strokes in each direction to check for satisfactory brake operation.

TABLE 1

Applied voltage	Load	Time for 1.4 in. stroke (sec.)		Max. current (amp.)	Length of stroke (in.)
		Min.	Max.		
29	50 lb. L.A.	9.0	13.0	1.5	1.4 ± 0.01
29	50 lb. L.O.	9.5	13.5	1.55	1.4 ± 0.01

L.A.—Load assisting motion  
L.O.—Load opposing motion

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## Appendix G

### STANDARD SERVICEABILITY TEST FOR ACTUATORS, PLESSEY, TYPE 1CZ135260, /A, /B AND /C

#### Introduction

1. When considered necessary the tests detailed in this appendix may be applied to the above-mentioned actuator immediately prior to installation in an aircraft, or when its serviceability is suspect.

#### Test equipment

2. The following test equipment is required:—

Linear actuator test rig (Ref. No. 4G/5420)

250V insulation resistance tester, Type C (Ref. No. 5G/152)

A 29V d.c. supply.

Tension gauge (Ref. No. 1H/59).

#### TEST PROCEDURE

##### Insulation resistance test

3. Using the insulation resistance tester,

check the resistance between each lead and the actuator body in turn. The reading in each instance must not be less than 50,000 ohms.

##### Brushgear

4. The brushgear is accessible when the cast end cover of the motor housing is removed. The length of the brushes should not be less than 0.25 in. and the brush spring pressure should be between 3.5 and 4.5 oz. (100 and 127 gm.). Refit the motor housing cover.

##### Function test

5. The linear actuator test rig is described and illustrated in A.P.4343S, Vol. 1, Book 2, Sect. 8. Set the actuator to be tested on the rig to ensure that it operates within the limits given in the appropriate Table.

6. Perform several inching strokes in each direction to check for satisfactory brake operation.

**TABLE 1**

1CZ135260

Applied voltage	Load	Time for 3 in. stroke (sec.)		Max. current (amp.)	Length of stroke (in.)
		Min.	Max.		
29	250 lb. L.A.	22	28	1.8	3 ± 0.03
29	250 lb. L.O.	28	35	2.0	3 ± 0.03

**TABLE 2**

1CZ135260/A

Applied voltage	Load	Time for 1.8 in. stroke (sec.)		Max. current (amp.)	Length of stroke (in.)
		Min.	Max.		
29	250 lb. L.A.	14.4	18.6	1.8	1.8 ± 0.03
29	250 lb. L.O.	16.8	23.4	2.0	1.8 ± 0.03

**TABLE 3**

1CZ135260/B

Applied voltage	Load	Time for 0.5 in. stroke (sec.)		Max. current (amp.)	Length of stroke (in.)
		Min.	Max.		
29	250 lb. L.A.	3.6	4.7	1.8	0.5 ± 0.03
29	250 lb. L.O.	4.6	5.9	2.0	0.5 ± 0.03

**TABLE 4**

1CZ135260/C

Applied voltage	Load	Time for 0.67 in. stroke (sec.)		Max. current (amp.)	Length of stroke (in.)
		Min.	Max.		
29	250 lb. L.A.	4.8	6.3	1.8	0.67 ± 0.03
29	250 lb. L.O.	6.2	7.9	2.0	0.67 ± 0.03

L.A.—Load assisting motion

L.O.—Load opposing motion

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## Appendix H

### STANDARD SERVICEABILITY TEST FOR ACTUATORS, PLESSEY, TYPES 1CZ135261, /A AND /B

#### Introduction

1. When considered necessary the tests detailed in this appendix may be applied to the above-mentioned actuator immediately prior to installation in an aircraft, or when its serviceability is suspect.

#### Test equipment

2. The following test equipment is required:—

Linear actuator test rig (Ref. No. 4G/5420)

250V insulation resistance tester, Type C (Ref. No. 5G/152)

A 29V d.c. supply.

Tension gauge (Ref. No. 1H/59).

#### TEST PROCEDURE

##### Insulation resistance test

3. Using the insulation resistance tester,

check the resistance between each lead and the actuator body in turn. The reading in each instance must not be less than 50,000 ohms.

##### Brushgear

4. The brushgear is accessible when the cast end cover of the motor housing is removed. The length of the brushes should not be less than 0.25 in. and the brush spring pressure should be between 3.5 and 4.5 oz. (100 and 127 gm.). Refit the motor housing cover.

##### Function test

5. The linear actuator test rig is described and illustrated in A.P.4343S, Vol. 1, Book 2, Sect. 8. Set the actuator to be tested on the rig and ensure that it operates within the limits given in the appropriate Table.

6. Perform several inching strokes in each direction to check satisfactory brake operation.

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**TABLE 1**

1CZ135261

Applied voltage	Load	Time for 2.5 in. stroke (sec.)		Max. current (amp.)	Length of stroke (in.)
		Min.	Max.		
29	250 lb. L.A.	20	26	1.8	2.5 ± 0.03
29	250 lb. L.O.	23	33	2.0	2.5 ± 0.03

**TABLE 2**

1CZ135261/A

Applied voltage	Load	Time for 0.45 in. stroke (sec.)		Max. current (amp.)	Length of stroke (in.)
		Min.	Max.		
29	250 lb. L.A.	3.6	4.7	1.8	0.45 ± 0.03
29	250 lb. L.O.	4.1	6.0	2.0	0.45 ± 0.03

**TABLE 3**

1CZ135261/B

Applied voltage	Load	Time for 0.73 in. stroke (sec.)		Max. current (amp.)	Length of stroke (in.)
		Min.	Max.		
29	250 lb. L.A.	5.8	7.6	1.8	0.73 ± 0.03
29	250 lb. L.O.	6.7	9.7	2.0	0.73 ± 0.03

L.A.—Load assisting motion

L.O.—Load opposing motion

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## Appendix I

### STANDARD SERVICEABILITY TEST FOR ACTUATOR, PLESSEY, TYPE 1CZ136150

#### Introduction

1. When considered necessary the tests detailed in this appendix may be applied to the above-mentioned actuator immediately prior to installation in an aircraft, or when its serviceability is suspect.

#### Test equipment

2. The following test equipment is required:—

Linear actuator test rig (Ref. No. 4G/5420)

250V insulation resistance tester, Type C (Ref. No. 5G/152)

A 29V d.c. supply.

Tension gauge (Ref. No. 1H/59).

#### TEST PROCEDURE

##### Insulation resistance test

3. Using the insulation resistance tester,

check the resistance between each lead and the actuator body in turn. The reading in each instance must not be less than 50,000 ohms.

##### Brushgear

4. The brushgear is accessible when the cast end cover of the motor housing is removed. The length of the brushes should not be less than 0.25 in. and the brush spring pressure should be between 3.5 and 4.5 oz. (100 and 127 gm.). Refit the motor housing cover.

##### Function test

5. The linear actuator test rig is described and illustrated in A.P.4343S, Vol. 1, Book 2, Sect. 8. Set the actuator to be tested on the rig and ensure that it operates within the limits given in Table 1.

6. Perform several inching strokes in each direction to check for satisfactory brake operation.

TABLE 1  
1CZ136150

Applied voltage	Load	Time for 3 in. stroke (sec.)		Max. current (amp.)	Length of stroke (in.)
		Min.	Max.		
29	250 lb. L.A.	22	28	1.8	3 ± 0.03
29	250 lb. L.O.	28	35	2.0	3 ± 0.03

L.A.—Load assisting motion  
L.O.—Load opposing motion

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## Appendix I

### ACTUATORS, PLESSEY, TYPES CZ63696/A and /B

#### LEADING PARTICULARS

Actuator, Type CZ63696/A	.....	.....	.....	.....	.....	Ref. No. 5W/1029
Type CZ63596/B	.....	.....	.....	.....	.....	Ref. No. 5W/1353

The following particulars are applicable to both variants:

Operating voltage range	.....	.....	.....	.....	.....	18 to 29V d.c.
Normal voltage	.....	.....	.....	.....	.....	28V d.c.
Working load — normal	.....	.....	.....	.....	.....	100 lb.
maximum	.....	.....	.....	.....	.....	150 lb.
Maximum static load	.....	.....	.....	.....	.....	1,000 lb.
Current consumption (normal load)	.....	.....	.....	.....	.....	1.75 amp.
Operating time (normal load)	.....	.....	.....	.....	.....	34 sec.
Ambient temperature range	.....	.....	.....	.....	.....	—60 to +90 deg. C.
Weight	.....	.....	.....	.....	.....	..... 3 lb.
Motor, Type C1606B/19	.....	.....	.....	.....	.....	Ref. No. 5W/3496
Weight	.....	.....	.....	.....	.....	12.5 oz.
Rating	.....	.....	.....	.....	.....	..... 1.5 minutes
Brushes, Type, CZ62177/1	.....	.....	.....	.....	.....	Ref. No. 5W/1041
Length — new	.....	.....	.....	.....	.....	0.355 to 0.385 in.
worn	.....	.....	.....	.....	.....	0.25 in.
Commutator diameter — new	.....	.....	.....	.....	.....	0.490 to 0.495 in.
worn	.....	.....	.....	.....	.....	0.450 in.
after skimming	.....	.....	.....	.....	.....	0.470 in.
Armature shaft journal diameter —	.....	.....	.....	.....	.....	.....
commutator end	.....	.....	.....	.....	.....	0.1249 to 0.1253 in.
drive end	.....	.....	.....	.....	.....	0.1874 to 0.1878 in.
Brake air gap	.....	.....	.....	.....	.....	0.008 to 0.011 in.
Brake disc and friction lining	.....	.....	.....	.....	.....	.....
Thickness — new	.....	.....	.....	.....	.....	0.120 to 0.125 in.
worn	.....	.....	.....	.....	.....	0.115 in.

The following table details the variation in stroke of each model:

Type	Stroke (in.)	Centres (from trunion centre point) Extended (in.)	Retracted (in.)
CZ63696/A	0.45	7.675	7.225
CZ63696/B	0.73	7.815	7.085

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1. The actuator, Type CZ63696, is similar to the typical actuator described and illustrated in the main chapter.

2. This model has no rear end eye fitting but is anchored by trunnions. The general construction is otherwise similar to the typical model.

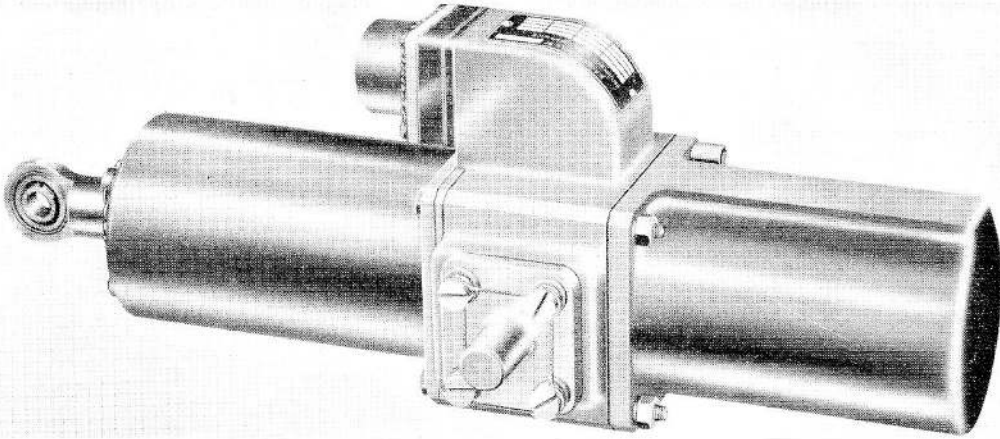


Fig. 1. General view of actuator, Type CZ63696



1. The actuator, Type CZ64250, is similar to the typical actuator described and illustrated in the main chapter. The peculiarities of this variant are given in Leading Particulars.

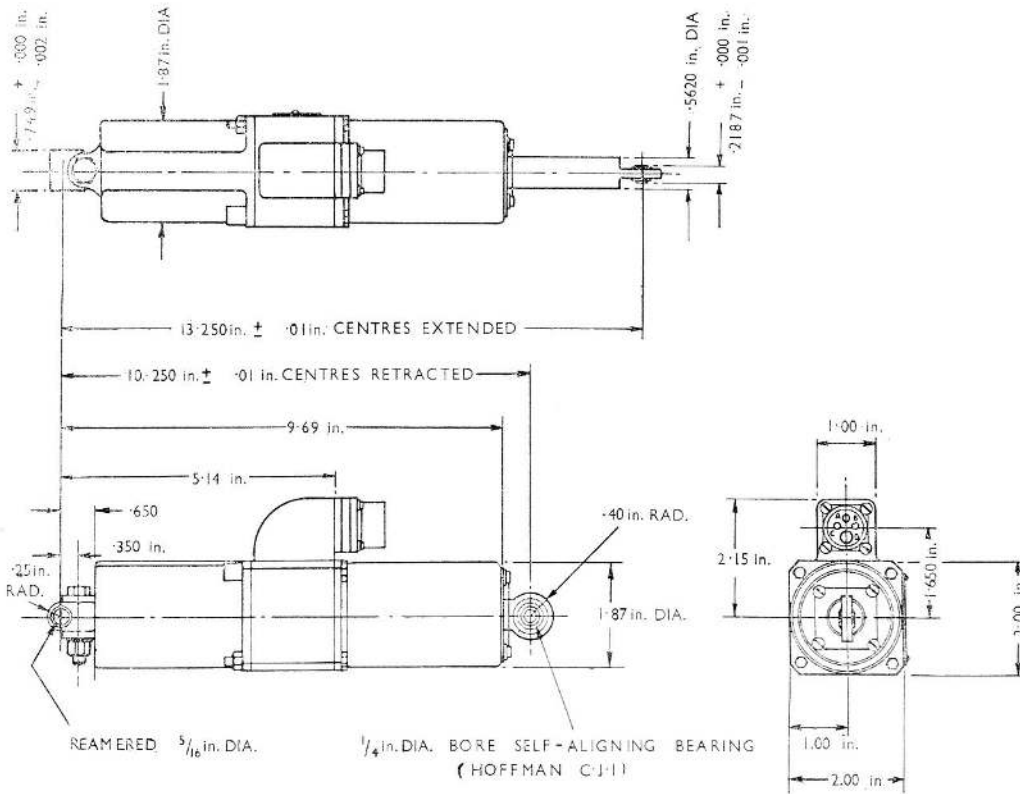


Fig. 1. Stroke setting and installation details

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### Appendix 3

## ACTUATOR, PLESSEY, TYPE CZ77790

### LEADING PARTICULARS

<b>Actuator, Type CZ77790</b> .....	Ref. No. 5W/990
<i>Operating voltage range</i> .....	18 to 29V d.c.
<i>Normal voltage</i> .....	28V d.c.
<i>Working load — normal</i> .....	100 lb.
<i>maximum</i> .....	250 lb.
<i>Maximum static load</i> .....	1,000 lb.
<i>Operational period (normal load and stroke)</i> .....	12 seconds
<i>Rating at normal working load</i> .....	1.5 minutes
<i>Length of stroke</i> .....	1.00 in.
<i>Travel between mechanical stops</i> .....	1.06 in.
<i>Fixing centres retracted</i> .....	4.565 ± 0.010 in.
<i>Fixing centres extended</i> .....	5.565 ± 0.010 in.
<i>Weight</i> .....	3 lb. 6 oz.

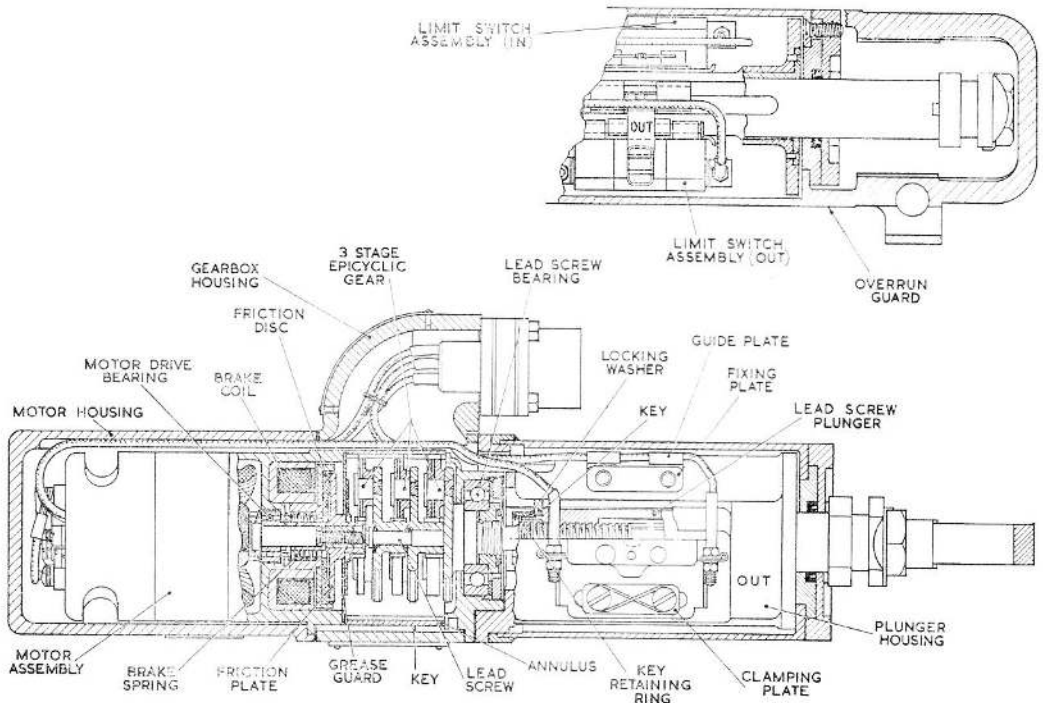


Fig. 1. Sectional view of actuator

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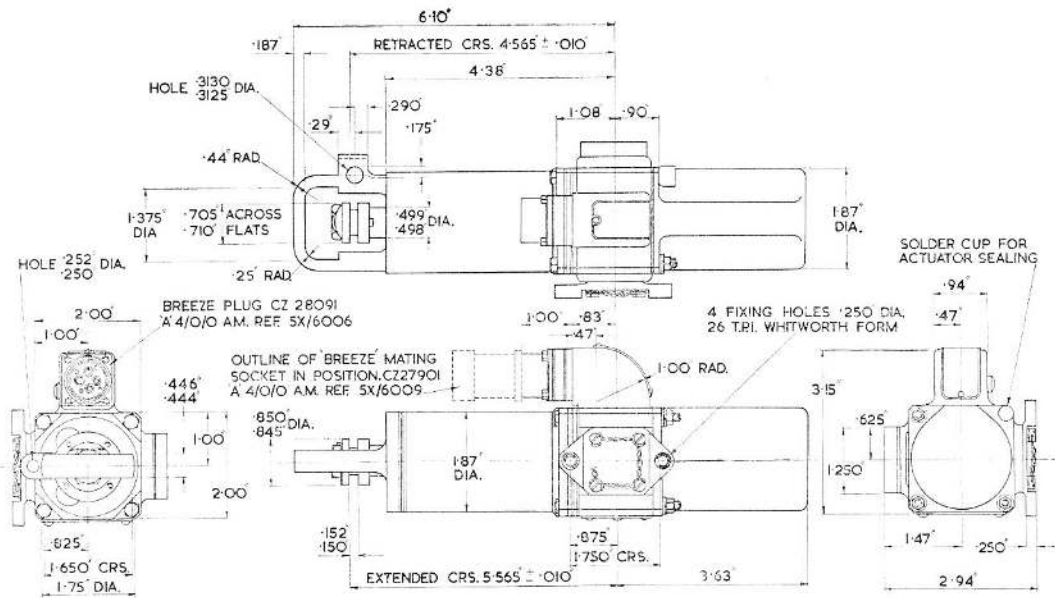


Fig. 2. Stroke setting and installation details

1. The actuator, Type CZ77790, is similar to the typical actuator described and illustrated in the main chapter.
2. The circuit diagram of this model differs from the typical actuator. An additional connection is made to pole D as shown in fig. 3.
3. This actuator has no rear end eye fitting,

but is fitted with two side mounting brackets, one disposed vertically, the other horizontally.

4. These side mounting brackets hold the actuator in its installation position in the aircraft, each bracket being secured by two  $\frac{1}{4}$  in. B.S.F. screws. The other end of the actuator is coupled by its special end fitting to the valve it will operate.

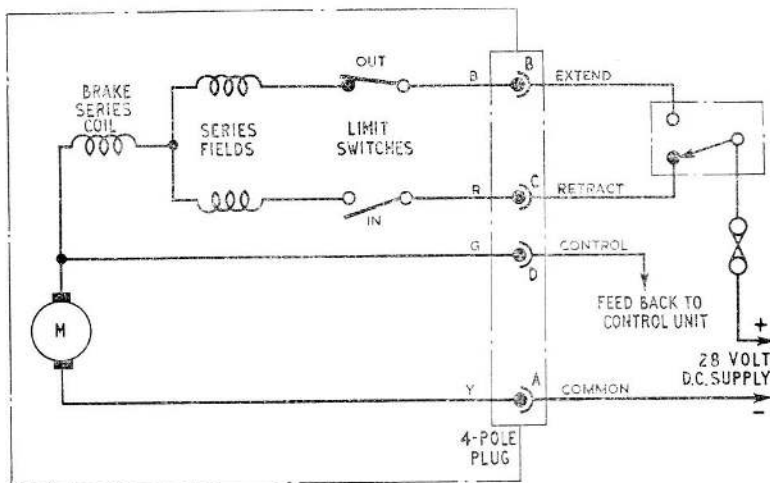


Fig. 3. Circuit diagram

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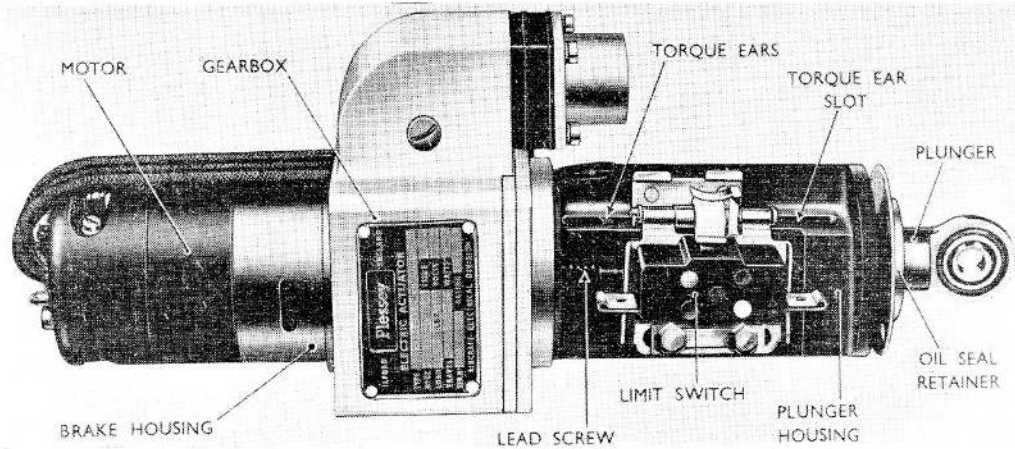


Fig. 1. View of actuator with covers removed

1. The actuator, Type 1CZ80358, is similar to the typical model described and illustrated in the main chapter. The peculiarities of this variant are given in Leading Particulars.

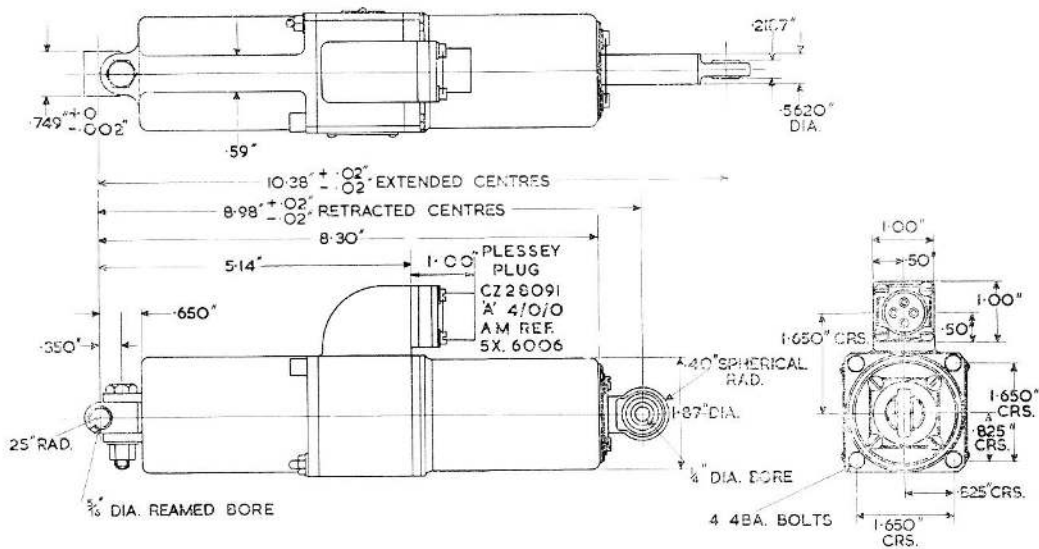


Fig. 2. Installation drawing

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## Appendix 6

### ACTUATOR, PLESSEY, TYPE 1CZ82400

#### LEADING PARTICULARS

<b>Actuator, Type 1CZ82400</b> .....	<i>Ref. No. 5W/1392</i>
<i>Operating voltage range</i> .....	18 to 29V d.c.
<i>Normal voltage</i> .....	28V d.c.
<i>Working load — normal</i> .....	50 lb.
<i>maximum</i> .....	150 lb.
<i>Maximum static load</i> .....	1,000 lb.
<i>Current consumption (normal load)</i> .....	1.5 amp.
<i>Operating time for 1.4 in. stroke</i> .....	9 seconds minimum
<i>at 29V d.c. (normal load)</i> .....	13.5 seconds maximum
<i>Plunger stroke</i> .....	1.4 in.
<i>Fixing centres — retracted</i> .....	8.98±0.020 in.
<i>extended</i> .....	10.38±0.020 in.
<i>Ambient temperature range</i> .....	-40 to +90 deg. C.
<i>Weight</i> .....	2 lb. 11 oz.
<b>Motor, Type C1606B/5</b> .....	<i>Ref. No. 5W/2727</i>
<i>Weight</i> .....	12.5 oz.
<i>Rating</i> .....	1.5 minutes
<b>Brushes, Type CZ62177/1</b> .....	<i>Ref. No. 5W/1041</i>
<i>Length — new</i> .....	0.355 to 0.385 in.
<i>worn</i> .....	0.25 in.
<i>Commutator diameter — new</i> .....	0.490 to 0.495 in.
<i>worn</i> .....	0.450 in.
<i>after skimming</i> .....	0.470 in.
<i>Armature shaft journal diameter —</i>	
<i>commutator end</i> .....	0.1249 to 0.1253 in.
<i>drive end</i> .....	0.1874 to 0.1878 in.
<i>Brake air gap</i> .....	0.008 to 0.0125 in.
<i>Brake disc and friction lining</i>	
<i>thickness — new</i> .....	0.120 to 0.125 in.
<i>worn</i> .....	0.115 in.

1. The actuator, Type 1CZ82400, is similar to the typical actuator described and illustrated in the main chapter. The peculiarities of this variant are given in Leading Particulars.

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## Appendix 8

## ACTUATORS, PLESSEY, TYPES 1CZ135261, /A and /B

## LEADING PARTICULARS

Actuator, Type 1CZ135261	.....	.....	.....	.....	Ref. No.
Type 1CZ135261/A	.....	.....	.....	.....	Ref. No. 5W/2575
Type 1CZ135261/B	.....	.....	.....	.....	Ref. No. 5W/2576

The following particulars are applicable to all three types:

Operating voltage range	.....	.....	.....	.....	22 to 29V d.c.
Normal voltage	.....	.....	.....	.....	28V d.c.
Working load — normal	.....	.....	.....	.....	250 lb.
maximum	.....	.....	.....	.....	375 lb.
Maximum static load	.....	.....	.....	.....	1,000 lb.
Plunger thread	.....	.....	.....	.....	Acme, single-start, R.H., 16T.P.I.
Current consumption (normal load)	.....	.....	.....	.....	1.95 amp.
Maximum operating frequency	.....	.....	.....	.....	50 inching cycles per hour
Ambient temperature range	.....	.....	.....	.....	-50 to +90 deg. C.
Weight	.....	.....	.....	.....	3 lb.
Motor, Type 1606B/19	.....	.....	.....	.....	Ref. No. 5W/3496
Weight	.....	.....	.....	.....	12.5 oz.
Rating	.....	.....	.....	.....	1.5 minutes
Brushes, Type CZ62177/1	.....	.....	.....	.....	Ref. No. 5W/1041
Length — new	.....	.....	.....	.....	0.355 to 0.385 in.
worn	.....	.....	.....	.....	0.25 in.
Commutator diameter — new	.....	.....	.....	.....	0.490 to 0.495 in.
worn	.....	.....	.....	.....	0.450 in.
after skimming	.....	.....	.....	.....	0.470 in.
Armature shaft journal diameter —	.....	.....	.....	.....	
commutator end	.....	.....	.....	.....	0.1249 to 0.1253 in.
drive end	.....	.....	.....	.....	0.1874 to 0.1878 in.
Brake air gap	.....	.....	.....	.....	0.008 to 0.011 in.
Brake disc and friction lining	.....	.....	.....	.....	
Thickness — new	.....	.....	.....	.....	0.120 to 0.125 in.
worn	.....	.....	.....	.....	0.115 in.
Reduction gear ratio	.....	.....	.....	.....	178 : 1

The following table details the variation in stroke of the three actuators.

Type	Stroke (in.)	Centres (in.)	
		Extended	Retracted
1CZ135261	2.5	8.700	6.200
1CZ135261/A	0.45	7.675	7.225
1CZ135261/B	0.73	7.815	7.085

Extended and retracted centres refer to the distances between the centres of the actuator trunnions and front end fittings, at the extreme limits of plunger travel.

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1. Plessey linear actuator, Type ICZ135261, is a Cub Puma model in the series described in the main chapter. Types ICZ135261/A and ICZ135261/B are variants of the basic Type ICZ135261. The three types are similar in construction and operation; the only difference between them is in the length of their working strokes, details of which are given in the Leading Particulars. Fig. 1 shows a cutaway view of the basic type. Actuator ICZ135261/A is easily identified by the engraving VAL STI 5 on the motor housing, no lettering being engraved on the housings of the other two types.

2. All three actuators follow the general working principle outlined in the main chapter where a fractional horsepower reversible motor extends and retracts a plunger through a gearbox and leadscrew. Rotary movement of the leadscrew is converted into linear motion of the plunger by the torque-reaction ears of a special key fitted to the plunger; the ears locate in slots machined in the plunger housing and restrain the plunger from rotating with the leadscrew. The limits of the plunger travel are controlled by switches mounted at the sides of the plunger housing. These switches are normally "ON" and interrupt power supply to the motor when actuated by the torque-reaction ears of the plunger key. The mounted positions of

the switches determine the length of the plunger stroke.

3. The actuators are mounted on trunnions and linked to associated equipment by front end fittings. The trunnions are secured by their integral flanges to the sides of the actuator gearboxes; the front end fittings are screwed and pinned to their respective plungers. The types of trunnions and end fittings used are the same for all three models.

4. Whilst the components of the three actuators are of the same design as those described in the main chapter, the material of the plunger tube differs in these actuators, in that it is phosphor-bronze.

5. A circuit diagram is shown in fig. 3 of the main chapter; it is applicable to all three actuators. Two single-pole, change-over type micro-switches and a four-pole plug are used, although only two terminals, at each switch, and three plug pins are connected into the circuit.

6. An installation diagram, applicable to all three actuators is shown in fig. 2. For details of an actuator installation in a particular aircraft, reference should be made to the appropriate aircraft Air Publication.

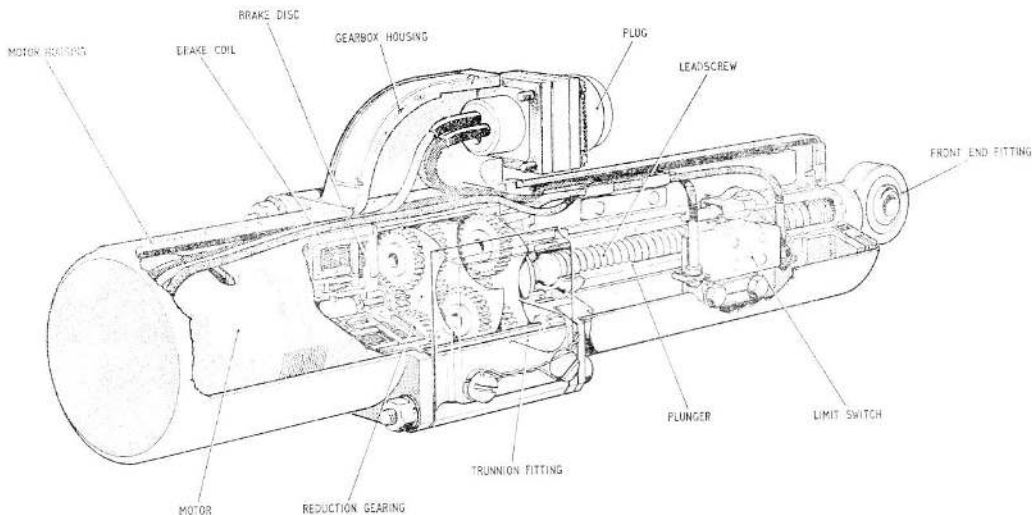
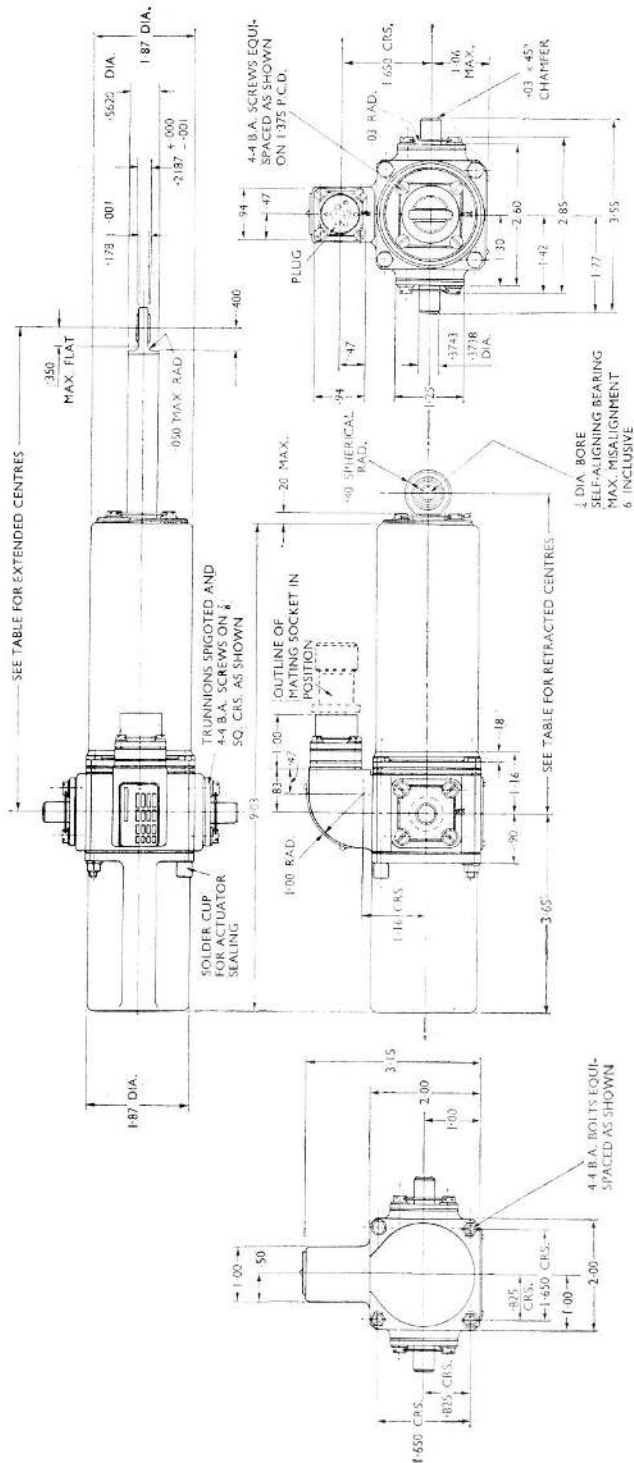


Fig. 1. Cutaway view of actuator, Type ICZ 135261

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Fig. 2. Installation diagram

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7. When fitting a new or overhauled unit, first check that the actuator is of the correct type for the installation; no adjustment of the plunger travel is possible in service since a special test rig is used to obtain the required accuracy. Ensure the actuator has not been damaged in transit and that all external screws, nuts and bolts are fully tightened and locked. After the actuator has been installed, check the security of the installation and

operate the actuator to ensure that it functions correctly.

8. The servicing instructions given in the main chapter are applicable to these actuators. The only additional information required is that regarding lubrication of the plunger which must be lightly greased at scheduled servicing periods.

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## Appendix 9

## ACTUATOR, PLESSEY, TYPE 1CZ136150

## LEADING PARTICULARS

<b>Actuator, Type 1CZ136150</b> .....	Ref. No. 5W/3848
<i>Operating voltage range</i> .....	22 to 29V d.c.
<i>Normal voltage</i> .....	28V d.c.
<i>Working load — normal</i> .....	250 lb.
<i>maximum</i> .....	375 lb.
<i>Maximum static load</i> .....	1,000 lb.
<i>Current consumption (normal load)</i> .....	1.95 amp.
<i>Maximum operating frequency</i> .....	50 inching cycles per hour
<i>Plunger — stroke</i> .....	3 in.
<i>thread</i> .....	Acme, single start, R.H., 18T.P.I.
<i>Centres — retracted</i> .....	10.25 in.
<i>extended</i> .....	13.25 in.
<i>End fitting — rear</i> .....	Z59598
<i>Ambient temperature range</i> .....	−50 to +90 deg. C.
<i>Weight</i> .....	3 lb.
<b>Motor, Type C1606B/19</b> .....	Ref. No. 5W/3496
<i>Weight</i> .....	12.5 oz.
<i>Rating</i> .....	1.5 minutes
<b>Brushes, Type CZ62177/1</b> .....	Ref. No. 5W/1041
<i>Length — new</i> .....	0.355 to 0.385 in.
<i>worn</i> .....	0.25 in.
<i>Commutator diameter — new</i> .....	0.490 to 0.495 in.
<i>worn</i> .....	0.450 in.
<i>after skimming</i> .....	0.470 in.
<i>Armature shaft journal diameter —</i>	
<i>commutator end</i> .....	0.1249 to 0.1253 in.
<i>drive end</i> .....	0.1874 to 0.1878 in.
<i>Brake air gap</i> .....	0.008 to 0.011 in.
<i>Brake disc and friction lining</i>	
<i>Thickness — new</i> .....	0.120 to 0.125 in.
<i>worn</i> .....	0.115 in.
<i>Reduction gear ratio</i> .....	178 : 1

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1. Plessey linear actuator, Type 1CZ136150, is a variant in the Cub Puma series described in the main chapter. A cutaway view of the actuator is shown in fig. 1. The following text describes the salient features of the actuator and establishes any departures, from the general design, not covered by the Leading Particulars.

2. The actuator operating principle follows that outlined in the main chapter where a fractional-horsepower reversible motor extends and retracts a plunger through a gearbox and leadscrew. Rotary movement of the leadscrew is converted into linear motion of the plunger by the torque-reaction ears of a special key fitted to the plunger; the ears locate in slots machined in the plunger housing and restrain the plunger from rotating with the leadscrew. The limits of plunger travel are controlled by switches mounted at the sides of the plunger housing. These switches are normally "ON" and interrupt power supply to the motor when actuated by the torque-reaction ears of the plunger key.

3. End fittings are provided for mounting the actuator and for linking it to associated equipment. The rear end fitting is bolted to a bracket integral with the motor cover; the front end fitting is screwed and pinned to the plunger.

4. Whilst the actuator components are of

the same design as those described in the main chapter, the plunger tube in this variant is made of phosphor-bronze.

5. Fig. 3 shows the schematic diagram for the wiring of this actuator; it includes the addition to the circuitry typified in the main chapter. Two single pole, change-over type micro-switches and a four-pole plug (Ref. No. 5X/6006), are used. The circuit is designed to provide normal remote control of the actuator and, in addition, to enable an external warning light to be connected into the circuit. The connections are so made that the light should automatically operate when the plunger has reached the pre-set limit of travel in the "extend" position.

6. The general information given in the main chapter regarding installation of the actuator is applicable to this variant. However, although the recommendation regarding lubrication of the plunger is still applicable, it is necessary to maintain a thin film of grease on the plunger at all times. This instruction has arisen as, subsequent to the issue of the original advice, it has been noted that lubricant applied during assembly has been inadvertently or erroneously wiped off the plunger prior to installation of the actuator. Reference must be made to the appropriate Air Publication for details of installation in a particular aircraft.

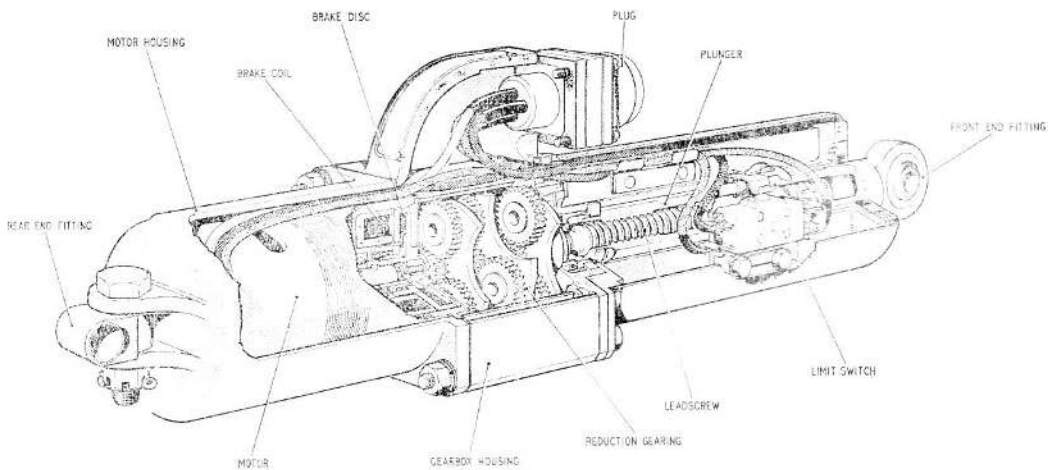


Fig. 1. Cutaway view of actuator

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7. The servicing instructions detailed in the main chapter are applicable to this actuator. The only additional information required is

that regarding lubrication of the plunger which must be lightly greased at scheduled servicing periods.

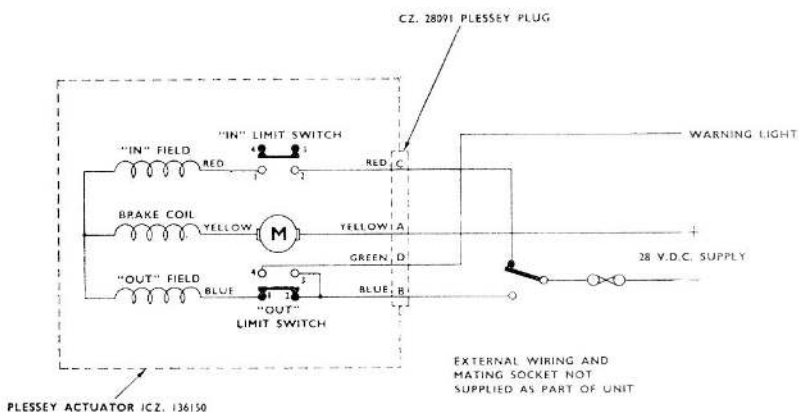


Fig. 3. Circuit diagram

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